# VOPO Suspension Final Design 

LIGO MIT Lab

Matichard, Fabrice
Fernandez Galiana, Alvaro

## VOPO Suspension

- Design based on HAM 6
- Holes availability (Dog Clamps)
- "Shooting" Area (for the outcome beam)
- In Vacuum RFPD Area
- Space for Tip Tilt


## Maximizing Bench Surface: $3.64 \mathrm{ft}^{2}$



## VOPO Suspension

- Blade with 3 -axial symmetry
- Blades clamps form equilateral triangle
- Triangle center close to "Geometric Center" (easy to balance)
- Stops separated enough



## VOPO Suspension

- Conclusions:
- Questions:
- Is it enough space for the Tip Tilt?
- Do we need a Tip Tilt?
- Is it enough shooting space?


## Optical Layout

- 2 different layouts:
- O2 Squeezer model (early squeezing)

- O3 Squeezer model with filter cavity + RFPD in Vacuum



## Optical Layout

- Solid Works Optical Layout
- Mirror:
- Mass: 205.87 g
- Quantity: 11 / 18

- Lens:
- Mass: 121.375 g
- Quantity: 7 / 9



## Optical Layout

- Solid Works Optical Layout
- Beam Dump:
- Mass: 164.61 g
- Quantity: 2 / 5

- QPD:
- Mass: 142.98 g
- Quantity: 2 / 3



## Optical Layout

## - Solid Works Optical Layout

- Polarizers:
- Mass: 206.71 g
- Quantity: 2+2 / 2+2


NOTE: At this point the polarizers have been modeled as mirrors. A more realistic model has to be done to the two kinds of polarizers that will be used.

- Wave Plate:
- Mass: 299.69 g
- Quantity: 1 / 2



## Optical Layout

## - Solid Works Optical Layout

- Fibers In:
- Mass: 252.01 g
- Quantity: 2 / 2


NOTE: At this point the fiber In have been modeled as mirrors with pico motors. A more realistic model has to be done to the two kinds of polarizers that will be used.

- RFPD in Vacuum
- Mass: 270.41 g
- Quantity: 0 / 2


NOTE: At this point the model used is for a 4 " beam height while the real one will be at $2.5 "$

## Optical Layout

- Solid Works Optical Layout
- Faraday Rotator:
- Mass:
- Base: 170.43 g
- Rotator: X g
- Quantity: 1 / 1


NOTE: At this point the Faraday Rotator has been modeled as just its base and with a rough design that will be improved once the Faraday rotator model will be obtained.

EOT Faraday
Rotator


Base (6061-T6 Al)

## Optical Layout

- Solid Works Optical Layout

- VOPO Cavity:
- Mass:: 1463 g
- Quantity: 1 / 1


## Optical Layout

- O3 (Filter Cavity + In Vacuum RFPD)



## Optical Layout

- O3 (Filter Cavity + In Vacuum RFPD)
- Option 1
- Problems



## Optical Layout

- O3 (Filter Cavity + In Vacuum RFPD)
- Option 1
- Problems



## Optical Layout

- O3 (Filter Cavity + In Vacuum RFPD)



## Optical Layout

- O3 (Filter Cavity + In Vacuum RFPD)
- Option 2



## Optical Layout



## Optical Layout

- CONCLUSIONS
- Injection Bench is large enough
-QUESTIONS


## Injection Bench

- Stiffener first design (dummy)



## Injection Bench

- Stiffener first design (dummy II)
- $\mathrm{f}=276.29 \mathrm{~Hz}$
- Mass: 15.977 kg



## Injection Bench: Mass Budget

## - O3 (Filter Cavity + RFPD)

- Optics: 12 kg (2.1kg contingency)
- Bench: 18 kg
- Balancing Mass: 6kg



## SUSPENDED MASS: 36 kg

NOTE I: The weight of the Injection Bench has been set at 18 kg but it may vary after the frequency FEA

NOTE II: In this calculation the Faraday Rotator is not considered, just its base, and the model for polarizers and fiber in is set as the one of a mirror mount

