

LASER INTERFEROMETER GRAVITATIONAL WAVE OBSERVATORY  
-LIGO-  
CALIFORNIA INSTITUTE OF TECHNOLOGY  
MASSACHUSETTS INSTITUTE OF TECHNOLOGY

Technical Note    LIGO-T040156- 00-    E    9/22/2004

**Mirror Reflective Phase-Maps for FFT Simulation**

Xiao Xu (Caltech 2004 SURF),  
Hiroaki Yamamoto (Caltech LIGO)

This is an internal working note  
of the LIGO Project.

**California Institute of Technology**  
**LIGO Project - MS 18-34**  
**Pasadena CA 91125**  
Phone (626) 395-2129  
Fax (626) 304-9834  
E-mail: [info@ligo.caltech.edu](mailto:info@ligo.caltech.edu)

**Massachusetts Institute of Technology**  
**LIGO Project - MS NW17-161**  
**Cambridge, MA 02139**  
Phone (617) 253-4824  
Fax (617) 253-4824  
E-mail: [info@ligo.mit.edu](mailto:info@ligo.mit.edu)

WWW: <http://www.ligo.caltech.edu/>

[file /home/xxu/public\\_html/pdf/T040156.pdf](file:/home/xxu/public_html/pdf/T040156.pdf)

**LIGO-T040154-00-E**

## Introduction

LIGO is currently in the commissioning phase. The scientists and engineers at the sites are constantly finetuning the interferometers. Despite the high demanding of precision and sensitivity, LIGO, composed of six mirrors, has various innate defects; one of them is the imperfection of the mirror surfaces. As designed, all mirrors should have perfectly smooth surfaces. However, as manufactured, mirror surfaces have a roughness of the order of nanometer. No one knows what kind of noise will this surface roughness introduces. Yet the people at the sites have observed some mysterious noises, which they suspect are attribute to the imperfection of mirror surfaces.

To assist the efforts at the sites, several computer programs are wrote to simulate the operation of LIGO. FFT program [1] is one of such programs, which staticly simulate the steady state of the E&M fields within the interferometer. Thus, FFT program is an ideal tool to test the effects of the imperfection of mirror surfaces. In order to model actual mirrors at LIGO, FFT program needs the input of reflective phase-maps and transmissive phase-maps. However, due to technical issues, transmissive phase-maps are very difficult, if not impossible, to measure. Only the reflective phase-maps have been measured at Caltech [2]. Yet, we guess that the effects of transmissive phase-maps have the same order of magnitude as that of reflective ones. Therefore, the reflective phase-maps are of great importance to the estimation of noises caused by mirror surface aberration.

## Data Conversion

For each mirror, reflective phase-map of the central region, a disc with diameter 150 ( $mm$ ), has been measured. The measured phase-maps can be represented by  $480 \times 576$  matrices<sup>1</sup>. The size of every cell of the matrices is  $0.266800 \times 0.311356$  ( $mm^2$ ). Since, FFT program uses *Fast Fourier Transformation*, the input data must be square matrix with dimensions of powers of 2, such as matrices of dimensions  $128 \times 128$  or  $256 \times 256$ . Therefore, the measured data need to be converted into the required format.

As indicated by Fig 1, the scheme which converts an  $n \times m$  matrix, with a cell size  $dm \times dn$ , to a coarser  $N \times N$  matrix, with cell size  $dN \times dN$ , is as follows [3]:

1. Impose the coarser matrix onto the finer one with the reference of a coordinate system.

---

<sup>1</sup>The data outside the measured circular region are marked as “Bad”. Before processing the data conversion, all values marked as “Bad” are substituted by zero.

