

Convolution Fatigue & SCC Lifetime

■ E2400425-v2

Initial Release

Fatigue life prediction (number of cycles) for convolutions (bellows, expansion joints and convoluted cylinders) using EJMA-1993 and ASME BPVC.VIII.2-2023

Estimation of possible Stress-Corrosion Cracking (SCC) Lifetime for convolutions

Dennis Coyne, 7-Dec-2024

Introduction

The Cosmic Explorer (CE) and Einstein Telescope (ET) projects are considering the use of corrugated, thin-walled, tubing for their beam tubes as an alternative to thick-walled, smooth, straight piping or ring-stiffened piping (for a number of reasons; see for example <https://dcc.ligo.org/LIGO-T2400337>). The unstiffened and ring-stiffened tubes require bellows (expansion joints) to allow for expansion during vacuum bakeouts, as well as adjustments for alignment. The corrugated tube alternative does not require these discrete bellows (expansion joints) since they are (in effect) long bellows. However, the corrugated tube option incorporates many more convolutions to the overall assembly than a set of discrete bellows. If reliability against UHV leaks scales with convolutions, then the reliability of the system may be insufficient.

The convolutions are typically formed by either mechanical expansion or hydroforming. In either case, high plastic strains are induced in the forming of the bellows. Regardless of the method employed in forming the initial cylindrical shell (roll formed or spiral welded), axisymmetric convolutions will cross the weld. The heat affected zone (HAZ) of the weld is often the initiation site of microcracks leading to leaks later during the operational phase of the system. (If the tube is spiral welded and also spirally corrugated, at the same helix angle, then the corrugations would not cross the weld.)

The CERN CE vacuum team have demonstrated crack formation issues where the corrugation crosses the weld when autogenous TIG welding AISI 430. They have also demonstrated dramatic reduction in grain coarsening and improved formability (for corrugating) when employing autogenous, continuous wave (CW), laser welding of AISI 441:

M. Dakshinamurthy and A. T. Pérez, CERN, Materials and manufacturing for the pilot sector, 22 Jan 2024

Fatigue

■ nomenclature

Following ASME BPVC.VIII.2-2023, Part 4.19, "Design Rules for Bellows Expansion Joints".

for unreinforced bellows

Part 4.19.5.6, "External Pressure Strength":

4.19.5.6(a) External pressure capacity per Table 4.19.2

4.19.5.6(b) and (c) Instability due to external pressure

Part 4.19.5.7, "Fatigue Evaluation", Table 4.19.4, Table 4.19.5, Table 4.19.6

A = cross-sectional area of one convolution

C1, C2 = coefficients used to determine the coefficients Cd, Cf, Cp

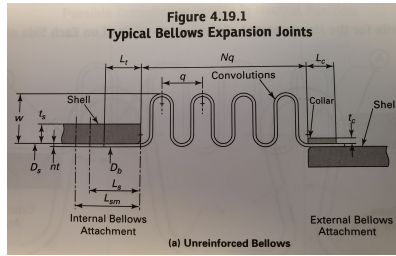
Cd, Cf, Cp = stress coefficients for U-shaped convolutions

Db = inside diameter of bellows convolution and end tangents

Dc = mean diameter of tangent collar

Dm = mean diameter of bellows convolutions

D_s = inside diameter of cylindrical shell or weld end on which the bellows is attached
 E_b = modulus of elasticity of bellows material at design temperature
 E_0 = modulus of elasticity of bellows material at room temperature
 E_c = modulus of elasticity of collar material at design temperature
 ϵ_f = bellows forming strain (depends on whether formed from cylinder of D_b or D_m diameter)
 E_s = modulus of elasticity of shell or weld end material at design temperature
 k = factor considering the stiffening effect of the attachment weld and the end convolution on the pressure capacity of the end tangent
 K_f = forming method factor: 1.0 for expanding mandrel or roll forming; 0.6 for hydraulic, elastomeric or pneumatic tube forming
 K_g = fatigue strength reduction factor, $1 < K_g < 4$
 K_m = 1.5 Y_{sm} for as-formed bellows; 1.5 for annealed bellows
 L_c = bellows collar length
 $L_s = \sqrt{(D_s + t_s) t_s} / 3$ = effective shell length
 L_t = end tangent length
 n = number of plies
 N_c = number of convolutions
 N_{alw} = allowable number of displacement cycles
 N_{spe} = number of specified (required) displacement cycles
 P = design pressure
 q = convolution pitch
 r_i = average internal radius of U-shaped convolution
 r_m = mean radius of U-shaped convolution
 S = allowable stress from Annex 3-A of bellows material at design temperature
 S_s = allowable stress from Annex 3-A of shell material at design temperature
 S_t = total stress range due to cyclic displacement
 S_1 = circumferential membrane stress in bellows tangent, due to pressure P
 S_{1ppp} = circumferential membrane stress in shell for internally attached bellows
 S_{2I} = circumferential membrane stress in intermediate convolutions
 S_3 = meridional membrane stress
 S_4 = meridional bending stress
 S_5 = meridional membrane stress due to the total equivalent axial displacement range Δq_e
 S_6 = bending stress due to the total equivalent axial displacement range Δq_e
 t = nominal thickness of a ply
 t_c = collar thickness
 t_p = thickness of one ply, corrected for thinning during forming
 t_s = nominal thickness of shell or weld end
 w = convolution height
 Y_{sm} = yield strength multiplier depending upon material
 x_e, x_c, y, θ = max axial extension, max axial compression, lateral and angular displacements
 Δq_e = total equivalent axial displacement per convolution



