

Modal Damping of a Quad Pendulum for Advanced Gravitational Wave Detectors

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Abstract

Motivation: Observe gravitational waves from astrophysical sources (supernovae, pulsars, black hole mergers, etc) using the LIGO observatories.

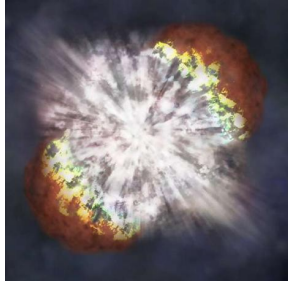
Problem: Multi-DOF isolation systems enhance ground motion at high Q resonances. Damped using active feedback. This control introduces additional noise. Optimal control required achieve adequate trade-off.

Solution: Modal damping to simplify and decouple optimization of each mode's damping. Also permits real-time tuning.

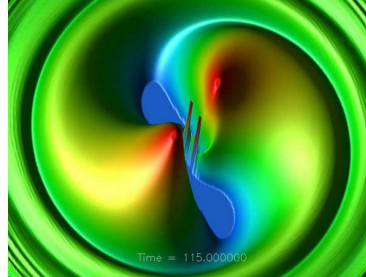
Outline

1. LIGO and gravitational waves
2. Seismic (vibration) isolation
3. Competing damping control goals
4. Method of modal damping
5. Optimization of modal damping
6. Results

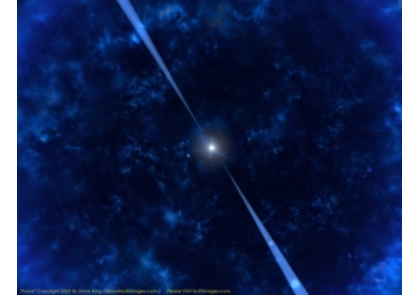
Gravitational Waves



Supernova

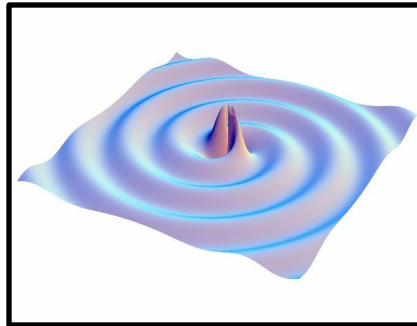


Merging Black Holes

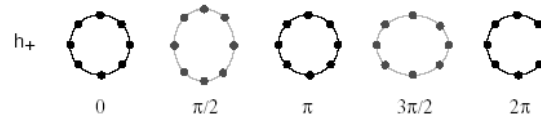
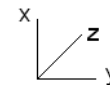


Pulsar

Wave of strain amplitude h



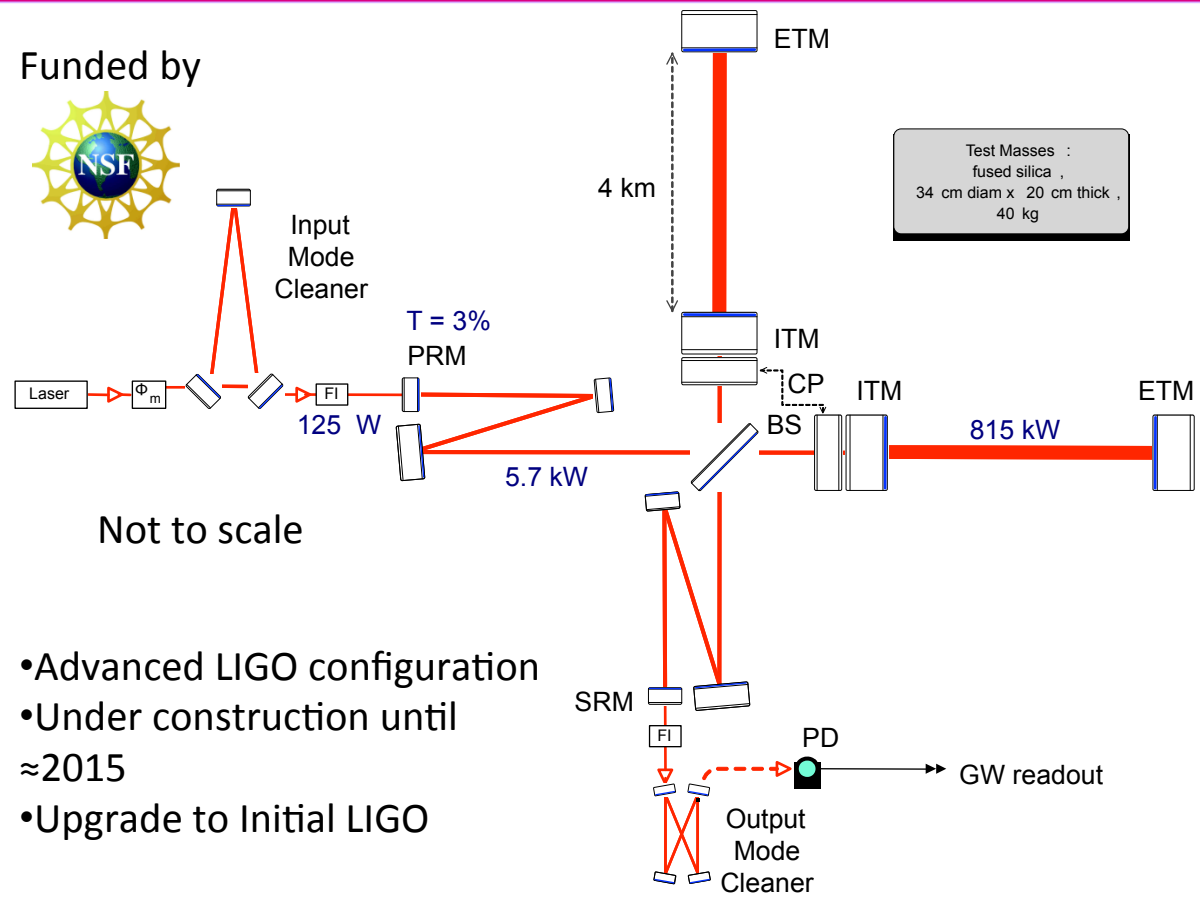
Wave traveling into screen along z axis



Time →

- **Supernovae**
 - Asymmetry required
- **Coalescing Binaries**
 - Black Holes or Neutron Stars Mergers
- **Pulsars**
 - Asymmetry required
- **Stochastic Background (Big bang, etc.)**

Gravitational-wave Observatory (LIGO)



Hanford, WA

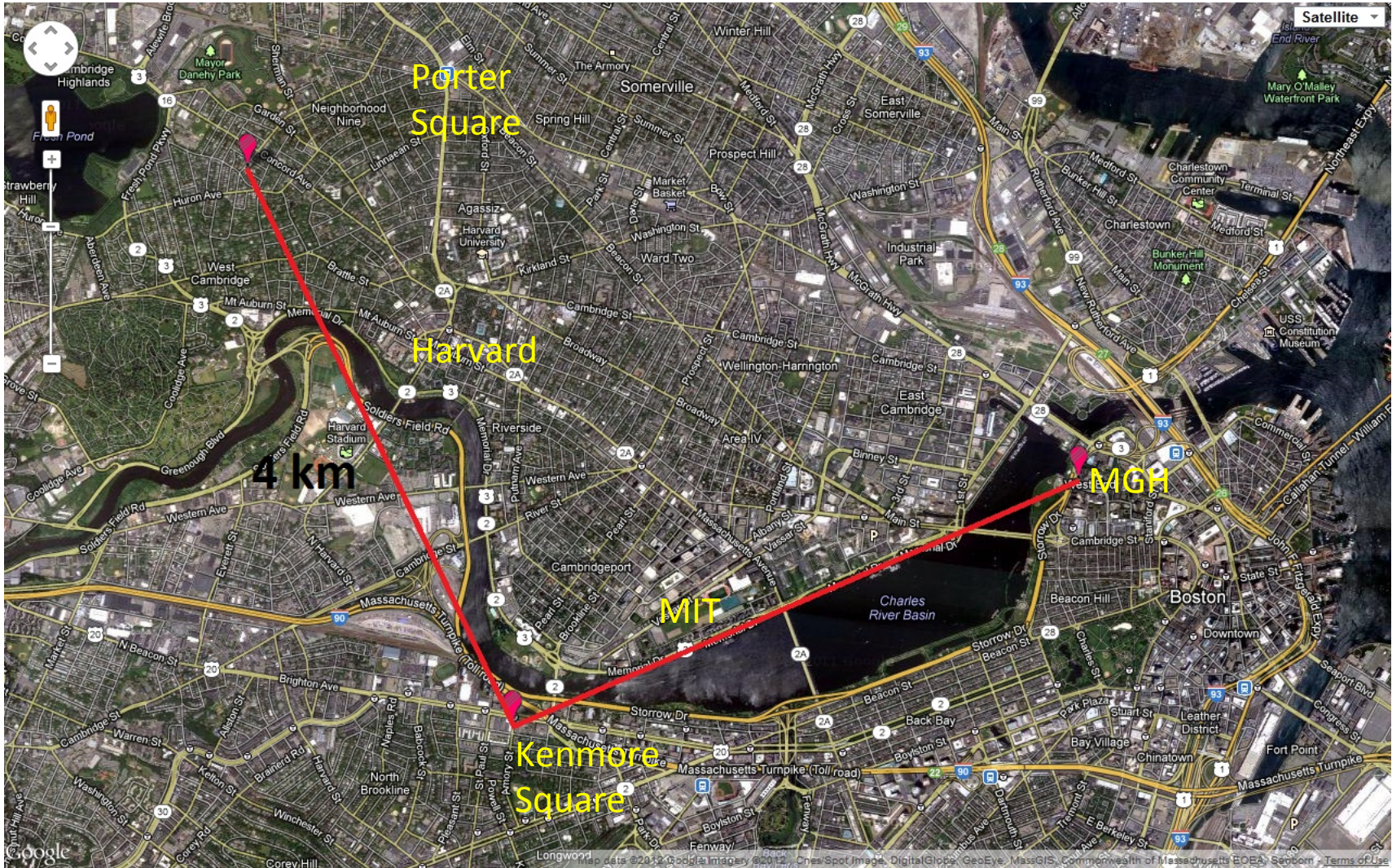


Livingston, LA

- Advanced LIGO configuration
- Under construction until ≈2015
- Upgrade to Initial LIGO

- **3, 4 km interferometers at 2 sites in the US**
- **Michelson interferometers with Fabry-Pérot arms**
- **Optical path enclosed in vacuum**
- **Sensitive to strains around 10^{-22} -> $10^{-19} m_{rms}$**
- **LIGO Budget ≈ \$60 Million per year from NSF.**
- **Operated by MIT and Caltech.**

If we put LIGO in Cambridge, MA



LIGO spans 16 km². Cambridge, MA covers 16.65 km² (wikipedia http://en.wikipedia.org/wiki/Cambridge,_Massachusetts).