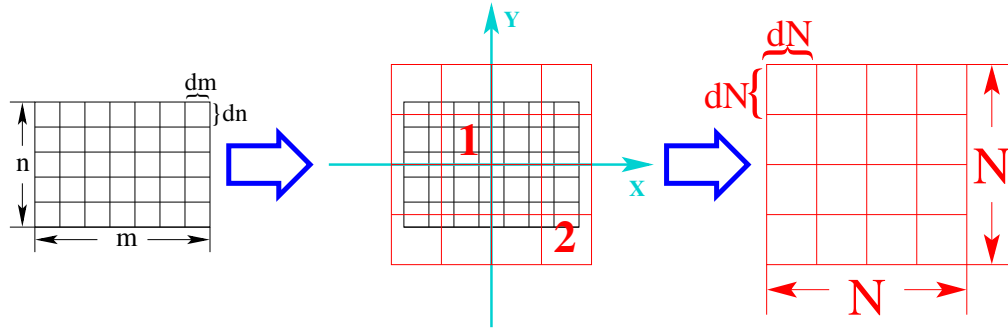


Figure 1: The Schematic Diagram of Conversion of Input Matrices

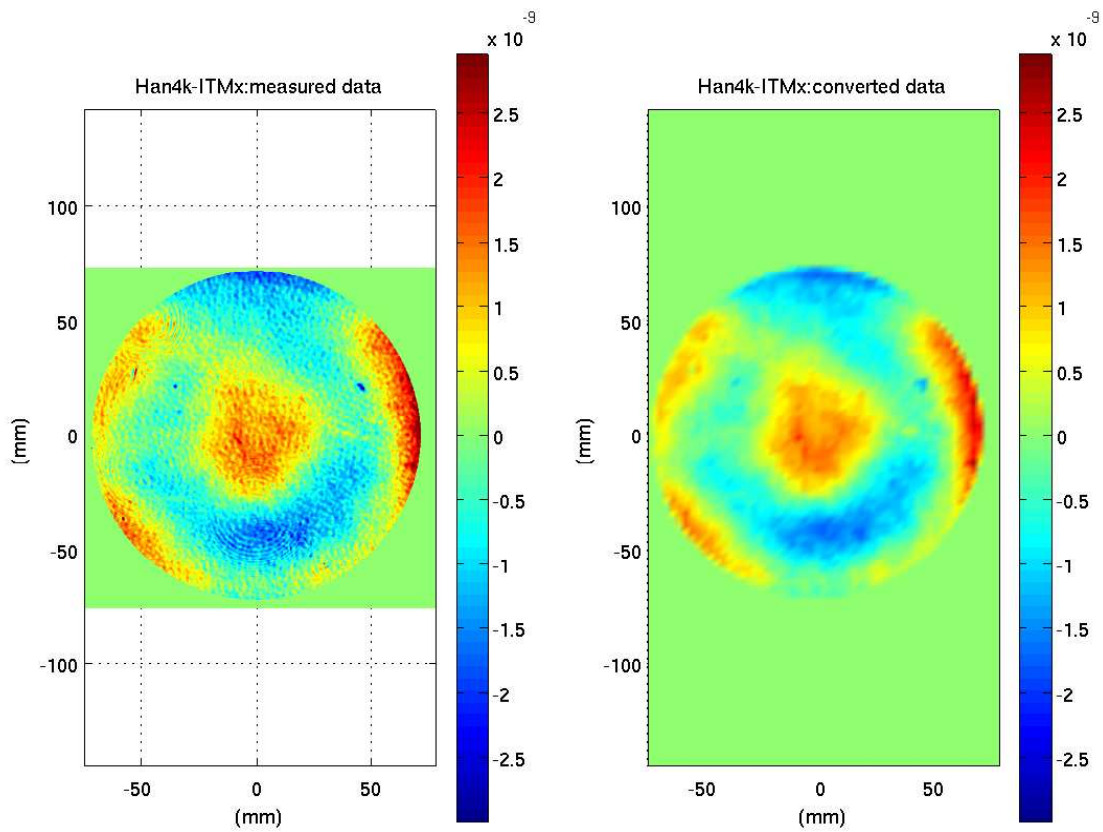


Figure 2: Example of Data conversion

2. If a cell of the coarser matrix is completely inside the finer matrix, such as cell 1 in Fig 1, then the value assigned to this cell is the weighted average of the values of the finer matrix enclosed by this cell. The weight of each value of the finer matrix is proportional to the area of this value which is within the boundary of the cell of the coarser matrix.
3. If a cell of the coarser matrix is partially or completely outside the finer matrix, such as cell 2 in Fig 1, then set the value of the area outside the finer matrix to be zero, and assign the cell weighted average of the values enclosed by it.

By such method, the measured phase-maps,  $480 \times 576$  matrices with cell size  $0.266800 \times 0.311356$  ( $mm^2$ ), have been converted to the required formats,  $128 \times 128$  matrices with cell size  $2.734375 \times 2.734375$  ( $mm^2$ )<sup>2</sup> (Fig 2).

## Extrapolation

In order to model an actual mirror with the coated circular area with 24 ( $cm$ ) diameter, the measured data within a circular area with 15 ( $cm$ ) diameter need to be extrapolated. Strictly speaking, there is no way to predict the phasemap in the unmeasured region. The best one can do is to extrapolate reasonably based on an educated guess<sup>3</sup>.

## Zernike Fit

The basic assumption is that the smooth structures of the phasemap up to 2nd order, i.e., piston, x-tilt, y-tilt, power, astigmatisms, are universal on the entire mirror surface. The measured phasemap is fit by using the 37 Wyko polynomials, and the coefficient of the first 6 Wyko polynomials [4] are used to smoothly extrapolate the phasemap. The Wyko polynomials is a particular set of the Zernike Polynomials [4], which are commonly used to describe the surface aberration of optics. The fitting coefficients can be found by minimizing  $\chi^2$ , that's to minimize the following:

$$\chi^2 = \sum_i \sum_j \left( \frac{f(x_i, y_j) - z(x_i, y_j)}{\sigma_{ij}^2} \right)^2, \text{ with} \quad (1)$$

---

<sup>2</sup>The converted matrix covers a  $35 \times 35$  ( $cm^2$ ) area.

<sup>3</sup>The systematic errors of simulation results are estimated by comparing results using phasemaps based on this extrapolation and results using phasemaps based on another assumption. This will be discussed in a separated document LIGO-T040173-00-E [7].

$$f(x_i, y_j) = \sum_k a_k w_k(\hat{x}_i, \hat{y}_j), \quad (2)$$

$$\hat{x}_i = \frac{x_i}{r}, \quad \hat{y}_j = \frac{y_j}{r}, \quad (3)$$

where  $(i, j)$  is the index of measured data points,  $z(x, y)$  is the measured data,  $f(x, y)$  is the fitting function with  $\{a_k\}$  the fitting coefficient,  $\sigma_{ij}$  is the standard deviation of each measurement,  $w_k(\hat{x}, \hat{y})$  is the Wyko Polynomials serving as the basis of the fitting, and  $r$  is the radius of the measured phasemap, ie. 7.5(cm).