■ coefficient Cp

■ Table 4.19.3 data

In[186]:= $c2 = \{0.2, 0.4, 0.6, 0.8, 1.0, 1.2, 1.4, 1.6, 2.0, 2.5, 3.0, 3.5, 4.0\};$

□ Set A of coefficients $a_0, a_1, a_2, a_3, a_4, a_5$ are for $c_1 \leq 0.3$:

In[187]:= $Aa0 = \{1.001, 0.999, 0.961, 0.955, 0.95, 0.95, 0.95, 0.95, 0.95, 0.95, 0.95, 0.95, 0.95\};$

$Aa1 = \{-0.448, -0.735, -1.146, -2.708, -2.524, -2.296, -2.477, -2.027, -2.073, -2.073, -2.073, -2.073, -2.073\};$

$Aa2 = \{-1.244, 0.106, 3.023, 7.279, 10.402, 1.63, 7.823, -5.264, -3.622, -3.622, -3.622, -3.622, -3.622\};$

$Aa3 = \{1.923, -0.585, -7.488, 14.212, -93.848, 16.03, -49.394, 48.303, 29.136, 29.136, 29.136, 29.136, 29.136\};$

$Aa4 = \{-0.398, 1.787, 8.824, -104.242, 423.636, -113.939, 141.212, -139.394, -49.394, -49.394, -49.394, -49.394, -49.394\};$

$Aa5 = \{-0.291, -1.022, -3.634, 133.333, -613.333, 240, -106.667, 160, 13.333, 13.333, 13.333, 13.333, 13.333\};$

In[205]:= $Afc2a0 = \text{Interpolation}[\text{Transpose}[\{c2, Aa0\}], \text{InterpolationOrder} \rightarrow 1];$

$Afc2a1 = \text{Interpolation}[\text{Transpose}[\{c2, Aa1\}], \text{InterpolationOrder} \rightarrow 1];$

$Afc2a2 = \text{Interpolation}[\text{Transpose}[\{c2, Aa2\}], \text{InterpolationOrder} \rightarrow 1];$

$Afc2a3 = \text{Interpolation}[\text{Transpose}[\{c2, Aa3\}], \text{InterpolationOrder} \rightarrow 1];$

$Afc2a4 = \text{Interpolation}[\text{Transpose}[\{c2, Aa4\}], \text{InterpolationOrder} \rightarrow 1];$

$Afc2a5 = \text{Interpolation}[\text{Transpose}[\{c2, Aa5\}], \text{InterpolationOrder} \rightarrow 1];$

In[211]:= $ACp[c1_, c2_] :=$

$Afc2a0[c2] + Afc2a1[c2] c1 + Afc2a2[c2] c1^2 + Afc2a3[c2] c1^3 + Afc2a4[c2] c1^4 + Afc2a5[c2] c1^5;$

□ Set B of coefficients $a_0, a_1, a_2, a_3, a_4, a_5$ are for $c_1 > 0.3$:

In[193]:= $Ba0 = \{1.001, 0.999, 0.961, 0.622, 0.201, 0.598, 0.473, 0.477, 0.935, 1.575, 1.464, 1.495, 2.037\};$

$Ba1 = \{-0.448, -0.735, -1.146, 1.685, 2.317, -0.99, -0.029, -0.146, -3.613, -8.646, -7.098, -6.904, -11.037\};$

$Ba2 = \{-1.244, 0.106, 3.023, -9.347, -5.956, 3.741, -0.015, -0.018, 9.456, 24.368, 17.875, 16.024, 28.276\};$

$Ba3 = \{1.932, -0.585, -7.488, 18.447, 7.594, -6.453, -0.03, 0.037, -13.228, -35.239, -23.778, -19.6, -37.655\};$

$Ba4 = \{-0.398, 1.787, 8.824, -15.991, -4.945, 5.107, 0.016, 0.097, 9.355, 25.313, 15.953, 12.069, 25.213\};$

$Ba5 = \{-0.291, -1.022, -3.634, 5.119, 1.299, -1.527, 0.016, -0.067, -2.613, -7.157, -4.245, -2.944, -6.716\};$

```

In[234]:= Bfc2a0 = Interpolation[Transpose[{c2, Ba0}], InterpolationOrder → 1];
Bfc2a1 = Interpolation[Transpose[{c2, Ba1}], InterpolationOrder → 1];
Bfc2a2 = Interpolation[Transpose[{c2, Ba2}], InterpolationOrder → 1];
Bfc2a3 = Interpolation[Transpose[{c2, Ba3}], InterpolationOrder → 1];
Bfc2a4 = Interpolation[Transpose[{c2, Ba4}], InterpolationOrder → 1];
Bfc2a5 = Interpolation[Transpose[{c2, Ba5}], InterpolationOrder → 1];

In[240]:= BCp[c1_, c2_] :=
  Bfc2a0[c2] + Bfc2a1[c2] c1 + Bfc2a2[c2] c1^2 + Bfc2a3[c2] c1^3 + Bfc2a4[c2] c1^4 + Bfc2a5[c2] c1^5;