LIGO

LIGO Scientific Collaboration



● Australian Consortium for Interferometric Gravitational Astronomy

● The Univ. of Adelaide

● Andrews University

● The Australian National Univ.

● The University of Birmingham

● California Inst. of Technology

● Cardiff University

● Carleton College

● Charles Sturt Univ.

● Columbia University

● CSU Fullerton

● Embry Riddle Aeronautical Univ.

● Eötvös Loránd University

● University of Florida

● German/British Collaboration for the Detection of Gravitational Waves

● University of Glasgow

● Goddard Space Flight Center

● Leibniz Universität Hannover

● Hobart & William Smith Colleges

● Inst. of Applied Physics of the Russian Academy of Sciences

● Polish Academy of Sciences

● India Inter-University Centre for Astronomy and Astrophysics

● Louisiana State University

● Louisiana Tech University

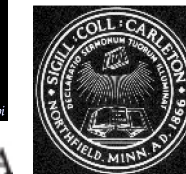
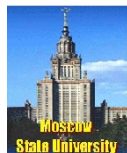
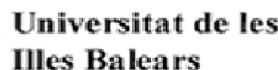
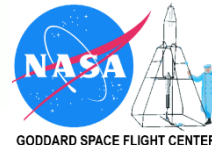
● Loyola University New Orleans

● University of Maryland

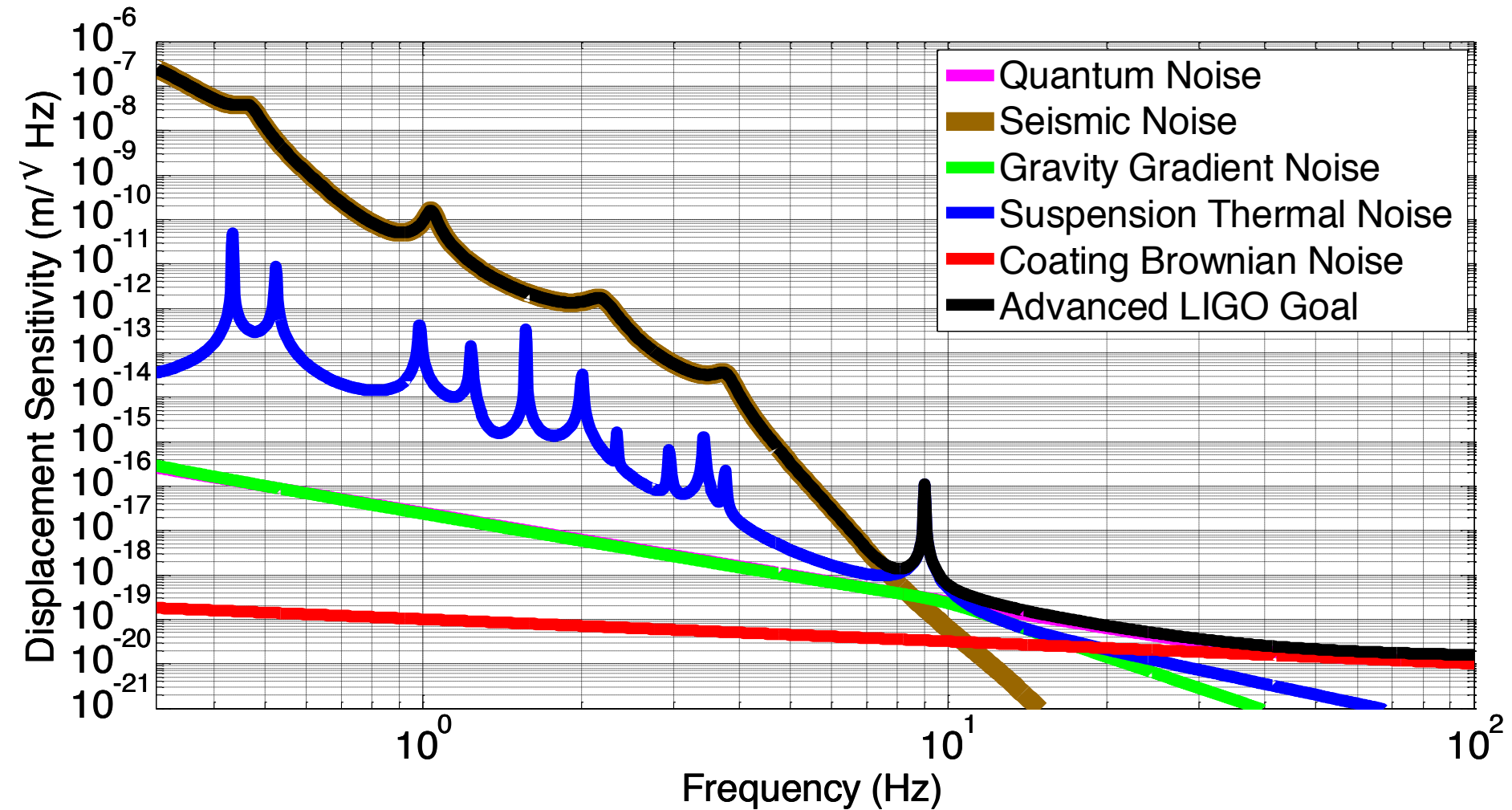
● Max Planck Institute for Gravitational Physics



- University of Michigan
- University of Minnesota
- The University of Mississippi
- Massachusetts Inst. of Technology
- Monash University
- Montana State University
- Moscow State University
- National Astronomical Observatory of Japan
- Northwestern University
- University of Oregon
- Pennsylvania State University
- Rochester Inst. of Technology
- Rutherford Appleton Lab
- University of Rochester
- San Jose State University
- Univ. of Sannio at Benevento, and Univ. of Salerno
- University of Sheffield
- University of Southampton
- Southeastern Louisiana Univ.
- Southern Univ. and A&M College
- Stanford University
- University of Strathclyde
- Syracuse University
- Univ. of Texas at Austin
- Univ. of Texas at Brownsville
- Trinity University
- Tsinghua University
- Universitat de les Illes Balears
- Univ. of Massachusetts Amherst
- University of Western Australia
- Univ. of Wisconsin-Milwaukee
- Washington State University
- University of Washington

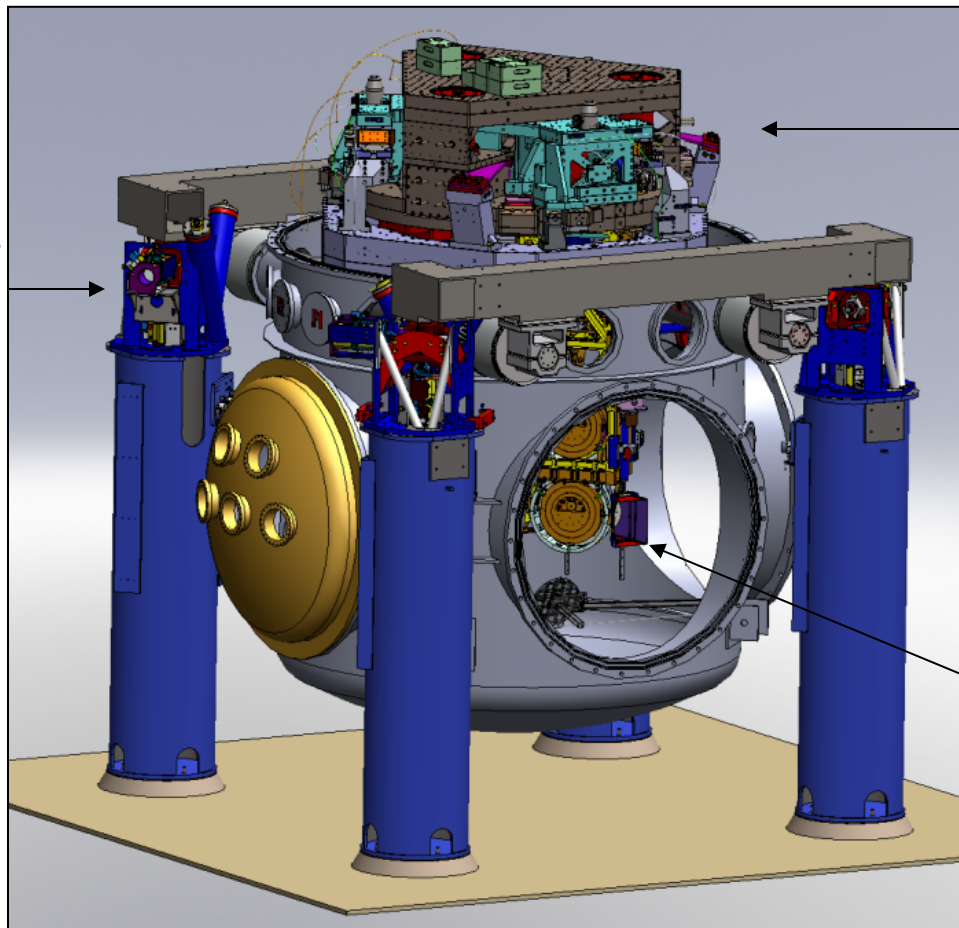


Projected Sensitivity for Advanced LIGO



Suspensions and Seismic Isolation

Advanced LIGO test mass isolation



active isolation platform (2 stages of isolation)