## MASS BUDGET

| ELEMENT TYPE | Name | Description | Unit Weight <br> (g) | Quantity | Total mass <br> (g) | Final Design? | Mass Checked? |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| OPTICS | MIRROR | With Beam Dump | 205.87 | 7 | 1441.09 |  | YES |
| OPTICS | MIRROR | With Beam Dump 2 | 207.55 | 6 | 1245.30 |  | no |
| OPTICS | MIRROR | Without Beam Dump | 192.90 | 3 | 578.70 |  | No |
| OPTICS | MIRRORII | Lens | 121.38 | 1 | 121.38 |  | No |
| OPTICS | MIRRORIII | Lens Sigle Base | 97.07 | 1 | 97.07 |  | no |
| OPTICS | Lens | Desc 2 | 121.38 | 9 | 1092.38 |  | no |
| OPTICS | beam dump | Without One Black Glass | 164.61 | 5 | 823.05 |  | No |
| OPTICS | QPD | Desc 5 | 142.98 | 3 | 428.94 |  | no |
| OPTICS | polarizer | Desc 6 | 203.05 | 4 | 812.19 |  | no |
| OPTICS | WAVE PLATE | Desc 7 | 299.69 | 2 | 599.38 |  | No |
| OPTICS | RFPD in Vacuum | Desc 8 | 270.41 | 2 | 540.82 |  | no |
| OPTICS | Fiber in | Desc 9 | 252.01 | 2 | 504.02 |  | no |
| OPTICS | faraday rotator | Desc 10 | 170.43 | 1 | 170.43 |  | No |
| OPTICS | vopo cavity | Desc 11 | 1463.27 | 1 | 1463.27 |  | no |
| SUSPENSION | injection bench | Desc 12 | 17797.97 | 1 | 17797.97 |  | no |
| SUSPENSION | LIMITERS \& CLAMPS | Desc 13 | 193.50 | 1 | 193.50 |  | no |
| MASS | BALANCE MASS 1 | Lateral | 3857.36 | 0 | 0.00 |  | no |
| MASS | BALANCE MASS 2 | Lateral Removable | 1302.18 | 0 | 0.00 |  | no |
| MASS | BALANCE MASS 3 | On Bench | 2631.72 | 0 | 0.00 |  | no |
| MASS | SCREWS | On Bench | 174.88 | 0 | 0.00 |  | No |
|  |  |  |  |  |  |  |  |
| OPTICS |  |  |  |  | 9.918 | kg |  |
| SUSPENSION |  |  |  |  | 17.991 | kg |  |
| MASS |  |  |  |  | 0.000 | kg |  |
|  |  |  |  |  |  |  |  |
| TOTAL WEIGHT |  |  |  |  | 27.909 | kg |  |
| Mass to 36 kg |  |  |  |  | 8.09 | kg |  |

## Injection Bench: Mass Budget

## - O2

- Optics: 8 kg (1.2kg contingency)
- Bench: 18 kg
- Balancing Mass: 10 kg


## SUSPENDED MASS: 36 kg

NOTE I: The weight of the Injection Bench has been set at 18 kg but it may vary after the frequency FEA

NOTE II: In this calculation the Faraday Rotator is not considered, just its base, and the model for polarizers and fiber in is set as the one of a mirror mount

## MASS BUDGET

| ELEMENT TYPE | Name | Description | Unit Weight <br> (g) | Quantity | Total mass <br> (g) | Final Design? | Mass Checked? |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| OPTICS | MIRROR | With Beam Dump | 205.87 | 5 | 1029.35 |  | YES |
| OPTICS | MIRROR | With Beam Dump 2 | 207.55 | 3 | 622.65 |  | NO |
| OPTICS | MIRROR | Without Beam Dump | 192.90 | 2 | 385.80 |  | No |
| OPTICS | MIRROR III | Lens Sigle Base | 97.07 | 1 | 97.07 |  | No |
| OPTICS | LENS | Desc 2 | 121.38 | 7 | 849.63 |  | No |
| OPTICS | BEAM DUMP | Without One Black Glass | 164.61 | 2 | 329.22 |  | No |
| OPTICS | QPD | Desc 5 | 142.98 | 2 | 285.96 |  | No |
| OPTICS | polarizer | Desc 6 | 206.71 | 4 | 826.84 |  | No |
| OPTICS | WAVE PLATE | Desc 7 | 299.69 | 1 | 299.69 |  | No |
| OPTICS | RFPD in Vacuum | Desc 8 | 270.41 | 0 | 0.00 |  | No |
| OPTICS | FIBER IN | Desc 9 | 195.18 | 2 | 390.35 |  | No |
| OPTICS | FARADAY ROTATOR | Desc 10 | 170.43 | 1 | 170.43 |  | No |
| OPTICS | VOPO CAVITY | Desc 11 | 1463.27 | 1 | 1463.27 |  | NO |
| SUSPENSION | INJECTION BENCH | Desc 12 | 18000.00 | 1 | 18000.00 |  | No |
| SUSPENSION | LIMITERS \& CLAMPS | Desc 13 | 100.92 | 1 | 100.92 |  | No |
| MASS | BALANCE MASS 1 | Lateral | 0.00 | 1 | 0.00 |  | No |
| MASS | BALANCE MASS 2 | On Bench | 0.00 | 1 | 0.00 |  | No |
|  |  |  |  |  |  |  |  |
| OPTICS |  |  |  |  | 6.750 | kg |  |
| SUSPENSION |  |  |  |  | 18.101 | kg |  |
| MASS |  |  |  |  | 0.000 | kg |  |
|  |  |  |  |  |  |  |  |
| TOTAL WEIGHT |  |  |  |  | 24.851 | kg |  |
| Mass to $\mathbf{3 6} \mathbf{~ k g}$ |  |  |  |  | 11.15 | kg |  |

## Injection Bench Assembly FEA

- Optical components for FEA
- Mirror:
- Mass: 205.87 g
- Mass FEA: 203 g
- Quantity: 11 / 18