```

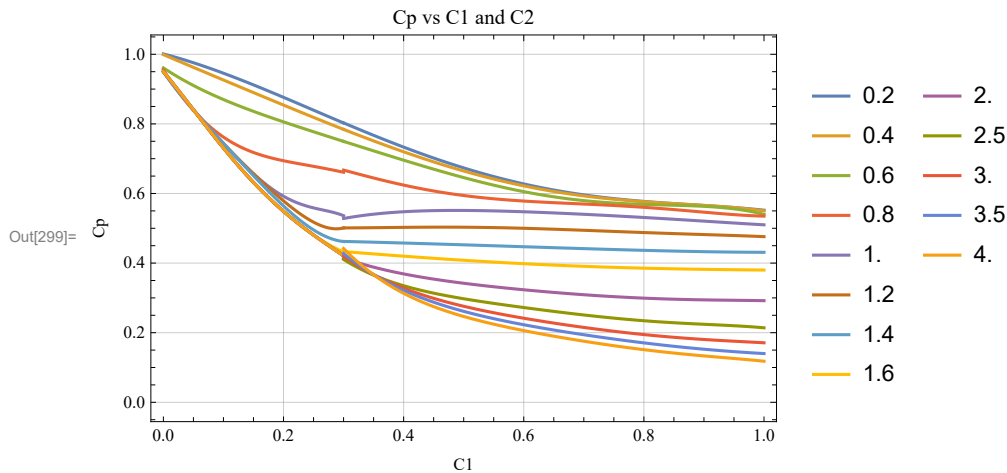
□ **plot Cp data**

Close match to Figure 4.19.12:

```

In[297]:= Aplt = Plot[{ACp[x, c2[[1]]], ACp[x, c2[[2]]], ACp[x, c2[[3]]], ACp[x, c2[[4]]],
  ACp[x, c2[[5]]], ACp[x, c2[[6]]], ACp[x, c2[[7]]], ACp[x, c2[[8]]], ACp[x, c2[[9]]],
  ACp[x, c2[[10]]], ACp[x, c2[[11]]], ACp[x, c2[[12]]], ACp[x, c2[[13]]]}, {x, 0, 0.3}];
Bplt = Plot[{BCp[x, c2[[1]]], BCp[x, c2[[2]]], BCp[x, c2[[3]]], BCp[x, c2[[4]]], BCp[x, c2[[5]]],
  BCp[x, c2[[6]]], BCp[x, c2[[7]]], BCp[x, c2[[8]]], BCp[x, c2[[9]]], BCp[x, c2[[10]]],
  BCp[x, c2[[11]]], BCp[x, c2[[12]]], BCp[x, c2[[13]]]}, {x, 0.3, 1}, PlotLegends → c2];
Show[Aplt, Bplt, PlotRange → {{0, 1}, {0, 1}}, Axes → False, Frame → True,
  GridLines → Automatic, FrameLabel → {"C1", "Cp"}, PlotLabel → "Cp vs C1 and C2"]

```



■ **coefficient Cf**

■ **Table 4.19.4 data**

```

In[ ]:= c2 = {0.2, 0.4, 0.6, 0.8, 1.0, 1.2, 1.4, 1.6, 2.0, 2.5, 3.0, 3.5, 4.0};

```

□ **coefficients a0,a1,a2,a3,a4,a5:**

```

In[318]:= Cfa0 = {1.006, 1.007, 1.003, 1.003, 0.997, 1, 1, 1.001, 1.002, 1, 0.999, 0.998, 1};
Cfa1 =
  {2.375, 1.82, 1.993, 1.338, 0.621, 0.112, -0.285, -0.494, -1.061, -1.31, -1.521, -1.896, -2.007};
Cfa2 = {-3.977, -1.818, -5.055, -1.717, -0.907, -1.41, -1.309,
  -1.879, -0.715, -0.829, -0.039, 1.839, 1.62};
Cfa3 = {8.297, 2.981, 12.896, 1.908, 2.429, 3.483, 3.662, 4.959,
  3.103, 4.116, 2.121, -2.047, -0.538};
Cfa4 = {-8.394, -2.43, -14.429, 0.02, -2.901, -3.044, -3.467,
  -4.569, -3.016, -4.36, -2.215, 1.852, -0.261};
Cfa5 = {3.194, 0.87, 5.897, -0.55, 1.361, 1.013, 1.191, 1.543, 0.99, 1.55, 0.77, -0.664, 0.249};

```

```

In[324]:= Cfa0 = Interpolation[Transpose[{c2, Cfa0}], InterpolationOrder → 1];
Cfa1 = Interpolation[Transpose[{c2, Cfa1}], InterpolationOrder → 1];
Cfa2 = Interpolation[Transpose[{c2, Cfa2}], InterpolationOrder → 1];
Cfa3 = Interpolation[Transpose[{c2, Cfa3}], InterpolationOrder → 1];
Cfa4 = Interpolation[Transpose[{c2, Cfa4}], InterpolationOrder → 1];
Cfa5 = Interpolation[Transpose[{c2, Cfa5}], InterpolationOrder → 1];

In[330]:= Cf[c1_, c2_] := Cfa0[c2] + Cfa1[c2] c1 + Cfa2[c2] c1^2 + Cfa3[c2] c1^3 + Cfa4[c2] c1^4 + Cfa5[c2] c1^5;

□ plot Cf data

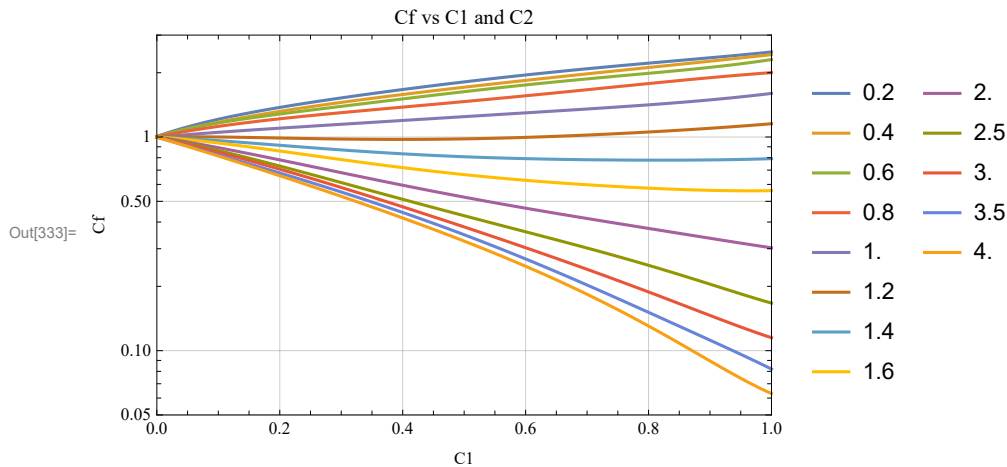
```