hydraulic external pre-isolator (HEPI) (one stage of isolation)

quadruple pendulum (four stages of isolation) with monolithic silica final stage

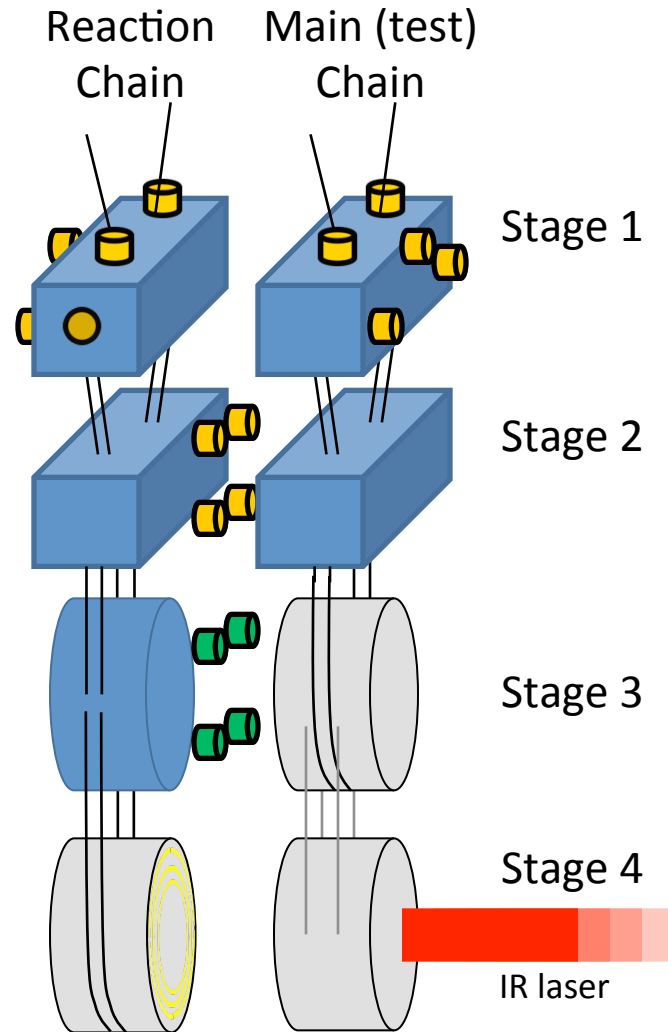
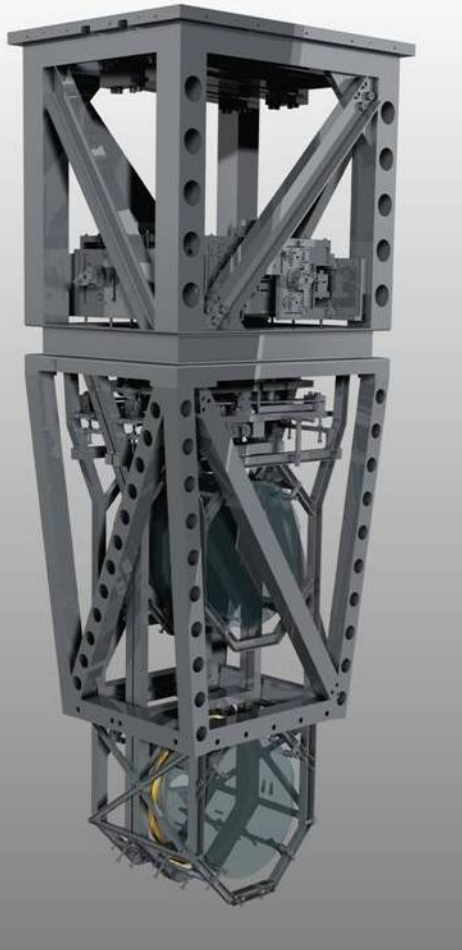


active isolation
platform (2 stages
of isolation)

quadruple pendulum (four
stages of isolation)

Installing prototype quad
pendulum with glass optic on
metal wires, Jan 2009 at MIT.

Quadruple Pendulum





Purpose

- Test mass (stage 4) isolation.
the test mass consists of a 40 kg high reflective mirror

Control

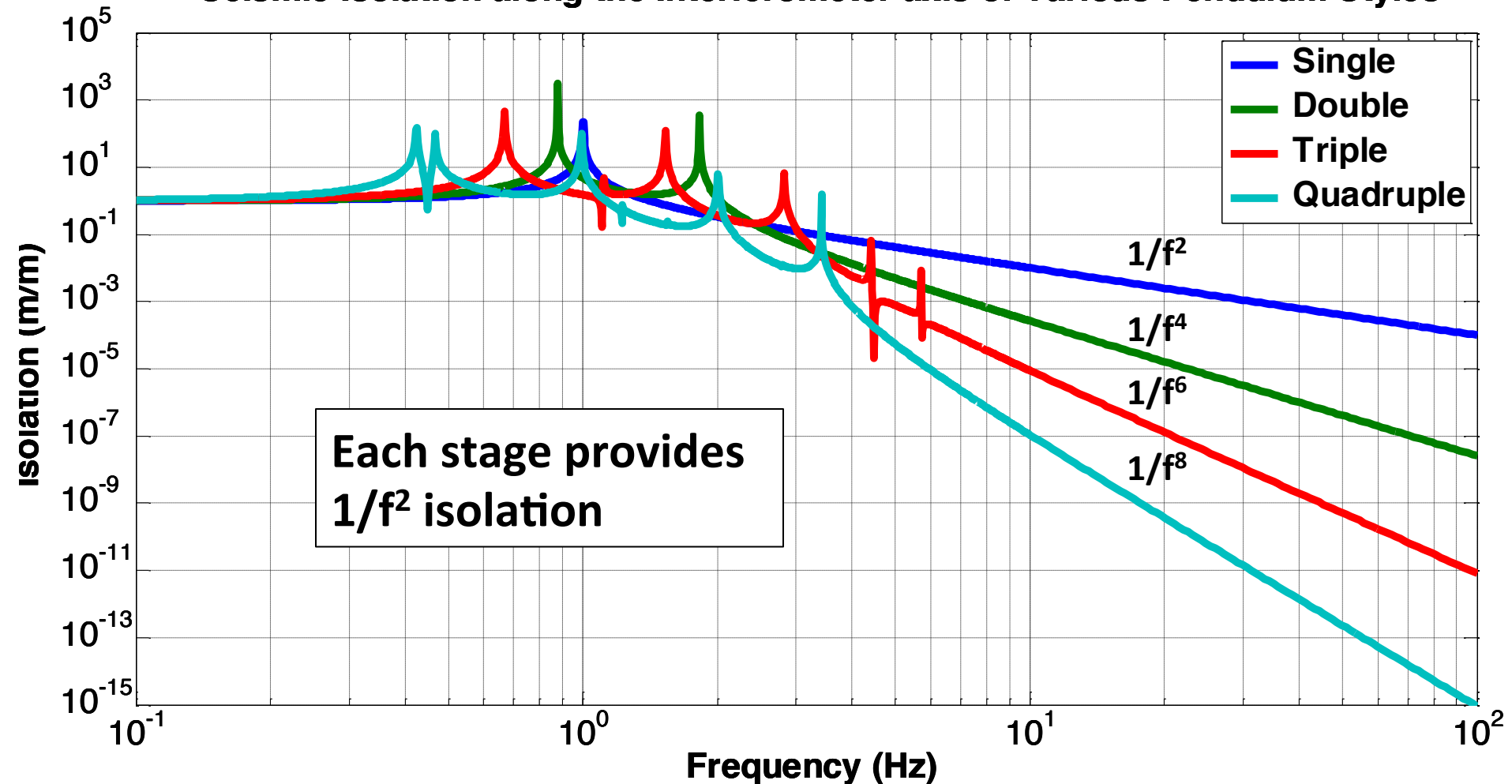
- Damping –stage 1
- Cavity length - all stages

Sensors/Actuators

-  BOSEMs at stage 1 & 2
-  AOSEMs at stage 3
- Opt. lev. and interf. sigs. at stage 2
- Electrostatic drive (ESD) at stage 4