- Lens:
- Mass: 121.375 g
- Mass FEA: 120 g
- Quantity: 7 / 10



## Injection Bench Assembly FEA

- Optical components for FEA
- Beam Dump:
- Mass: 164.61 g
- Mass to FEA: 163 g
- Quantity: 2 / 5


- QPD:
- Mass: 142.98 g
- Mass FEA: 143 g
- Quantity: 2 / 3



## Injection Bench Assembly FEA

- Optical components for FEA
- Polarizer:
- Mass: 206.71 g
- Mass FEA: 203 g
- Quantity: 2+2 / 2+2

- Wave Plate:
- Mass: 300 g
- Mass FEA: 297 g
- Quantity: 1 / 2



## Injection Bench Assembly FEA

- Optical components for FEA
- Fiber In:
- Mass: 252.01 g
- Mass FEA: 191 g
- Quantitv: 2 / 2

- RFPD in Vacuum
- Mass: 270.41 g
- Mass: 279 g
- Quantity: 0 / 2




## Injection Bench Assembly FEA

- Optical components for FEA
- Faraday Rotator:
- Mass:
- Base: 170.43 g
- Base FEA: 170 g
- Rotator: X g
- Quantity: 1 / 1

- VOPO Cavity:
- Mass: 1463 g
- Mass: 1460 g
- Quantity: 1 / 1



## Injection Bench Assembly FEA

- Injection Bench Assembly for FEA
- Global Contact: Bonded


## Injection Bench Assembly FEA

- Injection Bench Assembly for FEA (no MASSES)


NOTE: No significant difference has been observed between compatible and incompatible meshing

## BLADE DESIGN: Material

- Material: 440C SSTL ( $\mathrm{E}=210 \mathrm{GPa}, \sigma_{\mathrm{y}}=1800 \mathrm{MPa}, \mathrm{UTS}=1970 \mathrm{MPa}, \rho=7800 \mathrm{~kg} / \mathrm{m}^{3}, \mathrm{v}=0.285$ )
- Total Suspended Mass: m $=36 \mathrm{~kg}$
- Charge per blade: $\mathrm{P}=117.72 \mathrm{~N}(12 \mathrm{~kg})$

mechanical Properties
- Constraints:
- Factor of Safety: $\geq 33.3 \%(\mathrm{FoS} \geq 3)$
- Desired frequency: $\mathrm{f} \approx 1.5 \mathrm{~Hz}$

| Tempering Temperature <br> ('C) | Tensile Strength (MPa) | Yield strength $0.2 \%$ Proof (MPa) | Elongation (\% in 50mm) | Hardness Rockwell (HR C) | Impact charpy V (1) |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Annealed* | 758 | 448 | 14 | 269HB max ${ }^{\text {a }}$ | - |
| 204 | 2030 | 1900 | 4 | 59 | 9 |
| 260 | 1960 | 1830 | 4 | 57 | 9 |
| 316 | 1860 | 1740 | 4 | 56 | 9 |
| 371 | 1790 | 1660 | 4 | 56 | 9 |

## BLADE DESIGN: Dimensions

- Design Parameters:
- Blade Width: a
- Blade Length: 1
- Blade thickness: h


3 Parameters

- Equations (for triangular blades):
- $K z Z=\frac{E a h^{3}}{4 l^{3}}$
- $f=\frac{1}{2 \pi} \sqrt{\frac{K_{Z Z}}{m}}=1.5 \mathrm{~Hz}$
- $\sigma_{\max }=\frac{6 P l}{a h^{2}}$

2 Equations


NOTE: The choice of this length has been made after checking the optical layout

## BLADE DESIGN: Dimensions

- Theoretical Results (BladeDesign.m) for $m=36 \mathrm{~kg}, \mathrm{f}=1.5 \mathrm{~Hz}$ and $\mathrm{FoS}=3$ :
- $\mathrm{l}=280 \mathrm{~mm}$
- $\mathrm{b}=80.12 \mathrm{~mm}$
- $\mathrm{h}=2.0282 \mathrm{~mm}$
- $\mathrm{Kzz}=1065.9 \mathrm{~N} / \mathrm{m}$
- Tip deflection $=110.4 \mathrm{~mm}$

NOTE: 80 mm is less than what was previously used for the blade design ( 85 mm ) so it fits widely in the designed bench.


## BLADE DESIGN: Dimensions

- Theoretical Results (BladeDesign.m) for $\mathrm{m}=36 \mathrm{~kg}, \mathrm{f}=1.6 \mathrm{~Hz}$
- $\mathrm{l}=280 \mathrm{~mm}$
- $\mathrm{b}=85 \mathrm{~mm}$
- $\mathrm{h}=2.0761 \mathrm{~mm}$
- $\operatorname{FoS}=3.3346$ (30\%)
- $\mathrm{Kzz}=1212.8 \mathrm{~N} / \mathrm{m}$

NOTE: In order to increase the FoS and knowing that 280 mm length and 85 mm wide blade fits in the design, the desired frequency has been raised to 1.6 Hz .