Close match to Figure 4.19.13:

```

In[333]:= LogPlot[{Cf[x, c2[[1]]], Cf[x, c2[[2]]], Cf[x, c2[[3]]], Cf[x, c2[[4]]],
  Cf[x, c2[[5]]], Cf[x, c2[[6]]], Cf[x, c2[[7]]], Cf[x, c2[[8]]], Cf[x, c2[[9]]],
  Cf[x, c2[[10]]], Cf[x, c2[[11]]], Cf[x, c2[[12]]], Cf[x, c2[[13]]]}, {x, 0, 1},
  PlotRange → {{0, 1}, {0.05, 3}}, Axes → False, Frame → True, GridLines → Automatic,
  FrameLabel → {"C1", "Cf"}, PlotLabel → "Cf vs C1 and C2", PlotLegends → c2]

```



■ coefficient Cd

■ Table 4.19.4 data

```

In[351]:= c2 = {0.2, 0.4, 0.6, 0.8, 1.0, 1.2, 1.4, 1.6, 2.0, 2.5, 3.0, 3.5, 4};

```

□ coefficients a0,a1,a2,a3,a4,a5:

```

In[352]:= Cda0 = {1, 0.999, 1.003, 1.005, 1.001, 1.002, 0.998, 0.999, 1, 1, 1, 1, 1.001};
Cda1 =
  {1.151, 1.31, 2.189, 1.263, 0.953, 0.602, 0.309, 0.12, -0.133, -0.323, -0.545, -0.704, -0.955};
Cda2 = {1.685, 0.909, -3.192, 5.184, 3.924, 2.11, 1.135, 0.351, -0.46, -1.118, -0.42, -0.179, 0.577};
Cda3 = {-4.414, -2.407, 5.928, -13.929,
  -8.773, -3.625, -1.04, -0.178, 1.596, 3.73, 1.457, 0.946, -0.462};
Cda4 = {4.564, 2.273, -5.576, 13.828, 10.44, 5.166, 1.296, 0.942,
  -1.521, -4.453, -1.561, -1.038, 0.181};
Cda5 = {-1.645, -0.706, 2.07, -4.83, -4.749, -2.312, -0.087, -0.115, 0.877, 2.055, 0.71, 0.474, 0.08};

```

```

In[358]:= Cda0 = Interpolation[Transpose[{c2, Cda0}], InterpolationOrder → 1];
Cda1 = Interpolation[Transpose[{c2, Cda1}], InterpolationOrder → 1];
Cda2 = Interpolation[Transpose[{c2, Cda2}], InterpolationOrder → 1];
Cda3 = Interpolation[Transpose[{c2, Cda3}], InterpolationOrder → 1];
Cda4 = Interpolation[Transpose[{c2, Cda4}], InterpolationOrder → 1];
Cda5 = Interpolation[Transpose[{c2, Cda5}], InterpolationOrder → 1];

```

```

In[364]:= Cd[c1_, c2_] := Cda0[c2] + Cda1[c2] c1 + Cda2[c2] c1^2 + Cda3[c2] c1^3 + Cda4[c2] c1^4 + Cda5[c2] c1^5;

```

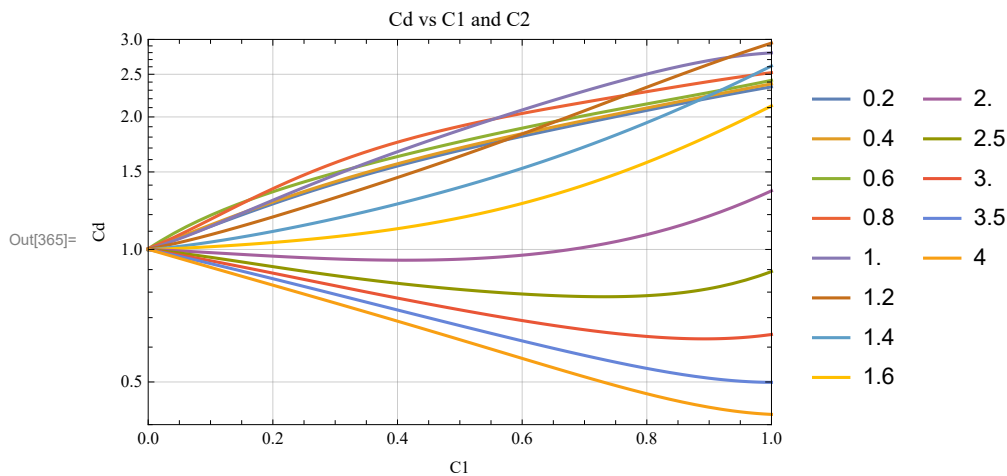
▣ **plot Cf data**

Close match to Figure 4.19.14:

```

In[365]:= LogPlot[{Cd[x, c2[[1]]], Cd[x, c2[[2]]], Cd[x, c2[[3]]], Cd[x, c2[[4]]],
  Cd[x, c2[[5]]], Cd[x, c2[[6]]], Cd[x, c2[[7]]], Cd[x, c2[[8]]], Cd[x, c2[[9]]],
  Cd[x, c2[[10]]], Cd[x, c2[[11]]], Cd[x, c2[[12]]], Cd[x, c2[[13]]]}, {x, 0, 1},
  PlotRange → {{0, 1}, {0.4, 3}}, Axes → False, Frame → True, GridLines → Automatic,
  FrameLabel → {"C1", "Cd"}, PlotLabel → "Cd vs C1 and C2", PlotLegends → c2]