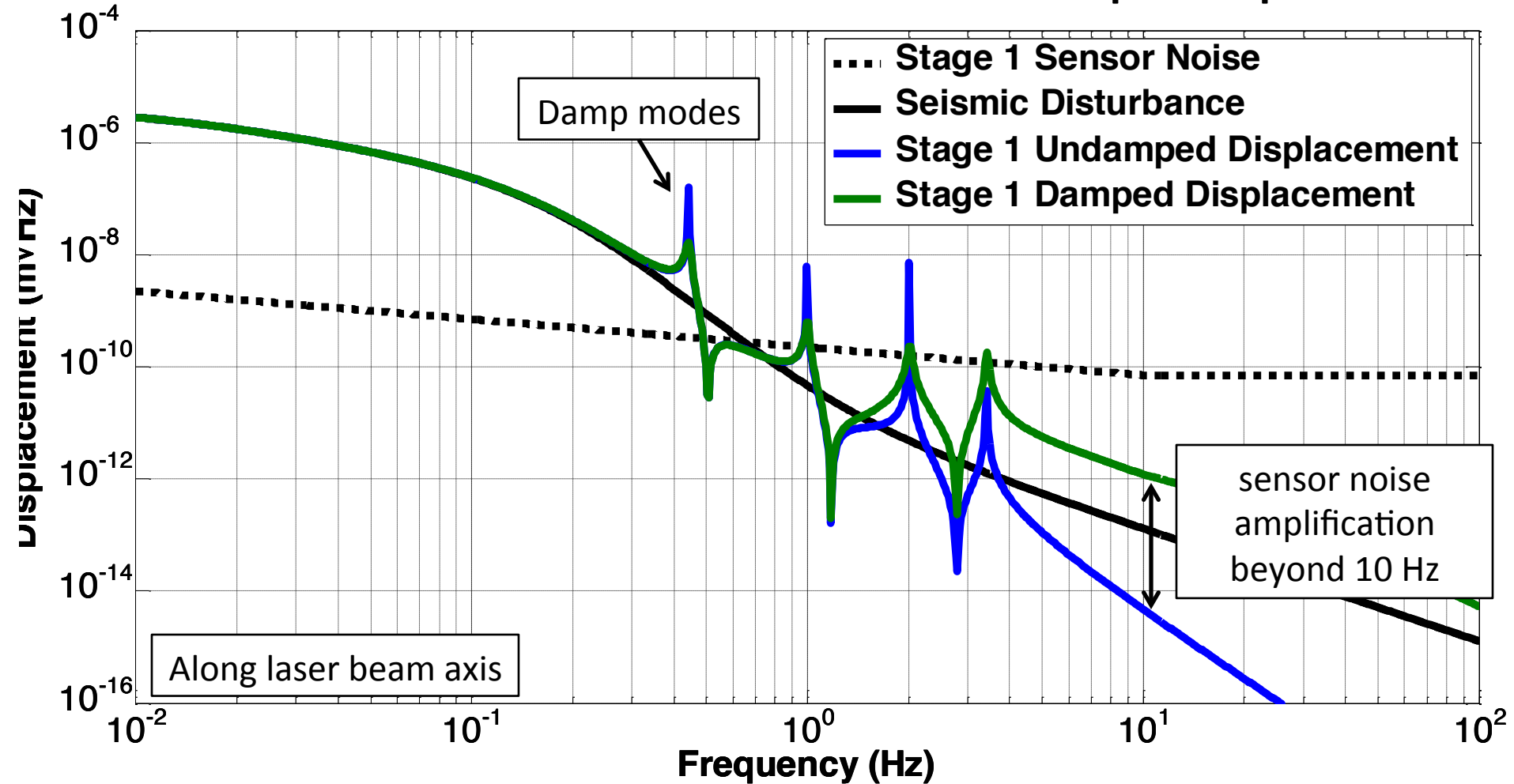
Multi-stage Isolation Performance

Seismic Isolation along the Interferometer axis of Various Pendulum Styles

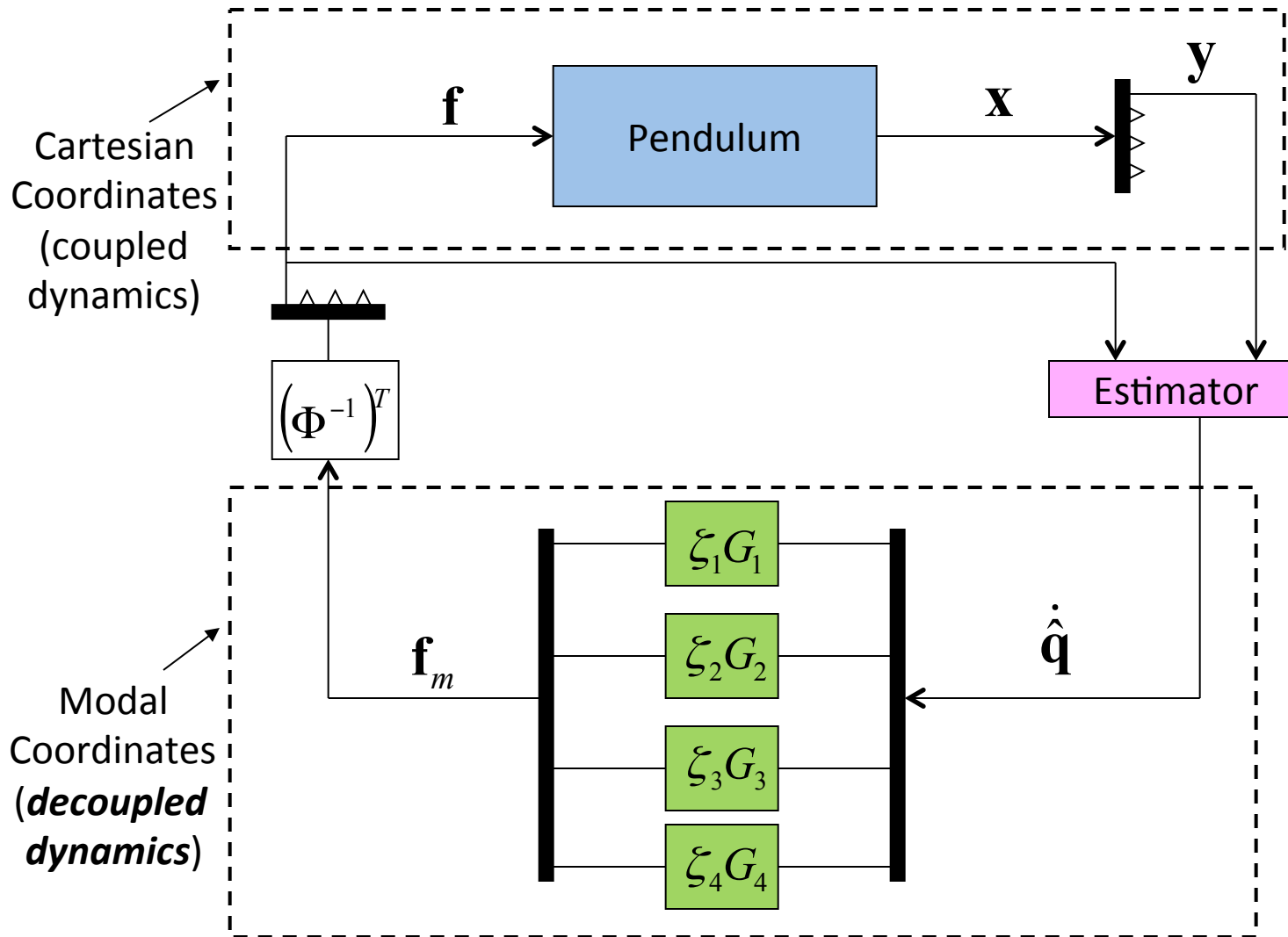


Two Competing Goals

Simulated Seismic Disturbance and Sensor Noise Amplitude Spectrum



Modal Damping with State Estimation

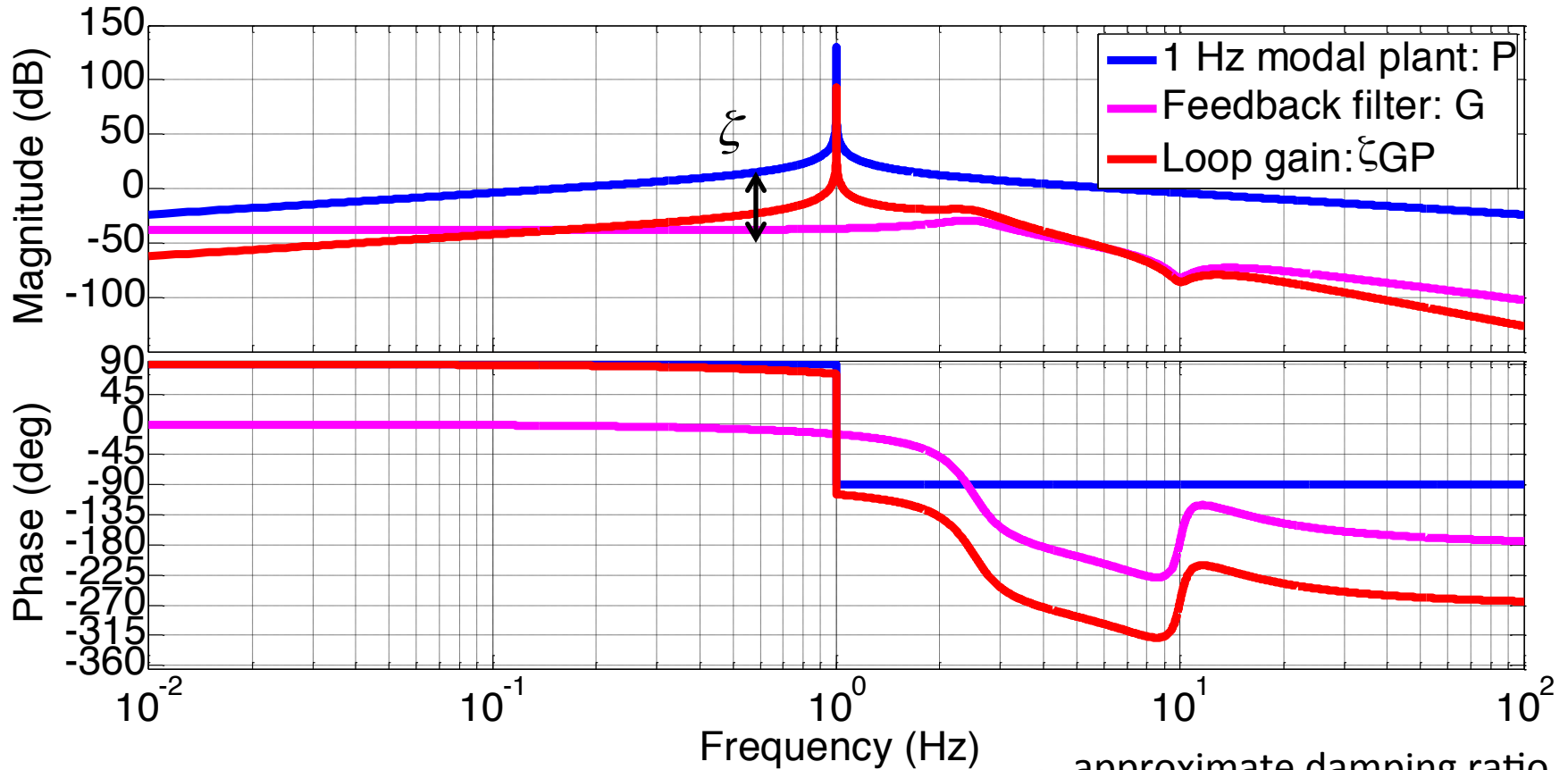


- $\mathbf{x} = \Phi \mathbf{q}$
- $\Phi =$ pendulum eigenvector matrix
- $\mathbf{x} =$ Cartesian coordinates
- $\mathbf{y} =$ sensor sig.
- $\hat{\mathbf{q}} =$ estimated modal coord.
- $\mathbf{f} =$ Cartesian damping forces
- $\mathbf{f}_m =$ modal damping forces
- $G_i = i_{th}$ mode damping filter
- $\xi_i = i_{th}$ mode damping gain