## BLADE DESIGN: Dimensions

- FINAL DIMENSIONS:
- $\mathrm{f}=1.5252 \mathrm{~Hz}$

- $\mathrm{f}=1.6277 \mathrm{~Hz}$
- $\mathrm{Kzz}=1255.1 \mathrm{~N} / \mathrm{m}$
- $\mathrm{FoS}=3.5473$

NOTE: The expected performances are those shown for 36 kg (that already include some contingency). However, the suspended mass could be raised up to 41 kg keeping $\mathrm{FoS} \geq 3$

## BLADE DESIGN: FEA

- First Analysis using SW Simulation
- Second Analysis using ANSYS
- $\mathrm{m}=36 \mathrm{~kg}$
- Force: Normal to Surface



## BLADE DESIGN: FEA

- Deformation


SW SIMULATION
ANSYS

## BLADE DESIGN: FEA

Stress (von Misses)

| 5.4768e8 Max |
| :---: |
| 4.868688 |
| 4.2604 e8 |
| 3.6522e8 |
| 3.044 e 8 |
| 2.4358 e 8 |
| $1.8276{ }^{\text {e }}$ |
| 1.2194 e 8 |
| 6.112 e 7 |
| 3.0088e5 Min |

Difference= $0.40 \%$

SW SIMULATION
ANSYS

## BLADE DESIGN: FEA

- Strain (Equivalent)


SW SIMULATION

| 0.002608 Max <br> 0.0023184 <br> 0.0020288 <br> 0.0017392 <br> 0.0014495 <br> 0.0011599 <br> 0.0008703 <br> 0.00058068 <br> 0.00029106 <br> 1.4327e-6 Min |
| :---: |
|  |  |
|  |  |
|  |  |
|  |  |
|  |  |
|  |  |
|  |  |
|  |  |
|  |  |

ANSYS

## BLADE DESIGN: FEA

- Conclusions:


| FEA |  |  |  |  |
| :--- | :---: | :---: | :---: | :---: |
|  | SW | ANSYS | Difference $\%$ |  |
| Stress (MPa) | 547.2 | 547.68 | 0.088 |  |
| Displacement $(\mathrm{mm})$ | 85.82 | 86.16 | 0.395 |  |
| Strain | $1.79 \mathrm{E}-03$ | $2.61 \mathrm{E}-03$ | 31.52 |  |
| FoS | 3.29 | 3.29 |  |  |

## FLEXURE DESIGN: Properties

- Material: Music Wire Steel ( $\mathrm{E}=154-201 \mathrm{GPa}, \sigma_{\mathrm{Y}}=1600-2000 \mathrm{Mpa}$ )
- Total Suspended Mass: m $=36 \mathrm{~kg}$
- Charge per wire: $\mathrm{P}=117.12 \mathrm{~N}$

| Instron 5500R: Tensile Strength Testing, Coiled Wires |  |  |  |  |  |
| :---: | :---: | ---: | ---: | ---: | :---: |
| untreated wires |  |  | cryotreated wires |  |  |
| diameter <br> $(\mathrm{mm})$ | max stress <br> $(\mathrm{Pa})$ | Young's <br> Modulus (Pa) | diameter <br> $(\mathrm{mm})$ | max stress <br> $(\mathrm{Pa})$ | Young's <br> Modulus (Pa) |
| 0.635 | $2.16 \mathrm{E}+09$ | $1.54 \mathrm{E}+11$ | 0.635 | $2.16 \mathrm{E}+09$ | - |
| 0.457 | $2.52 \mathrm{E}+09$ | $2.01 \mathrm{E}+11$ | 0.457 | $2.51 \mathrm{E}+09$ | $2.11 \mathrm{E}+11$ |
| 0.203 | $2.23 \mathrm{E}+09$ | $1.99 \mathrm{E}+11$ | 0.2 | $2.24 \mathrm{E}+09$ | $4.32 \mathrm{E}+11$ |



NOTE: T1500539 shows a $84.6 \%$ average relation between $\sigma_{y}$ and UTS. Therefore, in further calculations $\sigma_{\mathrm{y}}$ is calculated as $80 \%$ of the Minimum Tensile Strength

| D (mm) | 0.1194 | 0.1524 | 0.2007 | 0.2489 | 0.2692 | 0.3404 | 0.3556 | 0.4064 | 0.4572 | 0.6096 | 0.6350 | 0.7112 | 1.0998 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Considerated Yield Strenght (Mpa) | 1986 | 1931 | 1875 | 1820 | 1820 | 1765 | 1765 | 1710 | 1710 | 1600 | 1600 | 1600 | 1600 |
| Average Porportional Yield Strenght (Mpa) |  |  |  | 1930 | 1840 | 1780 | 1770 |  | 1930 | 2190 |  | 1620 | 2170 |

## FLEXURE DESIGN: Music Wire

- Material: Music Wire Steel ( $\mathrm{E}=201 \mathrm{GPa}, \sigma_{\mathrm{Y}}=1710 \mathrm{Mpa}$ )
- Total Suspended Mass: m $=36 \mathrm{~kg}$
- Charge per wire: $\mathrm{P}=117.72 \mathrm{~N}$