Differentiate equation (1) with respect to  $\{a_k\}$ , it becomes

$$\frac{\partial \chi^2}{\partial a_k} = \sum_i \sum_j 2w_l(\hat{x}_i, \hat{y}_j) \left( \frac{f(x_i, y_j) - z(x_i, y_j)}{\sigma_{ij}^2} \right). \quad (4)$$

To minimize  $\chi^2$ , equation (4) is set to equal zero. Thus,

$$\sum_i \sum_j w_l(\hat{x}_i, \hat{y}_j) \left( \frac{f(x_i, y_j) - z(x_i, y_j)}{\sigma_{ij}^2} \right) = 0 \quad (5)$$

$$\Rightarrow \sum_i \sum_j \frac{w_l(\hat{x}_i, \hat{y}_j) f(x_i, y_j)}{\sigma_{ij}^2} = \sum_i \sum_j \frac{w_l(\hat{x}_i, \hat{y}_j) z(x_i, y_j)}{\sigma_{ij}^2} \quad (6)$$

Plug equation (2) into equation (6),

$$\sum_i \sum_j \left[ \sum_k a_k w_k(\hat{x}_i, \hat{y}_j) \right] \frac{w_l(\hat{x}_i, \hat{y}_j)}{\sigma_{ij}^2} = \sum_i \sum_j \frac{w_l(\hat{x}_i, \hat{y}_j) z(x_i, y_j)}{\sigma_{ij}^2} \quad (7)$$

$$\Rightarrow \sum_k a_k \left[ \sum_i \sum_j \frac{w_l(\hat{x}_i, \hat{y}_j) w_k(\hat{x}_i, \hat{y}_j)}{\sigma_{ij}^2} \right] = \sum_i \sum_j \frac{w_l(\hat{x}_i, \hat{y}_j) z(x_i, y_j)}{\sigma_{ij}^2} \quad (8)$$

$$\Rightarrow \begin{cases} \sum_k a_k \left[ \sum_i \sum_j \frac{w_1(\hat{x}_i, \hat{y}_j) w_k(\hat{x}_i, \hat{y}_j)}{\sigma_{ij}^2} \right] = \sum_i \sum_j \frac{w_1(\hat{x}_i, \hat{y}_j) z(x_i, y_j)}{\sigma_{ij}^2} \\ \vdots \\ \sum_k a_k \left[ \sum_i \sum_j \frac{w_l(\hat{x}_i, \hat{y}_j) w_k(\hat{x}_i, \hat{y}_j)}{\sigma_{ij}^2} \right] = \sum_i \sum_j \frac{w_l(\hat{x}_i, \hat{y}_j) z(x_i, y_j)}{\sigma_{ij}^2} \\ \vdots \end{cases} \quad (9)$$

Equation system (9) can be represented in a matrix form,

$$\mathbf{WA} = \mathbf{Z}, \quad (10)$$

where

$$\mathbf{W} = \begin{pmatrix} W_{11} & W_{12} & \dots \\ W_{21} & W_{22} & \dots \\ \vdots & \vdots & \ddots \end{pmatrix}, \text{ with} \quad (11)$$

$$W_{lk} = \sum_i \sum_j \frac{w_l(x_i, y_j) w_k(\hat{x}_i, \hat{y}_j)}{\sigma_{ij}^2}; \quad (12)$$

$$\mathbf{A} = \begin{pmatrix} a_1 \\ a_2 \\ \vdots \end{pmatrix}; \quad (13)$$

and

$$\mathbf{Z} = \begin{pmatrix} Z_1 \\ Z_2 \\ \vdots \end{pmatrix}, \text{ with} \quad (14)$$

$$Z_l = \sum_i \sum_j \frac{w_l(\hat{x}_i, \hat{y}_j) z(x_i, y_j)}{\sigma_{ij}^2}. \quad (15)$$

Solve matrix equation (10), one can obtain the fitting coefficient  $\{a_k\}$ .

There is one small complication for the Livingston4K beam splitter(LLO BS). Since it was tilted  $45^\circ$  on the yaw direction when measured, when processing this mirror, one need to rotate it back, that's to stretch the x axis  $\sqrt{2}$  times. Thus for this mirror, substitute all  $x_i$  in the above formulae with  $\sqrt{2}x_i$ .

After obtaining the smooth structure of the phasemap, extrapolate it to the full extension of a mirror, (Fig 3). This is the base of extrapolation; the extrapolated high frequency surface aberrations will be built onto it later.

## High Spatial Frequency Surface Aberration Modeling

After the separation of the low spacial frequency structure, the remaining high spacial frequency components of the measured phasemaps need to be extrapolated. There are two issues to supplement this extrapolation. One is a smoothness of the connection between the measured region and the extrapolated region. The other is the generation of the high spacial frequency component, or random noise, in the extrapolated region.