```

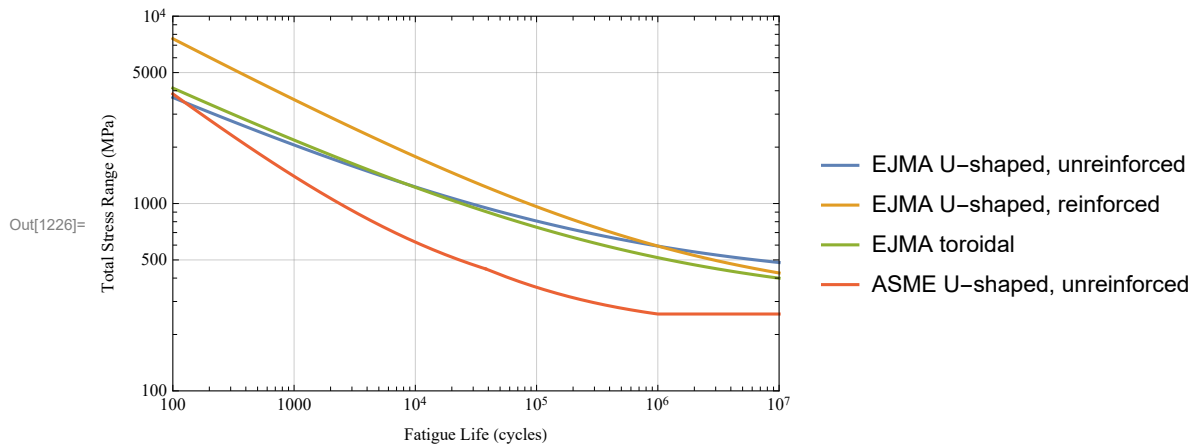


■ fatigue curves

```

In[1219]:= Ncu[St_] := (1.86 × 10^6 / (St - 54000))^3.4;
Stu[Nc_] := (1.86 × 10^6 / Nc^(1/3.4) + 54000) 0.00689476; (* MPa, U-shaped, unreinforced *)
Str[Nc_] := (5.18 × 10^6 / Nc^(1/2.9) + 41800) 0.00689476; (* MPa, U-shaped, reinforced *)
Stt[Nc_] := (2.30 × 10^6 / Nc^(1/3.25) + 41800) 0.00689476; (* MPa, toroidal *)
Sta[Nc_] := Module[{St1, St2, St3}, St1 = 35850 / Nc^(1/2) + 264;
  St2 = 46200 / Nc^(1/2) + 211;
  If[St1 > 448, St1, If[St2 > 257.2, St2, 257.2]]]; (* MPa, U-shaped, unreinforced *)
Ncmin = 10^2;
Ncmax = 10^7;
plt1 = LogLogPlot[{Stu[x], Str[x], Stt[x], Sta[x]},
  {x, Ncmin, Ncmax}, PlotRange → {{Ncmin, Ncmax}, {10^2, 10^4}}, Frame → True,
  FrameLabel → {"Fatigue Life (cycles)", "Total Stress Range (MPa)"},
  GridLines → Automatic, PlotLegends → {"EJMA U-shaped, unreinforced",
    "EJMA U-shaped, reinforced", "EJMA toroidal", "ASME U-shaped, unreinforced"}]

```



```

In[1228]:= Export["fatigue.png", plt1, ImageSize → 8 * 300, ImageResolution → 300];

```

```

In[1229]:= SystemOpen[DirectoryName[AbsoluteFileName["fatigue.png"]]]

```

■ LIGO Bellows

compare to CBI calculation per EJMA from LIGO-E950010-v1

■ input parameters

```
In[1252]:= Db = 48.750 × 0.0254; (* m *)
Dc = 0;
Ds = Db;
Eb = 28.3 × 10^6 × 6894.76; (* Pa *)
E0 = Eb;
Ec = 0;
Es = Eb;
Kg = 1;
Lc = 0;
Lt = 6 × 0.0254;
n = 1;
Nc = 6;
Nspe = 50 × 365;
P = 14.7 × 6894.76; (* Pa *)
q = 2.5 × 0.0254;
S = 115 × 10^6; (* Pa *)
Ss = S;
t = 0.105 × 0.0254;
tc = 0;
ts = 0.127 × 0.0254;
w = 3.020 × 0.0254;
xe = 1.210 × 0.0254;
xc = 0.430 × 0.0254;
y = 0.125 × 0.0254;
θ = 0.300 (Pi/180); (* rad *)
```