NOTE: Diameter of 0.41 mm enables to re-use the Faraday Isolator clamps but $\mathrm{FoS}=1.55$ is too low for our purposes

## FLEXURE DESIGN: Music Wire

- Material: Music Wire Steel ( $\mathrm{E}=201 \mathrm{GPa}, \sigma_{\mathrm{Y}}=1600 \mathrm{Mpa}$ )
- Total Suspended Mass: m $=36 \mathrm{~kg}$
- Charge per wire: $\mathrm{P}=117.72 \mathrm{~N}$

$$
\begin{aligned}
& \mathrm{Lf}=130 \mathrm{~mm} \\
& \cdot \mathrm{~d}=0.61 \mathrm{~mm}
\end{aligned}
$$

-     -         - --- - -

FlexureDesign.m

$$
\begin{aligned}
& \mathrm{f}=1.4203 \mathrm{~Hz} \\
& \cdot \mathrm{Kxx}=955.62 \mathrm{~N} / \mathrm{m} \\
& \cdot \mathrm{FoS}=2.9148
\end{aligned}
$$

NOTE: Diameter of 0.61 mm raise FoS up to almost 3 , which is considered enough as the minimal guaranteed tensile strength has been considered

Desired frequency: $\mathrm{f}<1.5 \mathrm{~Hz}$
$\mathrm{FoS} \geq 33.3 \%(\mathrm{FoS} \geq 3)$
FoS function of $D$ for $L f=130 \mathrm{~mm}$ and $m=36 \mathrm{~kg}$


## FLEXURE DESIGN: Theory

- Frequency calculation using pendulum theory

$$
\begin{gathered}
\ddot{\alpha}+\frac{g}{l z} \sin \alpha=0 \\
\omega^{2}=\frac{g}{l z} \\
f=\frac{1}{2 \pi} \sqrt{\frac{g}{l z}} \\
f_{\text {PEND }}=1.4203 \mathrm{~Hz}
\end{gathered}
$$

## CLAMP DESIGN: Material \& Preload

- SCREW MATERIAL


| Bolt | Hole | Measured <br> Coefficient | Expected <br> coefficient |
| :--- | :--- | :--- | :--- |
| Silver <br> plated | Steel | $0.30-0.31$ | --- |
| Silver <br> plated | Helicoil | $0.26-0.35$ | --- |
| Stee | Helicoil | $0.44-0.52$ | --- |
| Steel | Aluminum | $0.44-0.61$ | $0.61^{*}$ |


| GRADE' | general DESCRIPTION | MECHANICAL REQUIREMEN |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | BOLTS, SCREWS AND STUDS |  |  |  |  |  | NUTS |  |
|  |  | FULL SIZE BOLTS, SCREWS, STUDS |  | MACHINED TEST SPECIMENS OFBOLTS, SCREWS, STUDS |  |  | HARDNESSROCKWELLMin | $\begin{aligned} & \text { PROOF } \\ & \text { STRAD } \\ & \text { STRES } \\ & \text { Ks } \end{aligned}$ | $\begin{gathered} \text { HARDNESS } \\ \text { ROCKWELL } \\ \text { MIn } \end{gathered}$ |
|  |  | $\begin{aligned} & \text { YIELD }{ }^{2} \\ & \text { STRENGTH } \\ & \text { min } \mathrm{ks} \text { i } \end{aligned}$ | TENSILE STRENGTH min ks! <br> min ksi | $\begin{gathered} \text { YIELD } \\ \text { STRENGTH } \\ \text { min ksi } \end{gathered}$ | TENSILE STRENGTH min ksi | $\begin{aligned} & \text { ELON- } \\ & \text { GATON } \\ & \text { GAMIn } \end{aligned}$ |  |  |  |
|  | Austenitic <br> Stainless Steel Sol. Annealod | 30 | 75 | 30 | 75 | 20 | B75 | 75 | B75 |
|  | Austenitic Sold Workteel Cold Worked | 50 | 90 | 45 | 85 | 20 | B85 | 90 | B85 |
|  |  | ${ }_{30}$ | 75 | $\overline{{ }_{30}}$ | ${ }_{75}-$ | $\overline{20}$ | B70 | 75 | $\overline{\text { B70 }}$ |
| $\begin{aligned} & \text { 304.5H } \\ & \text { and } \\ & \text { 3056-SH } \\ & \text { 36-SH } \end{aligned}$ | Austenitic Stainless SteelStrain Hardened | $\begin{gathered} \text { See } \\ \text { Note } 6 \end{gathered}$ | $\begin{gathered} \text { Soe } \\ \text { Note } \end{gathered}$ | $\begin{gathered} \text { See } \\ \text { Note } \end{gathered}$ | $\begin{gathered} \text { See } \\ \text { Note } 6 \end{gathered}$ | 15 | C25 | Note 6 | c20 |
| $\begin{aligned} & 410-\mathrm{H} \\ & 46 \mathrm{H} \end{aligned}$ | $\begin{aligned} & \begin{array}{l} \text { Martensitic } \\ \text { Stainess Steel- } \\ \text { Hardened and } \\ \text { Tempered } \end{array} \\ & \hline \end{aligned}$ | 95 | 125 | 95 | 125 | 20 | C22 | 125 | C22 |
| $\begin{aligned} & 410-\mathrm{HT} \\ & 416-\mathrm{HT} \end{aligned}$ | Matrensitic Stainiess StelTempered | 135 | 180 | 135 | 180 | 12 | C36 | 180 | C36 |
| 430 | Ferritic | 40 | 70 | 40 | 70 | 20 | B75 | 70 | B75 |