### Noise Smearing

In the final phasemap with extrapolation, the measured data are retained when available. To ensure a smooth extrapolation, these measured data within 15 (cm) diameter are extended to 1 (cm) to outer side using the following algorithm (Fig 4):

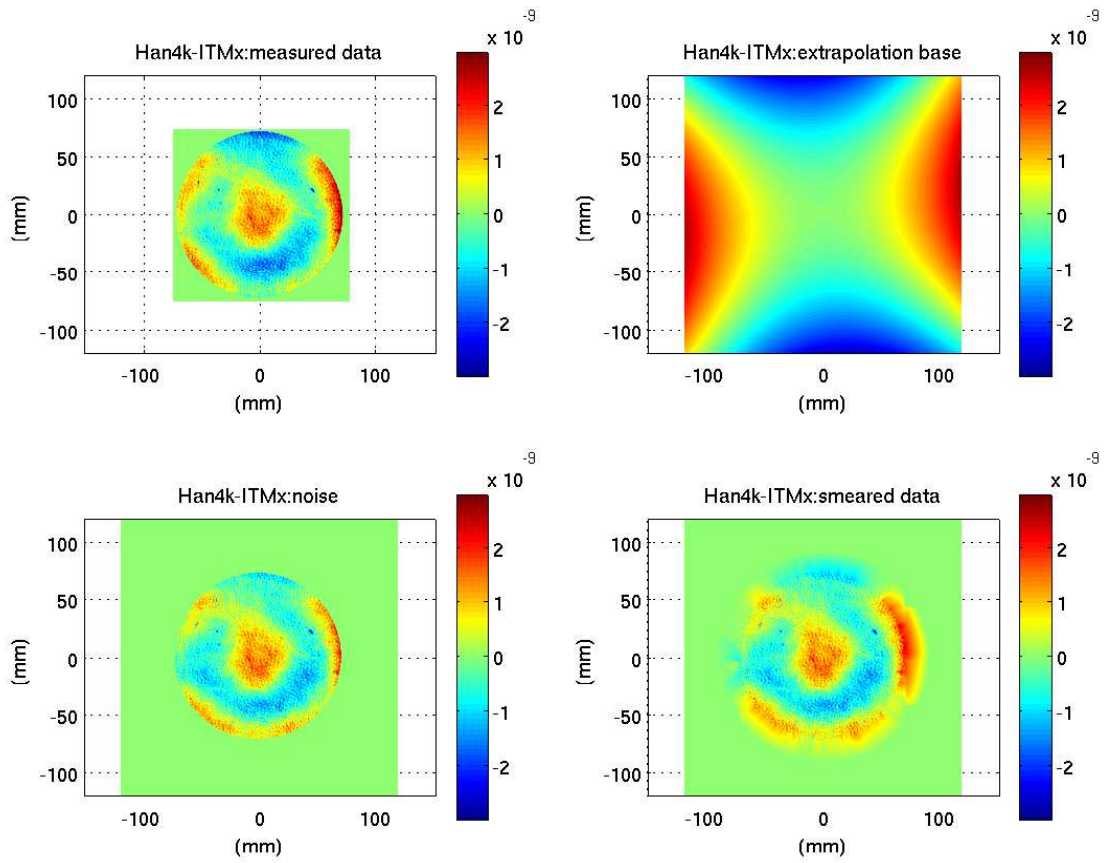


Figure 3: Example of Separation of base and noise and Smear of the noise

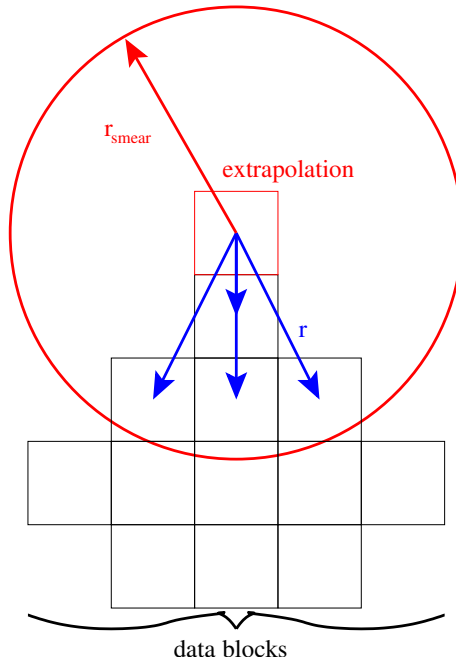


Figure 4: Scheme of smearing the noise

1. Set a radius  $r_{smeared}$  so that the value of the extrapolated datum will be a weighted average of the values fall in to the circular region of  $r_{smeared}$ .
2. Say the distance from the extrapolation to an existed datum block is  $r$ , then the weight assigned to the value of this datum block is  $\frac{1}{(r^2+1)^2}$ .
3. After adding up all the weighted values, normalize the sum, that's to divide the weighted sum by the sum of all weights,  $\frac{\sum_i \frac{z(r_i)}{(r_i^2+1)^2}}{\sum_i \frac{1}{(r_i^2+1)^2}}$ .
4. Apply a weight on to the smeared noise so that the noise gradually fades to zero while extending into the extrapolated region.

In that region, 1cm ring, this extrapolation based on the data averaging will have the effect that after superpose to the Zernike fit of the low frequency component, at the boundary of the measured data, the extrapolation is almost equal to the data and, at the outer boundary of the 1cm ring, it is almost equal to the Wyko polynomial based extrapolation. One resultant profile of smeared noise is shown in Fig 3.