■ calculated parameters

```
In[1338]:= k = Min[Lt / (1.5 Sqrt[Db t]), 1.0];
Ls = Sqrt[(Ds + ts) ts] / 3;
Dm = Db + w + n t;
tp = t Sqrt[Db / Dm];
ri = (q - 2 t) / 4;
rm = ri + n t / 2;
A = n tp (2 Pi rm + 2 Sqrt[(q / 2 - 2 rm) ^ 2 + (w - 2 rm) ^ 2]);
Kf = 1.0;
ef = Sqrt[(Log[1 + 2 w / Db]) ^ 2 + (Log[1 + n tp / (2 rm)]) ^ 2]; (* formed 100% to outside of Db *)
Ysmi = 1 + 9.94 (Kf ef) - 7.59 (Kf ef) ^ 2 - 2.4 (Kf ef) ^ 3 + 2.21 (Kf ef) ^ 4;
(* austenitic stainless steel *)
Ysm = Max[Min[Ysmi, 2], 1];
Km = 1.5 Ysm;
C1 = 2 rm / w;
C2 = 1.82 rm / Sqrt[Dm tp];
Cp = If[C1 ≤ 0.3, ACp[C1, C2], BCp[C1, C2]];
Δqxe = xe / Nc;
Δqxc = xc / Nc;
Δqye = 3 Dm y / (Nc (Nc q + xe));
Δqyc = 3 Dm y / (Nc (Nc q + xc));
Δqθ = Dm θ / (2 Nc);
Δq = Δqxe + Δqxc + Δqye + Δqθ;
```

■ stresses due to pressure P

```
In[547]:= S1 = P (Db + n t) ^ 2 Lt Eb k / (2 (n t (Db + n t) Lt Eb + tc Dc Lc Ec k)) (* Pa *)
S1 / 6895 (* psi *)
S1 < S

Out[547]= 2.3579 × 107

Out[548]= 3419.73

Out[549]= True

In[37]:= S1ppp = P (Ds + ts) ^ 2 (Ls + q / 2) Es / (2 (n t (Db + n t) Lt Eb + ts (Ds + ts) Ls Es))
S1ppp < Ss

Out[37]= 7.00935 × 106

Out[38]= True

In[550]:= S2I = P q Dm / (2 A) (* Pa *)
S2I / 6895 (* psi *)
S2I < S

Out[550]= 8.64692 × 106

Out[551]= 1254.09

Out[552]= True
```

```
In[558]:= S3 = P w / (2 n tp) (* Pa *)
          S3 / 6895 (* psi *)
          S4 = (P Cp / (2 n)) (w / tp) ^ 2 (* Pa *)
          S4 / 6895 (* psi *)
          S3 + S4 < Km S
```

```
Out[558]= 1.50354 × 106
```

```
Out[559]= 218.063
```

```
Out[560]= 3.12502 × 107
```

```
Out[561]= 4532.29
```

```
Out[562]= True
```

■ stresses due to cyclic deflection

part 4.19.5.7

```
In[563]:= S5 = Eb tp ^ 2 Δq / (2 w ^ 3 Cf [C1, C2]) (* Pa *)
          S5 / 6895 (* psi *)
```

```
Out[563]= 1.1639 × 107
```

```
Out[564]= 1688.04
```

```
In[565]:= S6 = 5 Eb tp Δq / (3 w ^ 2 Cd [C1, C2]) (* Pa *)
          S6 / 6895 (* psi *)
```

```
Out[565]= 1.11619 × 109
```

```
Out[566]= 161884.
```

```
In[567]:= St = 0.7 (S3 + S4) + (S5 + S6) (* Pa *)
          St / 6895 (* psi *)
```

```
Out[567]= 1.15076 × 109
```

```
Out[568]= 166897.
```

■ Fatigue life per ASME BPVC.VIII.2-2023

□ for CBI cyclic displacement estimate

part 4.19.5.7 and Table 4.19.6, operating conditions, 100F

The notes on Table 4.19.6 in ASME BPVC.VIII.2-2023, are as follows:

“(a) In the above equations, K_g is the fatigue strength reduction factor that accounts for geometrical stress concentration factors due to thickness variations, weld geometries, surface notches, and other surface or environmental conditions. The range for K_g is $1.0 \leq K_g \leq 4.0$ with its minimum value for smooth geometrical shapes and its maximum for 90 deg welded corners and fillet welds. Fatigue strength reduction factors may be determined from theoretical, experimental, or photoelastic studies. A factor has already been included in the above equations for N to account for normal effects of size, environment, and surface finish. If the expansion bellows does not have circumferential welds and satisfies all of the design and examination requirements of this paragraph, a $K_g = 1.0$ may be used.

(b) The allowable number of cycles given in this table includes a reasonable design margin (2.6 on cycles and 1.25 on stress) and represents the maximum number of cycles for the operating condition considered. Therefore an additional design margin should not be applied. ...”

```

In[662]:= Ste = Kg St (E0 / Eb) / 10^6 (* MPa *)
If[Ste > 448, Nalw = (35 850 / (Ste - 264)) ^ 2, \
  If[Ste > 257.2, Nalw = (46 200 / (Ste - 211)) ^ 2, \
    Nalw = 10^6]];
Nalw (* cycles *)
Nalw / 365 (* yrs *)

```

Out[662]= 1150.76

Out[664]= 1634.44

Out[665]= 4.47791

□ **for more realistic diurnal cyclic displacement estimate**

The diurnal temperature fluctuations to which the LIGO LLO BT is subjected are ~25C (see section 6.1.1.2 of T2000475-v1). This results in just 17.2 mm tube expansion (bellows compression), as compared to the 41.7 mm total axial displacement assumed by CBI. The result is a total equivalent displacement of ~70% of the value used by CBI:

```

In[685]:= CTE = 17.2 × 10^-6; (* m/m/C *)
ΔT = 25;
L = 40;
xe2 = 0;
xc2 = CTE ΔT L;
Δqxe2 = xe2 / Nc;
Δqxc2 = xc2 / Nc;
Δqye2 = 3 Dm y / (Nc (Nc q + xe2));
Δqyc2 = 3 Dm y / (Nc (Nc q + xc2));
Δqθ2 = Dm θ / (2 Nc);
Δq2 = Δqxe2 + Δqxc2 + Δqye2 + Δqθ2
Δq2 / Δq