- http://www.ssina.com/download a file/fasteners.pdf

```
- \(\sigma_{\mathrm{Y}}=344.74 \mathrm{Mpa}(50 \mathrm{ksi})\)
- \(\mu=0.5\)
```


clampDesign.m FoS $=3$

- Tension $=2290.7 \mathrm{~N}$
- $\mathrm{T}=8.8702 \mathrm{Nm}$ (78.51 in-lb)

Note: No Socket Head Screw for this material neither Silver Plated option (McMaster \& HoloKrome)

Note II: Torque similar to the recommended for Generic Screws in T1100066 (75 in-lb)

## CLAMP DESIGN: Material \& Preload

## - SCREW MATERIAL


http://www.nuttybolts.com/nutty-bolts/?page id=146
http://www.memaster.com/\#socket-head-cap-screws/=z2xdjy


- $\sigma_{\mathrm{y}}=448.16 \mathrm{Mpa}(65 \mathrm{ksi})$
- $\mu=0.5$
clampDesign.m FoS $=3$

$$
\begin{aligned}
& \text { Tension }=3066.9 \mathrm{~N} \\
& \mathrm{~T}=11.8757 \mathrm{Nm}(105.11 \mathrm{in}-\mathrm{lb})
\end{aligned}
$$

## CLAMP DESIGN: Material \& Preload



| Condition | Ultimate Tensile Strength (PSI) | $\underset{\substack{\text { (PSI) }}}{0.2 \% \text { Yield Strength }}$ | $\underset{\text { 2in.) }}{\substack{\text { Elongation (\% } \\ \text { In }}}$ | Reduction <br> Of Area <br> (\%) | Hardness Brinell | Hardness Rockwell |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| H900 | 190,000 | 170,000 | 10 | 40 | 388 | C40 |
| H1025 | 155,000 | 145,000 | 12 | 45 | 331 | C35 |
| H1075 | 145,000 | 125,000 | 13 | 45 | 311 | C32 |
| H1150 | 135,000 | 105,000 | 16 | 50 | 277 | C28 |
| H1150-M | 115,000 | 75,000 | 18 | 55 | 255 | C24 |
| H1150-D | 125,000 | 105,000 | 16 | 50 | $\begin{gathered} 255 \min _{\max }-311 \\ \hline \end{gathered}$ | C24-33 |

- http://www.deltafastener.com/17-4\ ph.html
- http://www.mcmaster.com/\#=z2x9xd

$$
\begin{aligned}
& \cdot \sigma_{\mathrm{Y}}=861.84 \mathrm{Mpa}(125 \mathrm{ksi}) \\
& \cdot \mu=0.5 \text { (conservative) } \\
& \text { clampDesign.m FoS }=3 \\
& \text { - Tension }=5897.8 \mathrm{~N} \\
& \text { - } \mathrm{T}=22.8376 \mathrm{Nm}(202.13 \mathrm{in}-\mathrm{lb})
\end{aligned}
$$

Note: No Socket Head Screw for this material neither Silver Plated option (McMaster \& HoloKrome)

Note II: Torque similar to the recommended for Carbon Steel 1960 Series in T1100066 (200 in-lb)

## CLAMP DESIGN: Number of Screws

- Preliminary calculation using ClampDesign.m

$\mathrm{Pb} \square d_{p}=d-\frac{1.299038}{n}$
$d_{n}=d-\frac{0.649519}{n}$
$A_{s}=\frac{\pi}{4}\left(\frac{d_{m}+d_{p}}{2}\right)^{2}$

$$
P_{B_{M A X}}=\frac{\sigma_{Y I E L D}}{F o S} A_{s}
$$

$$
\text { - } P_{B}=2460.5 \mathrm{~N}
$$

- 316 SSTL + HELICOILS
- Bolts needed: min 2 screw per row for FoS $>3$
- $\mathrm{FoS}=5.5859$ (with 2 screw per row)
- 18-8 SSTL + HELICOILS
- Bolts needed: min 1 screw per row for FoS > 3
- $\mathrm{FoS}=3.7393$ (with 1 screw per row)


## - 17-4 PH SSTL + HELICOILS

- Bolts needed: min 1 screw per row for FoS > 3
- FoS $=7.1909$ (with 1 screw per row)

Note: The final number of screws per row and the material election will be driven by the contact calculation between the blade and the platform

## Balancing Mass

- Preliminary Mass Balancing
- First Bench Design ( $\approx 17 \mathrm{~kg}$ )

```
Center of mass: ( millimeters)
    X=0.0
    Y=0.0
    Z=6.0
```

Note: The remaining 1 kg has been reserved for the Bench Optimization and final CoG adjustment

Note II: This model is the one that has been used for the injection bench optimization and will be tunned according to the results of this optimization.