Modal Feedback Design

Bode Diagram

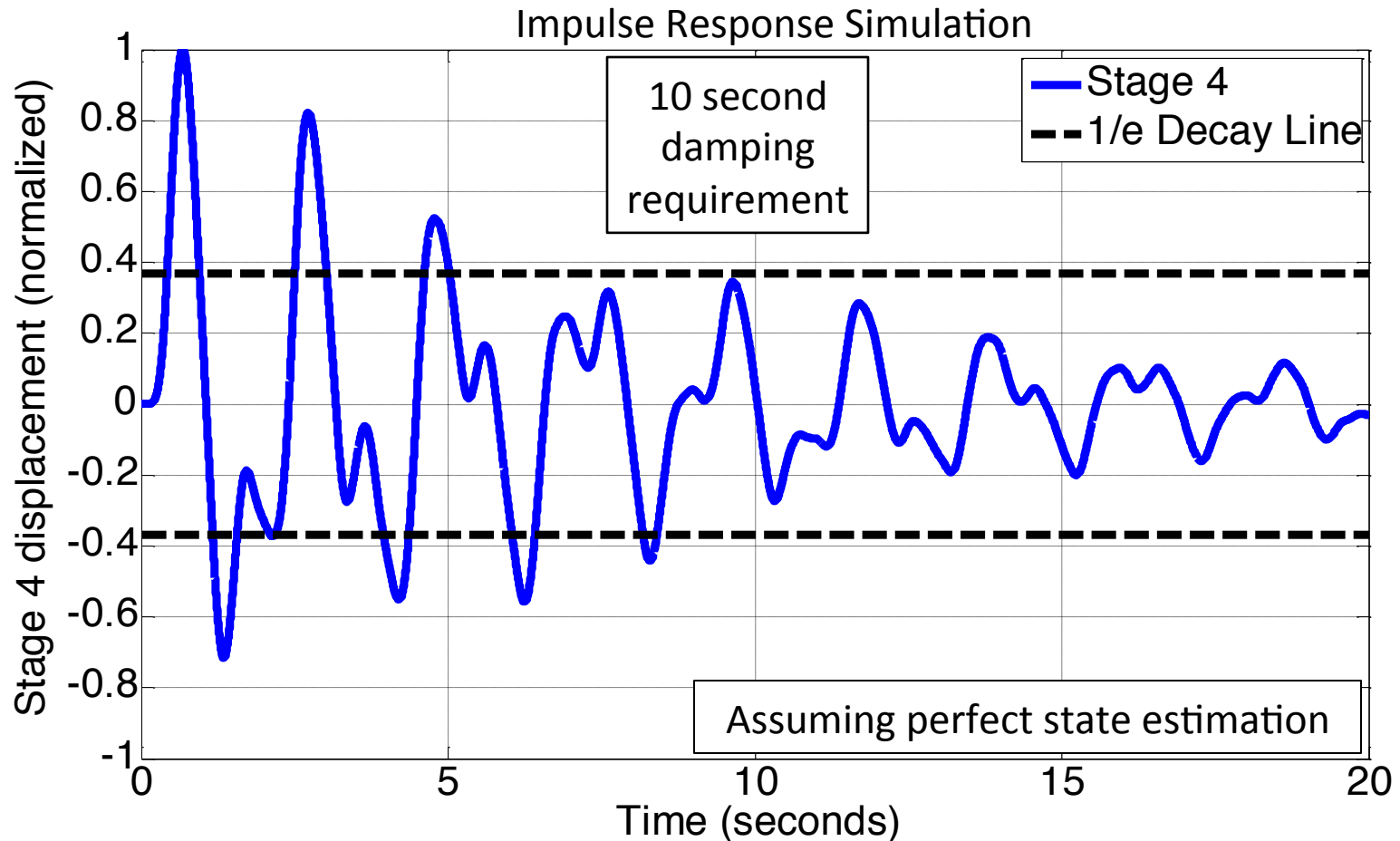
GM = 9.1865 (2.403 Hz), PM = 74.5731 (1.0464 Hz)



approximate damping ratio

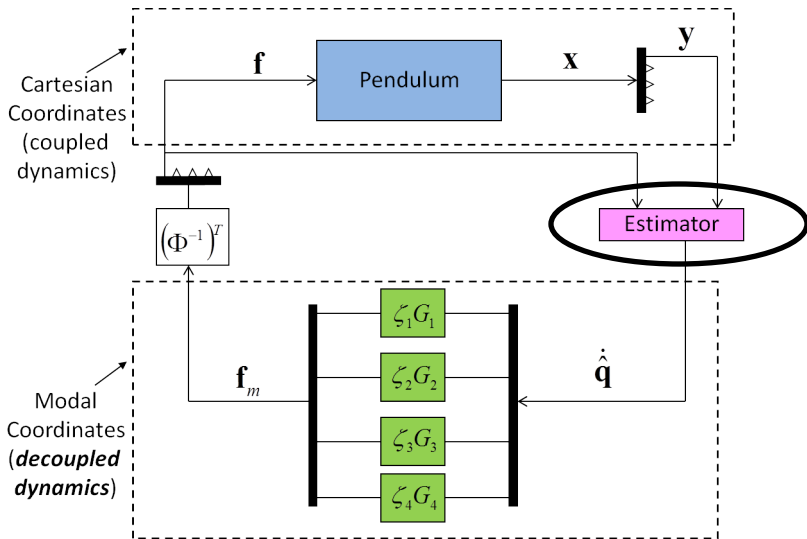
$$G = \frac{s^2 + 2\pi s + (20\pi)^2}{\left[s^2 + 5\sin(10^\circ)\omega_n s + (2.5\omega_n)^2 \right] \left[s^2 + 20\pi s + (20\pi)^2 \right]}, \quad 0 \leq \xi < 1, \quad P = \frac{s}{s^2 + \omega_n^2}, \quad \omega_n = 2\pi$$

Damped Response to Impulse from Gnd.



Modes	1	2	3	4
Freq (Hz)	0.443	0.996	2.001	3.416
ξ	0.040	0.018	0.009	0.001

Estimator Design



$$\begin{bmatrix} \dot{\hat{\mathbf{q}}} \\ \hat{\mathbf{q}} \end{bmatrix} = \mathbf{A}_m \begin{bmatrix} \hat{\mathbf{q}} \\ \dot{\hat{\mathbf{q}}} \end{bmatrix} + \mathbf{B}_m \begin{bmatrix} \mathbf{f} \\ 0 \end{bmatrix} - \mathbf{L}_m \left(\mathbf{C}_m \begin{bmatrix} \hat{\mathbf{q}} \\ \dot{\hat{\mathbf{q}}} \end{bmatrix} - \mathbf{y} \right)$$

$\mathbf{A}_m, \mathbf{B}_m \rightarrow$ Pendulum model

$\mathbf{C}_m \rightarrow$ Sensor matrix

$\mathbf{L}_m \rightarrow$ Estimator feedback matrix

Linear Quadratic Regulator (LQR) design

$$J = \int_0^{\infty} \left(\begin{bmatrix} \tilde{\mathbf{q}}^T & \dot{\tilde{\mathbf{q}}}^T \end{bmatrix} \mathbf{Q} \begin{bmatrix} \tilde{\mathbf{q}} \\ \dot{\tilde{\mathbf{q}}} \end{bmatrix} + \mathbf{z}_m^T \mathbf{R} \mathbf{z}_m \right) dt$$

$$\mathbf{L}_m = \arg \min_{\mathbf{L}_m} (J)$$



Q, R

Weight cost of using noisy sensor

Weight cost of estimation error

$\tilde{\mathbf{q}} = \hat{\mathbf{q}} - \mathbf{q} =$ estimation error

$\mathbf{z}_m = -\mathbf{L}_m^T \begin{bmatrix} \tilde{\mathbf{q}} \\ \dot{\tilde{\mathbf{q}}} \end{bmatrix} \rightarrow$ sensor noise amplification

Choosing Q and R: Not Unique

$$J = \int_0^\infty \left(\begin{bmatrix} \tilde{\mathbf{q}}^T & \dot{\tilde{\mathbf{q}}}^T \end{bmatrix} \mathbf{Q} \begin{bmatrix} \tilde{\mathbf{q}} \\ \dot{\tilde{\mathbf{q}}} \end{bmatrix} + \mathbf{z}_m^T \mathbf{R} \mathbf{z}_m \right) dt$$

$$\mathbf{L}_m = \underset{\mathbf{L}_m}{\operatorname{arg\,min}}(J)$$

Q

$$\begin{bmatrix} \tilde{q}_1^T & \dots & \dot{\tilde{q}}_{n-1}^T & \dot{\tilde{q}}_n^T \end{bmatrix} \begin{bmatrix} 0 & & & 0 \\ & \dots & & \\ & & m_{n-1}^{-2} & \\ 0 & & & m_n^{-2} \end{bmatrix} \begin{bmatrix} \tilde{q}_1 \\ \dots \\ \dot{\tilde{q}}_{n-1} \\ \dot{\tilde{q}}_n \end{bmatrix}$$

$$\mathbf{R} = \begin{bmatrix} R_1 & & & 0 \\ & R_2 & & \\ & & \dots & \\ 0 & & & R_m \end{bmatrix}$$

m_i = modal mass of mode i

m_i^{-1} = modal velocity impulse response amplitude

R is still to be determined

Solving the R matrix for MIMO Modal Damping

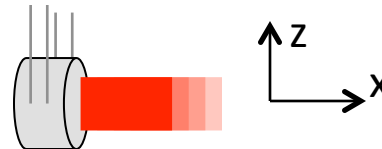
Try a bunch of R matrices and see what works best

$$J_R(R) = \max_i (T_{s,i}^2) + \max_i (N_i^2)$$
$$R = \arg \min(J_R)$$

Measure 'best' with an auxiliary cost function.