```

Out[695]= 0.00893169

Out[696]= 0.708984

Scale the stresses S5 and S6 by the ratio of revised displacement to previously assumed displacement:

```

In[703]:= St = (0.7 (S3 + S4) + (S5 + S6) (Δq2 / Δq)); (* Pa *)
Ste = Kg St (E0 / Eb) / 10^6 (* MPa *)
If[Ste > 448, Nalw = (35 850 / (Ste - 264)) ^ 2, \
  If[Ste > 257.2, Nalw = (46 200 / (Ste - 211)) ^ 2, \
    Nalw = 10^6]];
Nalw (* cycles *)
Nalw / 365 (* yrs *)

```

Out[704]= 822.541

Out[706]= 4119.73

Out[707]= 11.2869

If the cycle margin of 2.6 is factored out:

```

In[709]:= Nalw 2.6 (* cycles *)
Nalw 2.6 / 365 (* yrs *)

```

Out[709]= 10 711.3

Out[710]= 29.346

■ Fatigue life per EJMA-1993

E950010-00-B, pg. 2, operating conditions, 100F

```
In[659]:= Ste = 167.742; (* ksi *)
          Ste 6.89476 (* MPa *)
          Nalw = (1860 / (Ste - 54)) ^ 3.4 (* cycles *)
          Nalw / 365 (* yrs *)

Out[659]= 1156.54

Out[660]= 13 372.5

Out[661]= 36.637
```

■ GEO600 Convoluted Beam Tubes

as an example of a corrugated tube which has been in service for > 20 yrs, consider the GEO600 beam tube

■ input parameters

```
In[723]:= Db = 0.6; (* m *)
          Dc = 0;
          Ds = Db;
          Eb = 28.3 × 10^6 × 6894.76; (* Pa *)
          E0 = Eb;
          Ec = 0;
          Es = Eb;
          Kg = 1;
          Lc = 0;
          Lt = 6 × 0.0254;
          n = 1;
          Nspe = 50 × 365;
          P = 14.7 × 6894.76; (* Pa *)
          q = 0.030;
          S = 115 × 10^6; (* Pa *)
          Ss = S;
          t = 0.0009;
          tc = 0;
          ts = t;
          w = 0.017;
```

calculate diurnal compression displacement

```
In[752]:= CTE = 17.2 × 10-6; (* m/m/C *)
ΔT = 25;
L = 40;
Nc = L / q;
y = 0;
θ = 0; (* rad *)
xe = 0;
xc = CTE ΔT L;
Δqxe = xe / Nc;
Δqxc = xc / Nc;
Δqye = 3 Dm y / (Nc (Nc q + xe));
Δqyc = 3 Dm y / (Nc (Nc q + xc));
Δqθ = Dm θ / (2 Nc);
Δq = Δqxe + Δqxc + Δqye + Δqθ
```

```
Out[763]= 0.0000129
```

■ calculated parameters

```
In[764]:= k = Min[Lt / (1.5 Sqrt[Db t]), 1.0];
Ls = Sqrt[(Ds + ts) ts] / 3;
Dm = Db + w + n t;
tp = t Sqrt[Db / Dm];
ri = (q - 2 t) / 4;
rm = ri + n t / 2;
A = n tp (2 Pi rm + 2 Sqrt[(q / 2 - 2 rm)2 + (w - 2 rm)2]);
Kf = 1.0;
ef = Sqrt[(Log[1 + 2 w / Db])2 + (Log[1 + n tp / (2 rm)])2]; (* formed 100% to outside of Db *)
Ysmi = 1 + 9.94 (Kf ef) - 7.59 (Kf ef)2 - 2.4 (Kf ef)3 + 2.21 (Kf ef)4;
(* austenitic stainless steel *)
Ysm = Max[Min[Ysmi, 2], 1];
Km = 1.5 Ysm;
C1 = 2 rm / w;
C2 = 1.82 rm / Sqrt[Dm tp];
Cp = If[C1 ≤ 0.3, ACp[C1, C2], BCp[C1, C2]];
```

■ stresses due to pressure P

```
In[780]:= S1 = P (Db + n t)2 Lt Eb k / (2 (n t (Db + n t) Lt Eb + tc Dc Lc Ec k)) (* Pa *)
S1 / 6895 (* psi *)
S1 < S
```

```
Out[780]= 3.3835 × 107
```

```
Out[781]= 4907.18
```

```
Out[782]= True
```

```
In[783]:= S1ppp = P (Ds + ts)2 (Ls + q / 2) Es / (2 (n t (Db + n t) Lt Eb + ts (Ds + ts) Ls Es))
S1ppp < Ss
```

```
Out[783]= 4.80673 × 106
```

```
Out[784]= True
```

```
In[785]:= S2I = P q Dm / (2 A) (* Pa *)
          S2I / 6895 (* psi *)
          S2I < S
```

```
Out[785]= 2.07187 × 107
```

```
Out[786]= 3004.89
```

```
Out[787]= True
```

```
In[788]:= S3 = P w / (2 n tp) (* Pa *)
          S3 / 6895 (* psi *)
          S4 = (P Cp / (2 n)) (w / tp) ^ 2 (* Pa *)
          S4 / 6895 (* psi *)
          S3 + S4 < Km S
```

```
Out[788]= 971 396.
```

```
Out[789]= 140.884
```

```
Out[790]= 1.05153 × 107
```

```
Out[791]= 1525.06
```

```
Out[792]= True
```