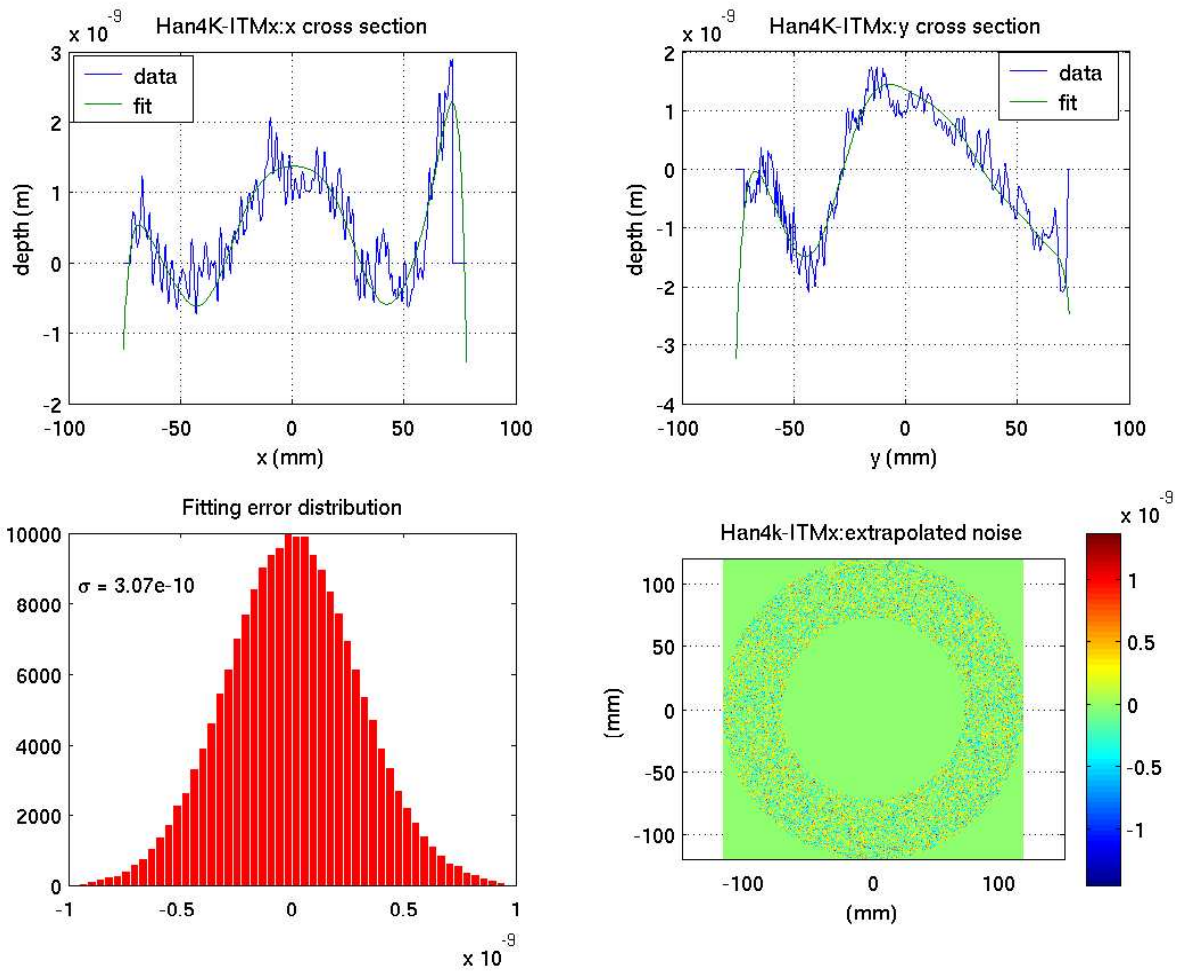


Figure 5: Example of Extrapolation of noise

## Random Noise Generation

The random noise was estimated as follows. In the measured region, the residual of the data after subtracting the Wyko fit using 37 terms were fit by a gaussian form to find the width (Fig 5). The spacial frequency distribution in the frequency space is flat if random noises with this width are assigned to the data cell. By the assumption that the random noise in the extrapolated region is of the same distribution, the extrapolated random noise is produced by a normally distributed random number generator with the mean zero and standard deviation the width obtained from above (Fig 5).

However, when comparing the extrapolated noise to the measured noise, it's apparant that the extrapolation is of much higher frequency than the measured one (Fig 6). In order to