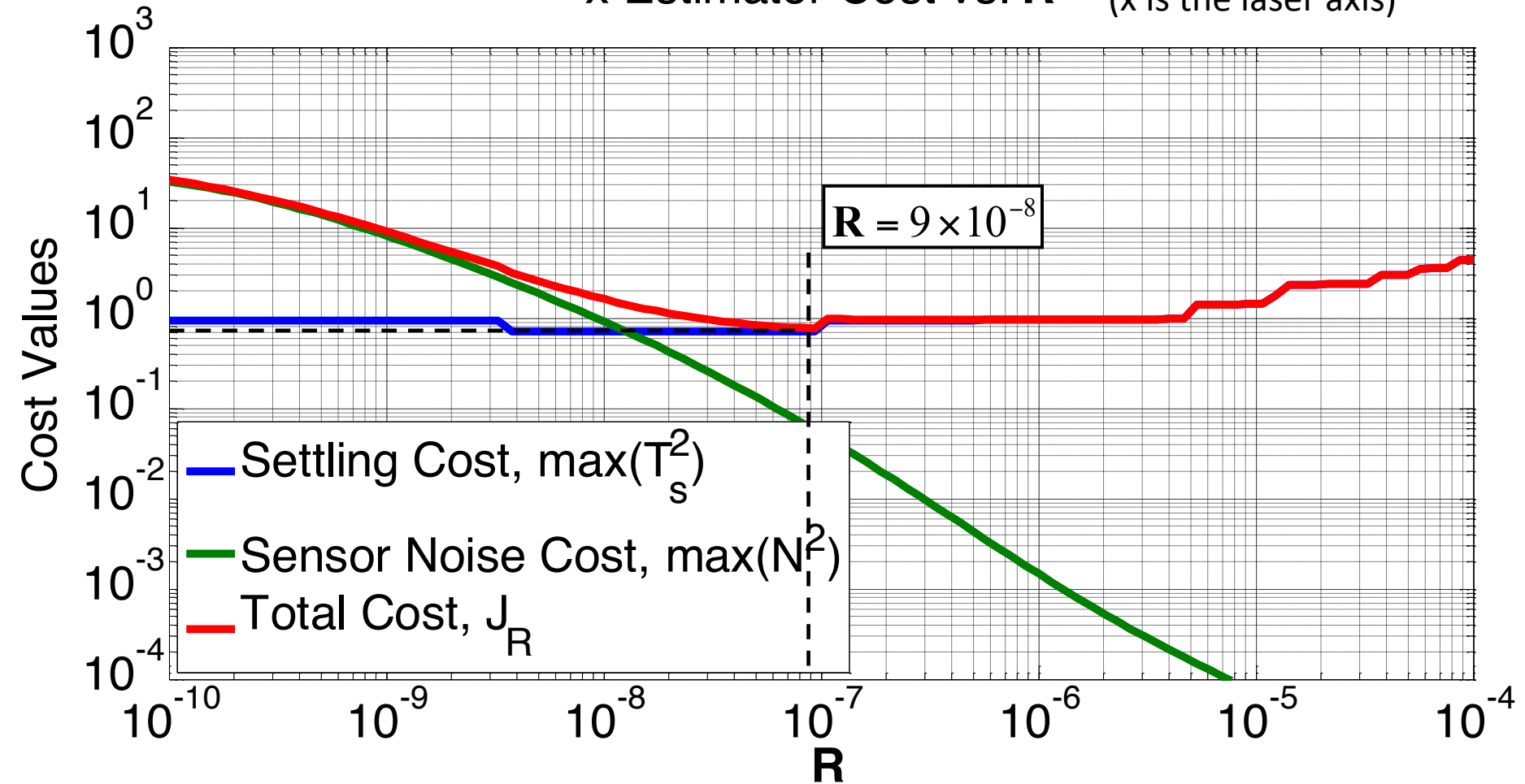
- $T_{s,i} = \frac{\text{Stage 4 settling time for DOF } i}{10 \text{ seconds}}$
- $N_i = \frac{\text{Stage 4 sensor noise for DOF } i \text{ at } 10 \text{ Hz}}{\text{Stage 4 noise requirement for DOF } i \text{ at } 10 \text{ Hz}}$

DOFs i are: x, y, z, yaw, pitch, roll

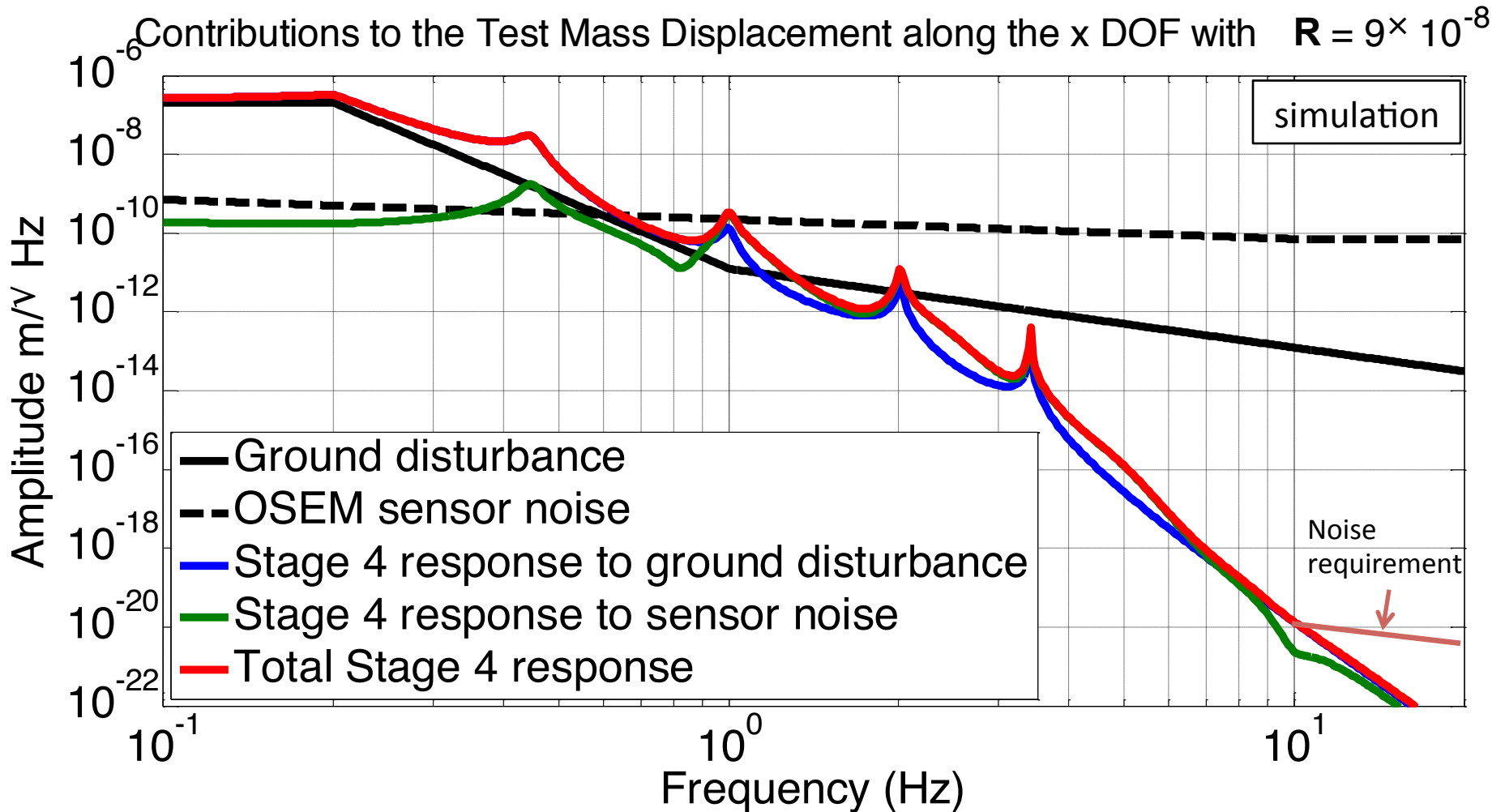


Modal Estimation Cost

x Estimator Cost vs. R (x is the laser axis)