■ stresses due to cyclic deflection

part 4.19.5.7

```
In[793]:= S5 = Eb tp ^ 2 Δq / (2 w ^ 3 Cf [C1, C2]) (* Pa *)
          S5 / 6895 (* psi *)
```

```
Out[793]= 95 711.5
```

```
Out[794]= 13.8813
```

```
In[795]:= S6 = 5 Eb tp Δq / (3 w ^ 2 Cd [C1, C2]) (* Pa *)
          S6 / 6895 (* psi *)
```

```
Out[795]= 5.73304 × 106
```

```
Out[796]= 831.478
```

```
In[797]:= St = 0.7 (S3 + S4) + (S5 + S6) (* Pa *)
          St / 6895 (* psi *)
```

```
Out[797]= 1.38694 × 107
```

```
Out[798]= 2011.52
```

■ Fatigue life per ASME BPVC.VIII.2-2023

part 4.19.5.7 and Table 4.19.6, operating conditions, 100F

The notes on Table 4.19.6 in ASME BPVC.VIII.2-2023, are as follows:

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(b) The allowable number of cycles given in this table includes a reasonable design margin (2.6 on cycles and 1.25 on stress) and represents the maximum number of cycles for the operating condition considered. Therefore an additional design margin should not be applied. ...”

```

In[803]:= Ste = Kg St (E0 / Eb) / 10^6 (* MPa *)
If[Ste > 448, Nalw = (35 850 / (Ste - 264)) ^ 2, \
  If[Ste > 257.2, Nalw = (46 200 / (Ste - 211)) ^ 2, \
    Nalw = 10^6]];
Nalw // N(* cycles *)
Nalw / 365 // N(* yrs *)

Out[803]= 13.8694

Out[805]= 1. × 106

Out[806]= 2739.73

```

SCC

```

In[901]:= time = 3 × 10^-3 / 10^-11 (* sec *)
time / (60 × 60 × 24 × 365) // N(* yrs *)

Out[901]= 300 000 000

Out[902]= 9.51294

extrapolating to K = 1 MPa

In[906]:= time = 3 × 10^-3 / (2 × 10^-12) (* sec *)
time / (60 × 60 × 24 × 365) // N(* yrs *)

Out[906]= 1 500 000 000

Out[907]= 47.5647

M = 2.897 × 10^-2; (* air, kg/mol *)
μ = 1.831 × 10^-5; (* air viscosity, Pa·s *)
P1 = 101 325; (* atm pressure, Pa *)
P2 = 0;
Mdot = Pi M R^4 (P1^2 - P2^2) / (16 μ L R T);

https://www.nist.gov/pml/sensor-science/thermodynamic-metrology/unit-conversions

In[905]:= Q = 10^-6; (* Torr-L/s *)
Q 0.01270903 (* sccm *)

Out[905]= 1.2709 × 10^-8

In[914]:= 5 / Sqrt[Pi 3 × 10^-6] // N

Out[914]= 1628.68

```

■ Molecular Slip Flow

M.R.Prisco, et. al., Predictions of Vacuum Loss of Evacuated Vials from Initial Air Leak Rates, Journal of Pharmaceutical Sciences, volume 102, issue 8, Aug 2013, pp. 2730-2737.

<https://doi.org/10.1002/jps.23652>

```

In[1029]:= P1 = 101325; (* atm pressure, Pa *)
P2 = 10*133.3224;
R = 287.09; (* m^2/s^2/K *)
TC = 22.2;
T = TC + 273.15;
TF = TC (9/5) + 32;
 $\mu_{\text{cgs}} = 2.7104 \times 10^{-7} \text{ TF} + 1.6270 \times 10^{-4};$  (* dyne-s/cm^2 *)
 $\mu = \mu_{\text{cgs}} 0.1;$  (* Pa-s *)
f1 = 0.9;
L = 3.625*10^-3;
Clear[Q];
Q[d_] := (P1) ((Pi d^4 / (256  $\mu$  L)) ((P1^2 - P2^2) / P1) +
(Pi d^3 / (4 L)) ((P1 - P2) / P1) Sqrt[Pi R T / 32] ((2 - f1) / f1));

In[1063]:= Qtsccm = {10^-8, 10^-7, 10^-6, 10^-5, 10^-4, 10^-3, 10^-2, 10^-1, 1, 10, 100, 1000}; (* sccm *)
Qt = Qtsccm 0.001689; (* Pa m^3/s *)
dia = Table[0, Length[Qt]];
Do[dia[[i]] = d 10^6 /. NSolve[Q[d] == Qt[[i]], d, PositiveReals][[1]], {i, 1, Length[Qt]}]
dia

Out[1065]= {0.182811, 0.377641, 0.758963, 1.47772, 2.79391,
5.16403, 9.39896, 16.9425, 30.3647, 54.2376, 96.6927, 172.191}

In[1066]:= P2 = 133.3224*10^-9; (* Pa *)
L = 3.2*10^-3;
Qtorr = {10^-9, 10^-8, 10^-7, 10^-6}; (* Torr-L/s *)
Qt = Qtorr 0.133322; (* Pa m^3/s *)
dia = Table[0, Length[Qt]];
Do[dia[[i]] = d 10^6 /. NSolve[Q[d] == Qt[[i]], d, PositiveReals][[1]], {i, 1, Length[Qt]}]
dia

Out[1071]= {0.33646, 0.679878, 1.33132, 2.52936}

In[1374]:= K1 = 1151 Sqrt[Pi dia[[1]] 10^-6]

Out[1374]= 1.18336

```