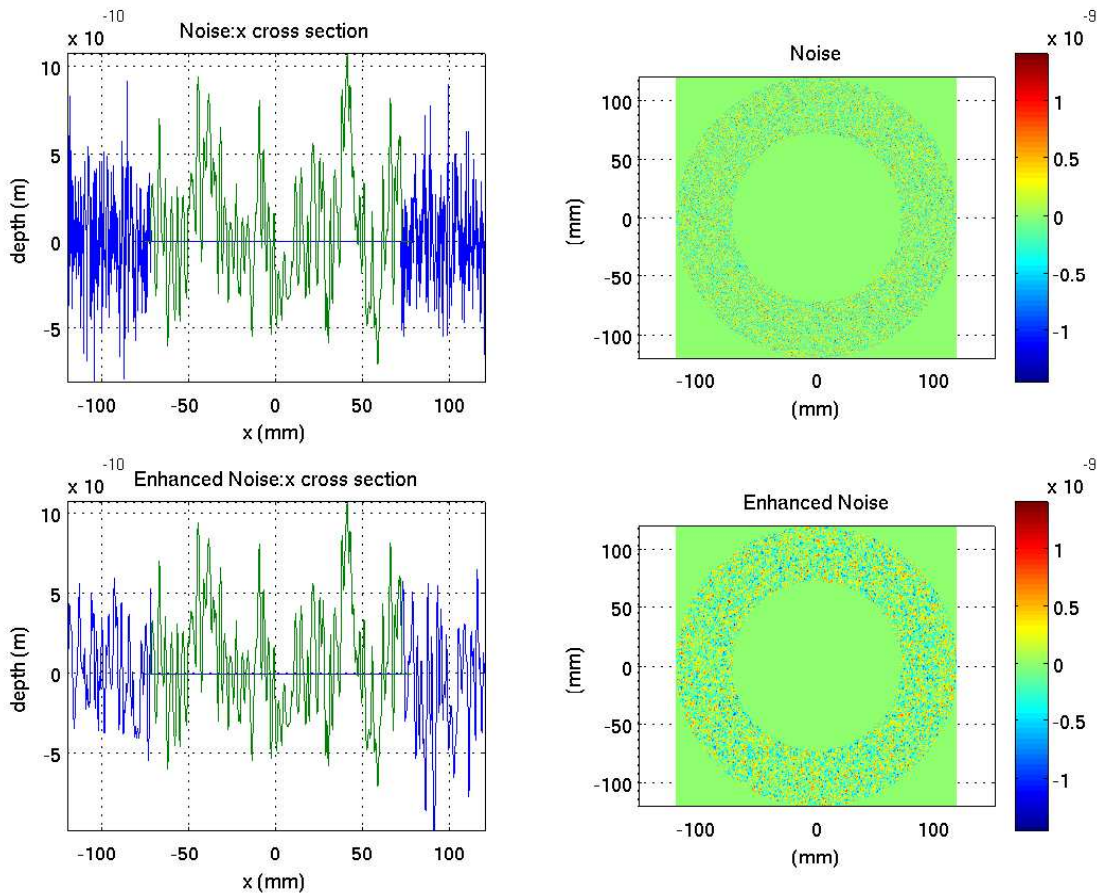


Figure 6: Example of the final noise extrapolation

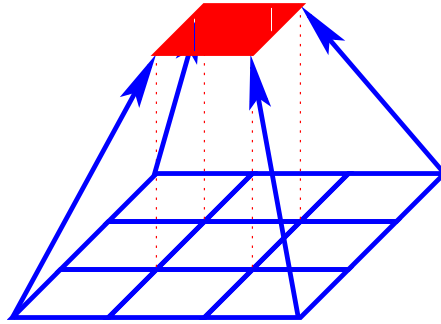


Figure 7: Scheme of integrating the noise

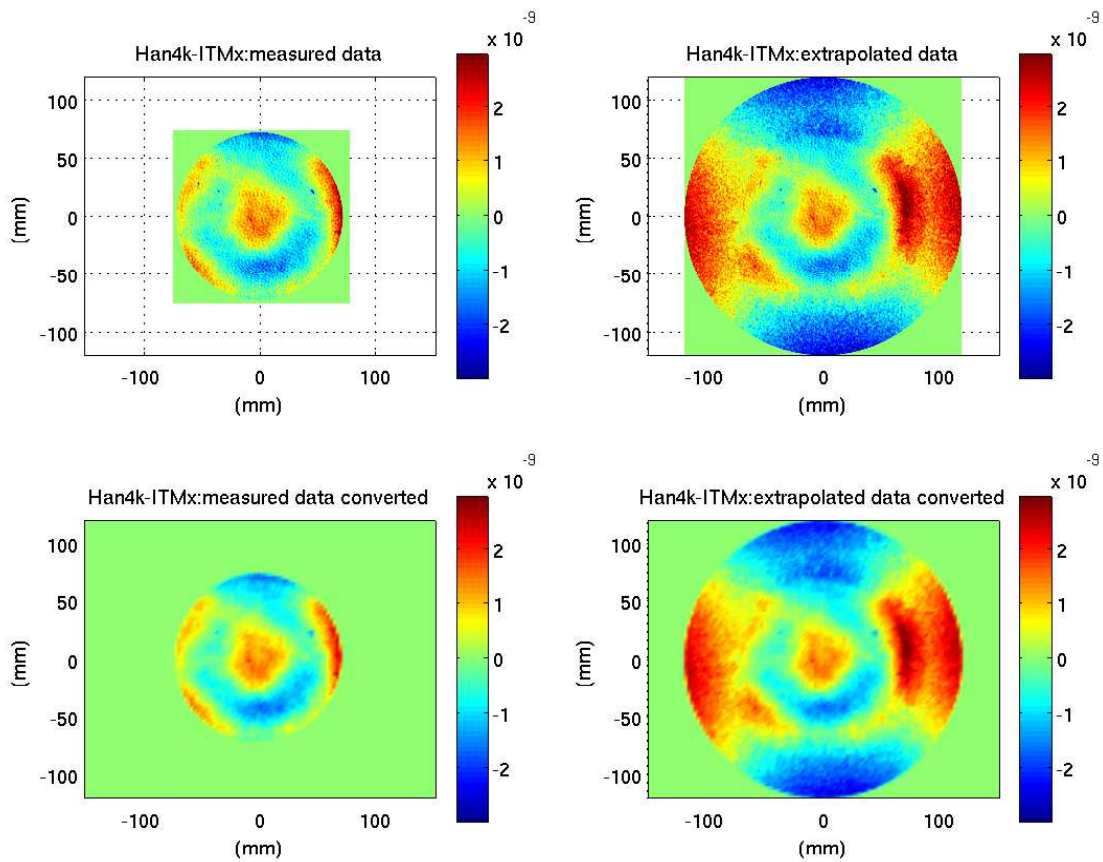


Figure 8: Example of final extrapolation and conversion

enhance the low frequency component of the extrapolation, the noise is integrated twice<sup>4</sup> as is described below (Fig 7):

1. For each cell in a phase-map, form a  $3 \times 3$  matrix with this cell as the center and its immediate 8 neighbouring cells as the boundary.
2. Assign the mean of the values of the nine cells of the  $3 \times 3$  matrix to the cell in the center.

After this process, the extrapolation becomes much closer to the measurement (Fig 6).

After superposing the extended base, the smeared measurement, and the extrapolated random noise, and setting all values outside the mirror range to be zero, the final extrapolation of the measured phase-map (Fig 8) is accomplished. After the conversion of the extrapolation by the process indicated in Fig 1, the phasemaps are but one step from being done.

## Tilt Removal

The beam profile of incident laser of LIGO is Gaussian, so most of the beam's power is concentrated in the central region, characterized by the diameter of this region, or Spot Size (Fig 9). Therefore, the central region, characterized by spot size, of the mirror surface is of uncomparable importance. Thus the final step of preparing a phasemap is to maximally reduce the tilts on its central region, together with piston. This task is accomplished by a Tilt Removal program, developed by B.Bochner [1].

---