## Injection Bench Assembly FEA

- Injection Bench Assembly for FEA with Masses

- Global Contact: Bonded
- Mesh: Compatible

Mass $=35301.6$ grams
Volume $=675.6$ cubic inches
Surface area $=1493992.4$ square millimeters
Center of mass: (millimeters)
$\mathrm{X}=0.3$
$\mathrm{Y}=1.3$
$Z=6.5$

| OPTICS | 9.933 | kg | 9.94927 |
| :--- | ---: | :--- | ---: |
| SUSPENSION | 17.131 | $\mathbf{k g}$ | 17.1857 |
| MASS | 8.131 | $\mathbf{k g}$ | 8.167 |
|  |  |  |  |
| TOTAL WEIGHT | 35.195 | $\mathbf{k g}$ | 35.301 |
| Mass to 36 kg | 0.80 | kg | 0.70 |

## Injection Bench Assembly FEA

- Injection Bench Assembly for FEA

- Bench Mass $=17.06 \mathrm{~kg}$

NOTE: The remaining 940 g will be used to position the CoG

## Injection Bench Assembly FEA

- Injection Bench Assembly for FEA (ANSYS)

- Equivalent Strain


NOTE: The areas with high Strain should be reinforced using the 940 g

## Injection Bench Assembly FEA

## - Injection Bench Assembly for FEA (no MASSES)

Density $=0.044$ kilograms per cubic inch

## Mass $=17.060$ kilograms

Volume $=385.579$ cubic inches
Surface area $=781537.629$ square millimeters
Center of mass: ( millimeters)
$X=21.518$
$Y=-13.145$
$Z=-20.716$

NOTE: The remaining 940 g will be used to position the CoG and to improve the high strain areas


## Center of Mass

## - Optics O3

## Mass $=9932.652$ grams

Volume $=0.003$ cubic meters
Surface area $=1575061.645$ square millimeters
(Ceñéer ồ màss: (mililimeters)
$\begin{array}{ll}\mathrm{X}=35.471 \\ 1 \\ \mathrm{Y}=-16.171 \\ \mathrm{Z} & =37.586\end{array}$

Note: The Center of Mass of the optics is already close to the desired CoM (the barycenter of the clamps attachment points)

Desired Center of Mass


## Balancing Mass

## - Preliminary Mass Balancing

- First Bench Design ( $\approx 17 \mathrm{~kg}$ )
- Lateral Masses

Mass $=32637.107$ grams

Volume $=0.010$ cubic meters
Surface area $=2530447.011$ square millimeters
(Center of mass: ( millimeters) )
$X=19.147$
Y $\quad$ = 15.506
$Z=5.293$

| OPTICS |  | 9.933 |
| :--- | ---: | ---: |
| kg |  |  |
| SUSPENSION |  | 17.131 |
| Mg |  |  |
| MASS |  | 5.500 |

## TOTAL WEIGHT

32.564 kg

Mass to $\mathbf{3 6} \mathbf{~ k g}$
3.44 kg

Note: The remaining 3.44 kg ( 940 g in the injection bench and the rest as balancing masses on the bench) should be placed near the dashed area in order to balance the assembly


## Balancing Mass



Note: The balancing masses are more distributed in the left part of the assembly. That could be solved by "naturally" placing the CoG of the Injection Bench opposed to the Optics one.

## Center of Mass

- FEA

Mass $=33909.985$ grams
Volume $=638.177$ cubic inches
Surface area $=2301.473$ square inches
Center of mass: (inches)

$$
\begin{aligned}
& X=0.346 \\
& Y=0.405 \\
& Z=-0.079
\end{aligned}
$$



## Center of Mass

- With balancing masses

Mass properties of D1500292 aLIGO VOPO Injection Bench Assembly TO FEA Configuration: Default
Coordinate system: Coordinate System1
Mass $=37.37$ kilograms
Volume $=727.18$ cubic inches
Surface area $=2533.66$ square inches
Center of mass: (inches)
$X=0.08$
$y=1.21$ | $\quad \mathrm{Y}=1.21$
Z $=0.09$
Principal axes of inertia and principal moments of inertia: (kilograms * square inches) Taken at the center of mass.

| $\mathrm{Ix}=(-0.99,-0.01,0.17)$ | $\mathrm{Px}=1150.51$ |
| :--- | :--- |
| $\mathrm{Iy}=(0.17,-0.03,0.99)$ | $\mathrm{Py}=1818.71$ |