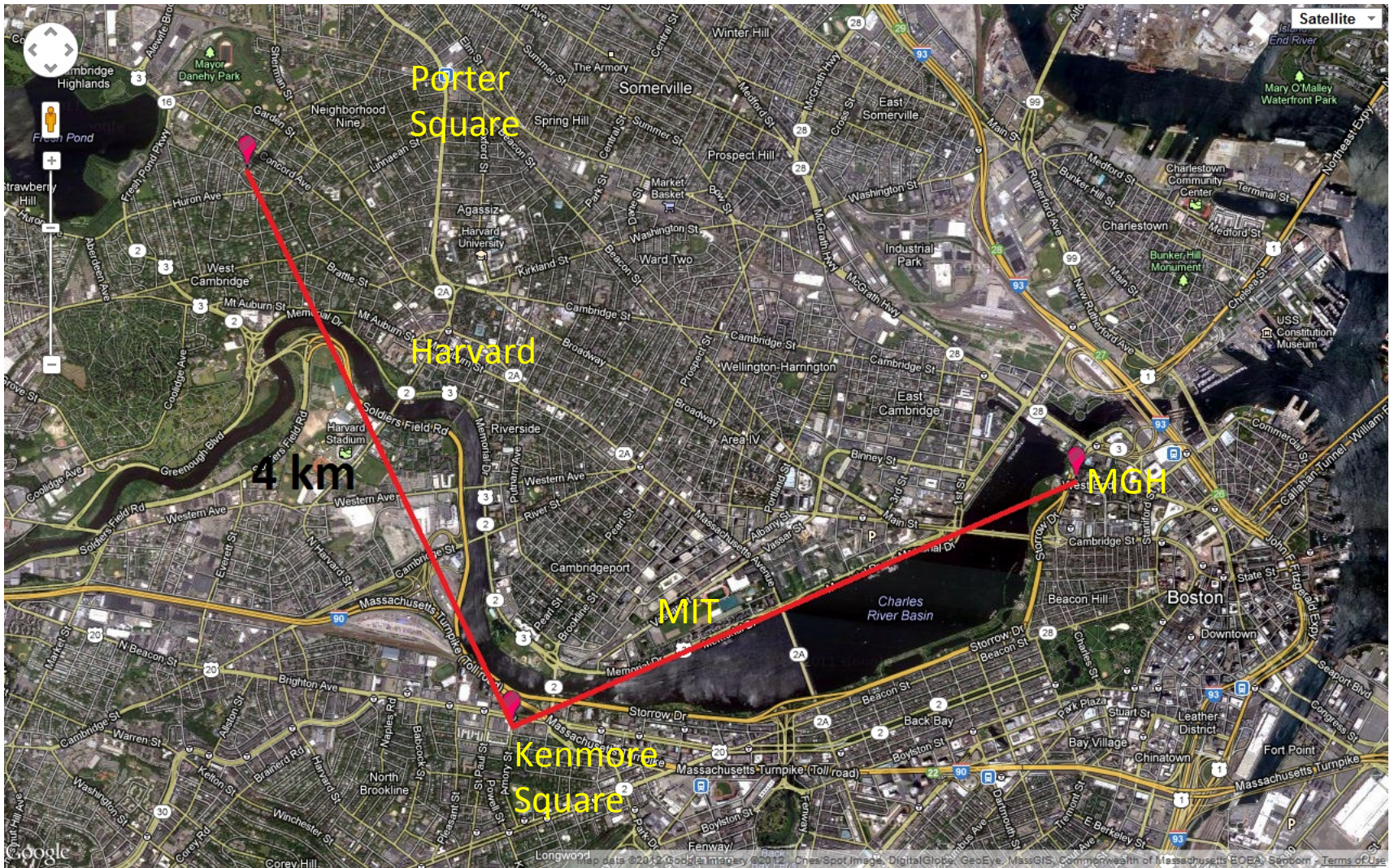
Optimal Noise Amplification



Conclusions

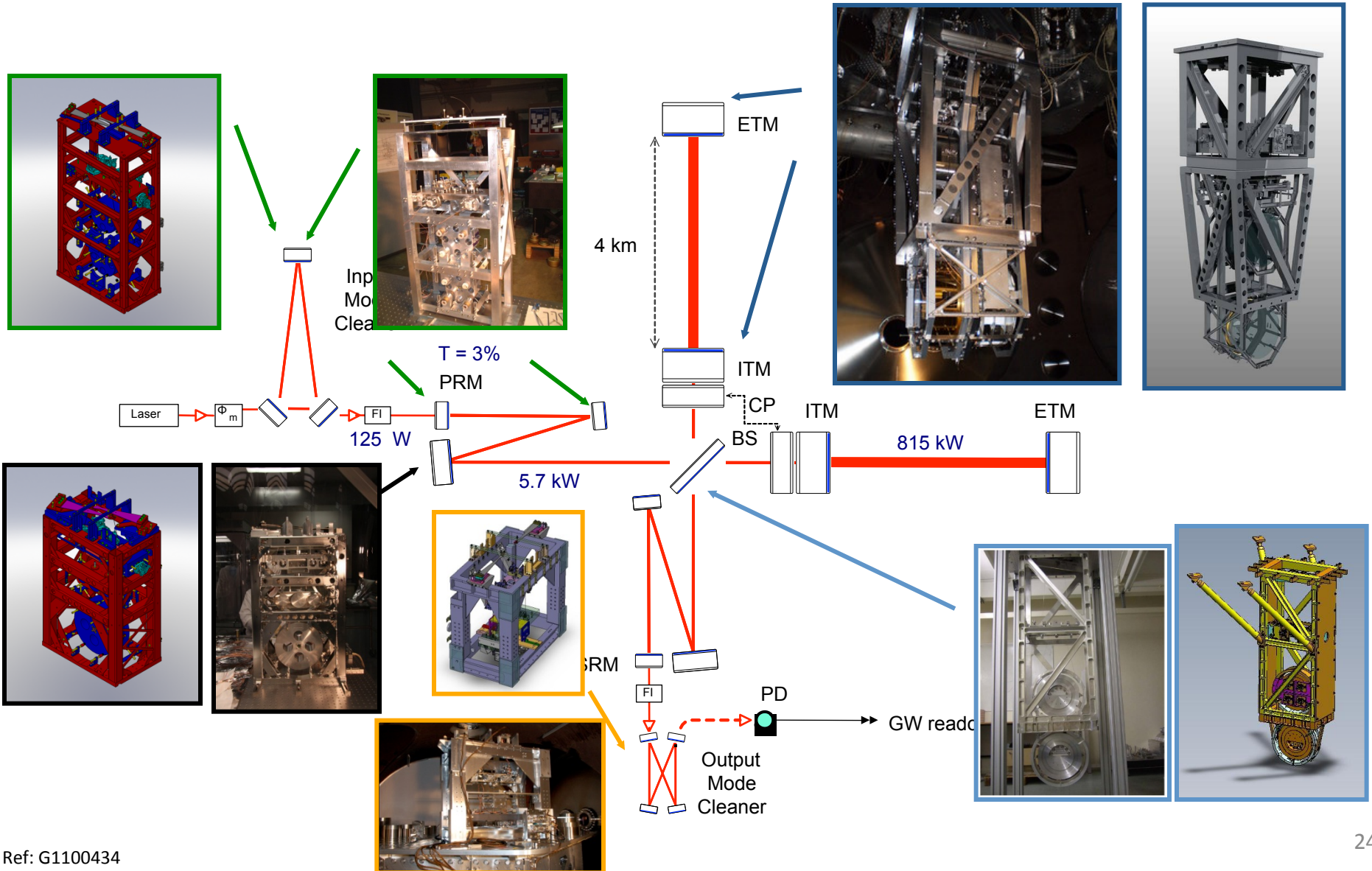
- Modal damping provides an intuitive way to optimize a highly coupled, many DOF system, with strict noise performance.
- Real-time or adaptive tuning possible by adjusting gains on each mode.
- Future work to involve implementation on a true Advanced LIGO interferometer.

Backups



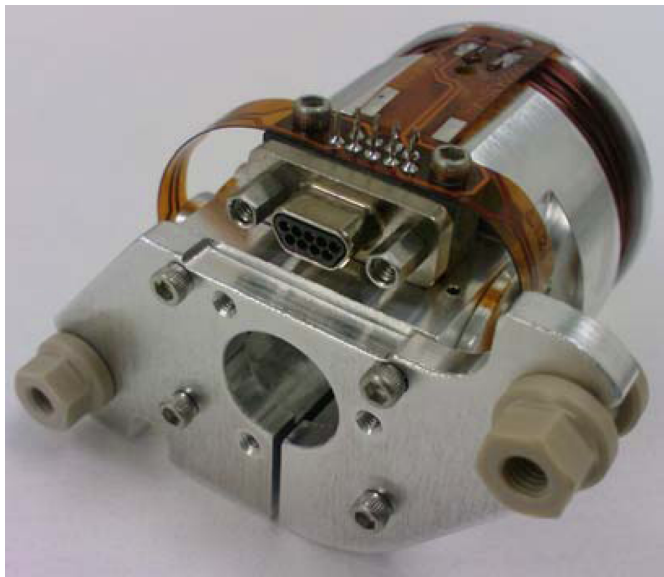
LIGO spans 16 km². Cambridge, MA covers 16.65 km² (wikipedia http://en.wikipedia.org/wiki/Cambridge,_Massachusetts).

Five Pendulum Designs

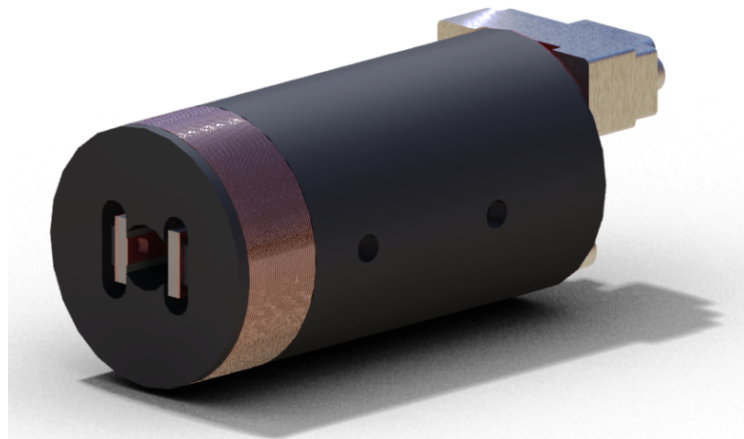




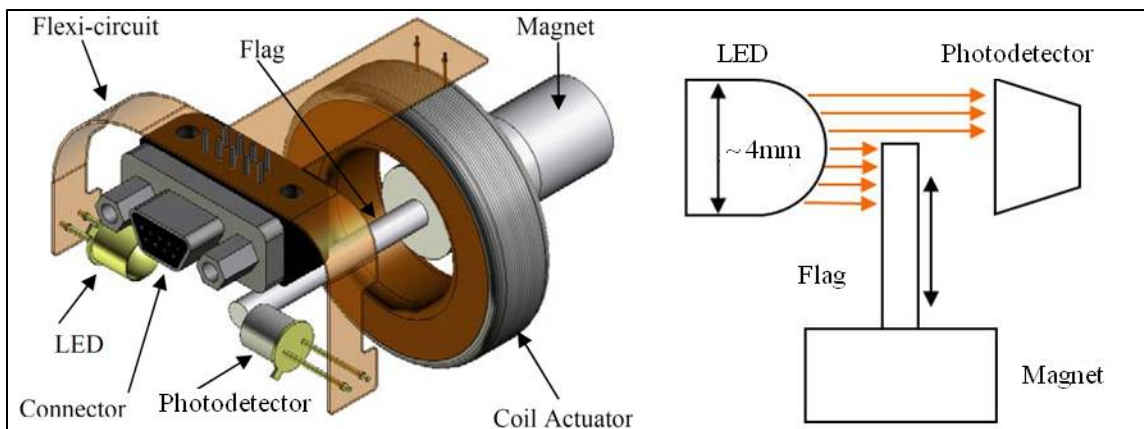
Backups: Optical Sensor ElectroMagnet (OSEM)