<sup>4</sup>Integration has an effect to smooth function out. We can consider an simple 1-D case. For a random function  $f(x)$ , after Fourier decomposition, it becomes

$$f(x) = \sum_k a_k e^{ikx} ,$$

where  $k$  is the wave number, or spatial frequency. After integration, it becomes

$$\int f(x)dx = \sum_k \frac{a_k}{ik} e^{ikx} .$$

Observe that  $k$  comes to the denominator, and therefore the smaller the frequency is, the larger the weight of this frequency becomes. Thus, the high frequency components are subdued and the low frequency components are enhanced.

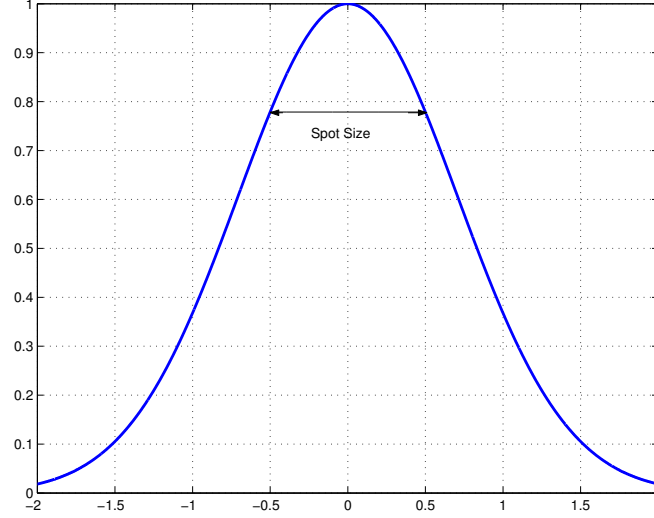


Figure 9: Profile of incident beam with its spot size

## Reflection

To obtain the information of piston and tilts, first, one creates a beam of perfect  $TEM_{00}$  mode [6], characterized by the interested spot size  $w_0$  and wavelength  $\lambda = 1.064$  ( $\mu m$ ). This beam will be impinged onto a phasemap, and the reflected beam will contain the desired information. Before that, the phasemap needs modification. Originally, the value of every cell of the phasemap is the depth,  $z$ , at this location respect to a reference plain. To accomodate the calculation of reflection, the depth  $z$  needs to be converted to phase  $\phi$  by the following formula,

$$\phi(x, y) = e^{-i2kz(x,y)}, \text{ with} \quad (16)$$

$$k = \frac{2\pi}{\lambda}. \quad (17)$$

Multiply the modified phasemap to the incident beam, one can obtain the reflected beam profile.

## Modal Decomposition

To extract the information of piston and tilts from the reflected beam profile,  $Refl$ , one needs to apply the technique of modal decomposition [5][6]. The basic idea is that any beam profile