$\mathrm{y}=(0.17,-0.03,0.99) \quad \mathrm{Py}=1818.71$
$\mathrm{Iz}=(-0.01,1.00,0.04) \quad \mathrm{Pz}=2833.14$
Moments of inertia: (kilograms * square inches)
Taken at the center of mass and aligned with the output coordinate system.
$L x x=1169.36$
$L y x=18.00$
$L y y=18.00$
$L y=2831.62$ $L x z=-109.72$
$\operatorname{lyx}=18.00$
$L y y=2831.62$
$L z y=-38.10$ $L z z=1801.38$

Moments of inertia: (kilograms * square inches ) Taken at the output coordinate system.

| Ixx $=1224.02$ | Ixy $=21.63$ | Ixz $=-109.46$ |
| :--- | :--- | :--- |
| Iyx $=21.63$ | Iyy $=2832.14$ | Iyz $=-34.21$ |
| Izx $=-109.46$ | Izy $=-34.21$ | Izz $=1856.00$ |

$y y=2832.14$
Izy $=-34.21$

Geometric Center

Center of
Mass

## Center of Mass

- Optics

Mạss $=9893.8$ grams
Volume $=0.0$ cubic meters
Surface area $=1564570.5$ square millimeters
(Center of mass: ( millimeters)

| $\mathrm{X}=35.2$ | I |
| :--- | :--- |
| $\mathrm{Y}=-12.3$ | I |
| $\mathrm{Z}=37.5$ |  |

Note: The Center of Mass of the optics is already close to the desired CoM (the barycenter of the clamps attachment points)

Desired Center of Mass

## Balancing Mass

- Preliminary Mass Balancing
- First Bench Design ( $\approx 16 \mathrm{~kg}$ )



## Balancing Mass

## - Optics + Balancing Mass

## Mass $=36054.2$ grams

Volume $=0.0$ cubic meters
Surface area $=2528676.9$ square millimeters

| $\begin{aligned} & \text { (Center of mass: ( } \text { ( } \text { illimeters) } \\ & \begin{array}{l} \quad X=0.0 \\ \quad \mathrm{Y}=0.0 \end{array} \end{aligned}$ |
| :---: |
|  |  |
|  |  |
|  |  |


| ELEMENT | Name | Description | Unit Weight |
| :--- | :--- | :--- | :--- |
| TYPE |  | (g) |  |

Note: The balancing masses are basically placed in the lateral of the Injection Bench ( 5.4 kg ). However, 2.6 kg are placed on the Bench as contingency for the optics weight

Desired Center of Mass


## Center of Mass

## - FEA

Mass $=35805.3$ grams
Volume $=689.3$ cubic inches
Surface area $=1496522.6$ square millimeters
Center of mass: ( millimeters)

$$
\begin{aligned}
& X=-0.4 \\
& Y=-1.5 \\
& z=6.3
\end{aligned}
$$

## Center of Mass

- FEA
- First Frequency Study
- $f_{\text {Compatible }}=205.19 \mathrm{~Hz}$



## Center of Mass

- Optics

Mass $=6773.1$ grams
Volume $=0.0$ cubic meters
Surface area $=929838.5$ square millimeters


Note: The Center of Mass of the optics far from the desired but there is a lot of available space to locate the balancing masses


## Balancing Mass

- Preliminary Mass Balancing
- First Bench Design ( $\approx 18 \mathrm{~kg}$ )

| OPTICS | 6.750 | kg |
| :--- | ---: | :--- |
| SUSPENSION | 18.134 | kg |
| MASS | 11.068 | kg |
|  |  |  |
| TOTAL WEIGHT | 35.952 | kg |
| Mass to 36 kg | 0.05 | kg |

Note: The weight of the suspension (basically the Injection Bench) is approximate and will be defined after the optimization to increase the frequency


## Balancing Mass

## - Optics + Balancing Mass



Note: In this case there is more balancing mass placed on the bench than in the laterals as it has to compensate the lack of optics


## Center of Mass

- FEA

Mass $=35532.6$ grams
Volume $=638.1$ cubic inches
Surface area $=1370261.1$ square millimeters
Center of mass: (millimeters)

$$
\begin{aligned}
& X=2.7 \\
& Y=-0.8 \\
& Z=4.8
\end{aligned}
$$



## Center of Mass

- FEA
- First Frequency Study
- $f_{\text {Compatible }}=211.84 \mathrm{~Hz}$



## Optimization

- FEA
- Bench Optimization for O3
- $f_{\text {Compatible }}=209.28 \mathrm{~Hz}$
- Bench Mass $=17.296 \mathrm{~kg}$



## Fixer

- Screw \#10-24 Static Calculation
- Load $=\mathrm{P}=117.72 \mathrm{~N}$
- $\mathrm{FoS}=32.1622$
- Maximum Play
- Vertical: $0.5-2 \frac{0.19}{2}=0.31 \mathrm{in}(7.874 \mathrm{~mm})$
- Horizontal: $0.5-2 \frac{0.19}{2}=0.31 \mathrm{in}(7.874 \mathrm{~mm})$
- Lateral: $0.32-2 \frac{0.19}{2}=0.13 \mathrm{in}(3.302 \mathrm{~mm})$