Birmingham OSEM (BOSEM)



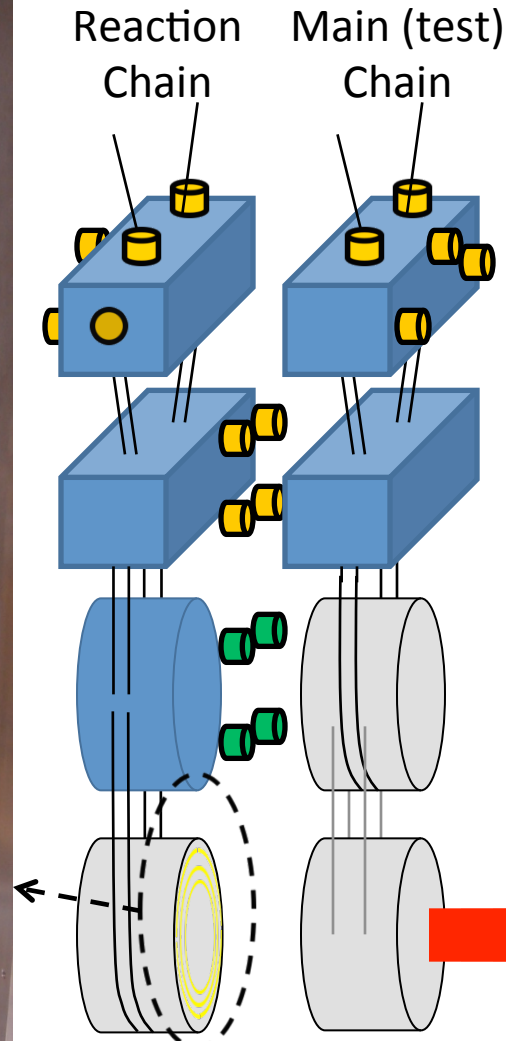
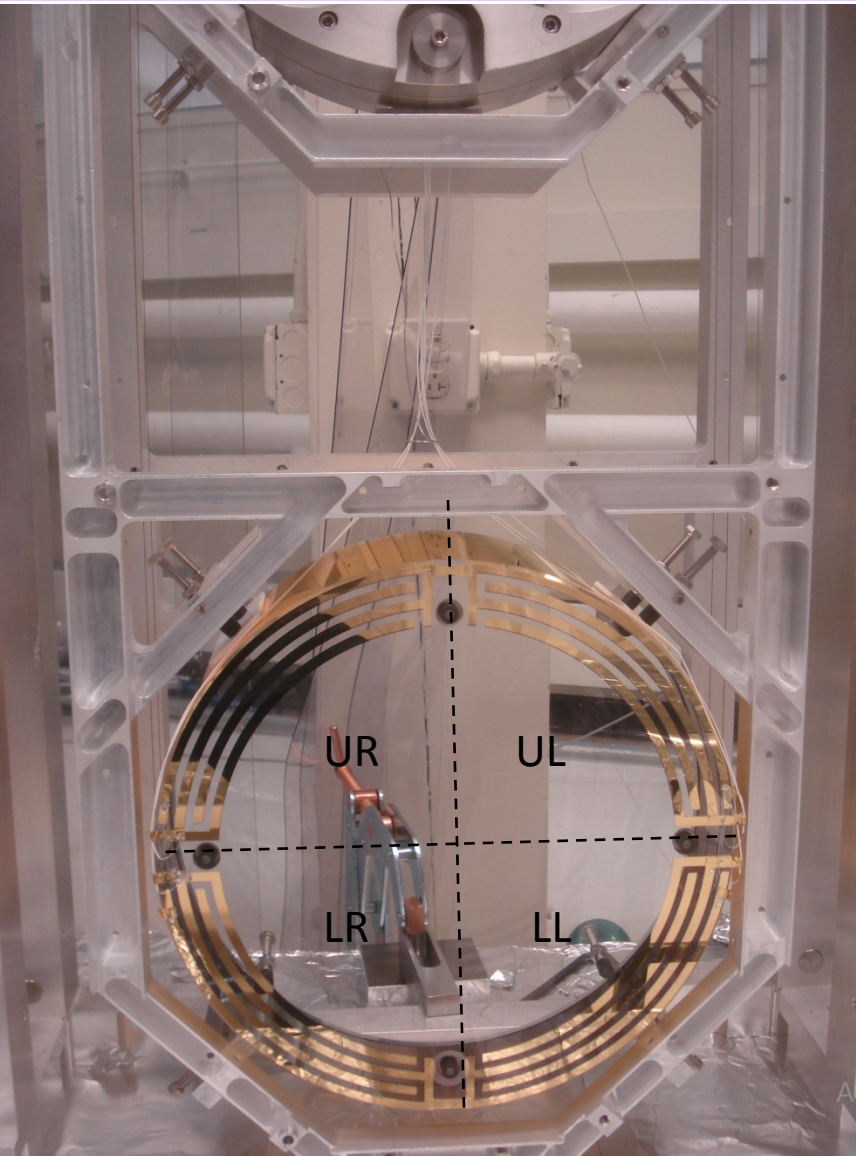
Advanced LIGO OSEM (AOSEM)
- modified iLIGO OSEM



BOSEM Schematic

- Magnet Types (M0900034)
- BOSEM – 10 X 10 mm, NdFeB , SmCo
 - AOSEM – 2 X 3 mm, SmCo
 - AOSEM – 2 X 6 mm, SmCo
 - AOSEM – 2 X 0.5 mm, SmCo

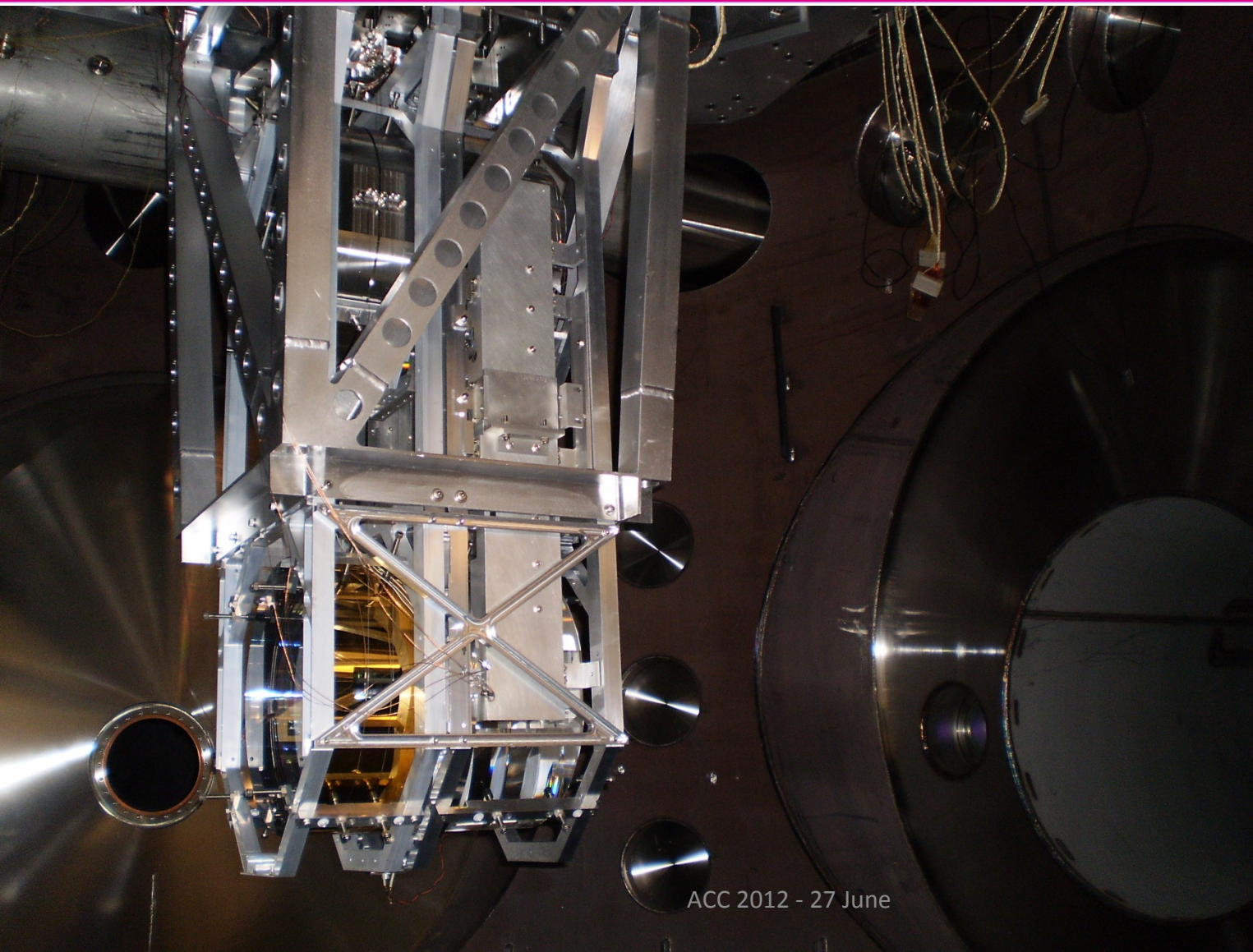
Backups: Quadruple Suspension ESD



The electrostatic drive (ESD) acts directly on the test ITM and ETM test masses.

- $\pm 400 \text{ V}$ ($\Delta V 800 \text{ V}$)
 $\approx 100 \mu\text{N}$
- Each quadrant has an independent control channel
- Common bias channel over all quadrants

Backups: Quadruple Suspension



MIT
monolithic
quad in BSC

June 2010