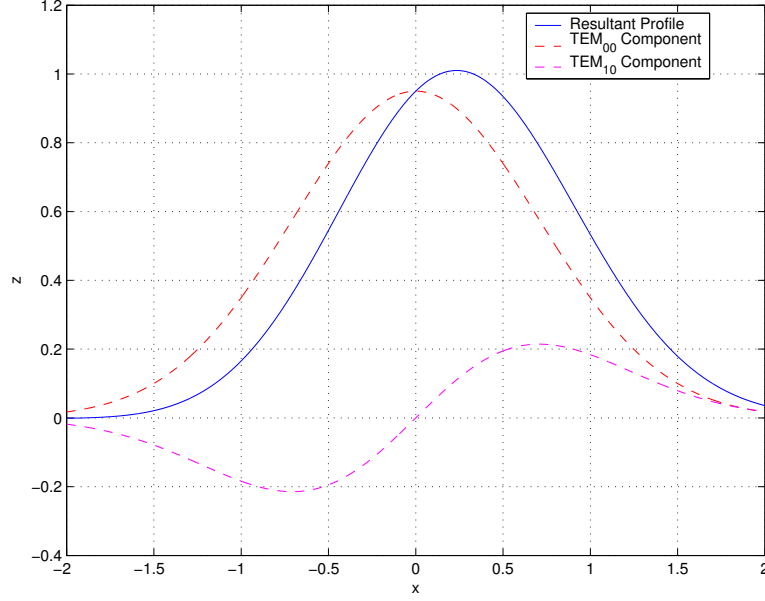


Figure 10: Example of the modal decomposition

can be described as a superposition of the Gauss-Hermite Modes,  $TEM_{mn}$ , as in Fig 10. Since  $TEM_{mn}$  is an orthonormal base, any superposition of these modes can be decomposed by the manner similar to Fourier Decomposition.

The reflection of the incident beam can be considered as an operation on the beam. Any operation can be expressed by an operation matrix [6]:

$$\mathbf{M}^{\text{Op}} = \begin{pmatrix} M_{00,00} & M_{10,00} & M_{01,00} & \dots \\ M_{00,10} & M_{10,10} & M_{01,10} & \dots \\ M_{00,01} & M_{10,01} & M_{01,01} & \dots \\ \vdots & \vdots & \vdots & \ddots \end{pmatrix}, \text{ with} \quad (18)$$

$$M_{nm,kl} = \int_{-\infty}^{\infty} TEM_{nm}(x, y, z)^* \mathbf{Op}[TEM_{kl}(x, y, z)] dx dy, \quad (19)$$

where  $A^*$  is the complex conjugate of  $A$ . Since the misalignments are very slight, the greatest effect are in the modes of the lowest order. Thus, all one needs to do is to check the first 3 modes,  $TEM_{00}$ ,  $TEM_{01}$ , and  $TEM_{10}$ . And therefore, the operation matrix will be simplified to,

$$\mathbf{M}^{\text{Op}} = \begin{pmatrix} M_{00,00} & M_{10,00} & M_{01,00} \\ M_{00,10} & M_{10,10} & M_{01,10} \\ M_{00,01} & M_{10,01} & M_{01,01} \end{pmatrix}. \quad (20)$$

When one apply the matrix in (20) to an incident beam of  $TEM_{00}$  mode, the following will be obtained,

$$\begin{aligned}
\mathbf{Op}[TEM_{00}] &= \begin{pmatrix} M_{00,00} & M_{10,00} & M_{01,00} \\ M_{00,10} & M_{10,10} & M_{01,10} \\ M_{00,01} & M_{10,01} & M_{01,01} \end{pmatrix} \begin{pmatrix} a_{00} \\ a_{10} \\ a_{01} \end{pmatrix} \\
&= \begin{pmatrix} M_{00,00} & M_{10,00} & M_{01,00} \\ M_{00,10} & M_{10,10} & M_{01,10} \\ M_{00,01} & M_{10,01} & M_{01,01} \end{pmatrix} \begin{pmatrix} 1 \\ 0 \\ 0 \end{pmatrix} \\
&= \begin{pmatrix} c_{00} \\ c_{10} \\ c_{01} \end{pmatrix}, \tag{21}
\end{aligned}$$

where  $a_{nm}$  is the coefficient of  $TEM_{nm}$  component in the incident beam;  $c_{nm}$  is the coefficient of  $TEM_{nm}$  component in the reflected beam.

### Piston

The information of piston,  $d$ , can be extracted from the coefficient of  $TEM_{00}$  mode. Suppose one can write the phasemap  $z(x, y)$  as the superposition of a series *Zernike Polynomials*,  $\{Z_{nm}(x, y)\}$ [4] as

$$z(x, y) = \sum_{n,m} a_{nm} Z_{nm}(x, y), \tag{22}$$

where  $a_{00}Z_{00} = a_{00} \times 1 = a_{00} = d$ . Then by equation (16), the reflection beam is

$$Ref l = TEM_{00} e^{-i2kz(x,y)} = TEM_{00} e^{-i2k \sum_{n,m} a_{nm} Z_{nm}(x,y)}. \tag{23}$$

Assume that the dominant effect comes from the lowest order of Zernike polynomials, then equaiton (23) can be simplified as

$$Ref l \approx TEM_{00} e^{-i2ka_{00}Z_{00}(x,y)} = TEM_{00} e^{-i2kd}. \tag{24}$$

Thus, in order to extract information of piston, we need to do the following,

$$\begin{aligned}
c_{00} &= \int Ref l \cdot TEM_{00}^* dx dy \\
&= \int TEM_{00} e^{-i2kd} \cdot TEM_{00}^* dx dy \\
&= e^{-i2kd} \int TEM_{00} \cdot TEM_{00}^* dx dy \\
&= Ae^{i\alpha} \tag{25}
\end{aligned}$$

$$\Rightarrow d = \frac{\alpha}{-2k} \tag{26}$$

where  $k$  is the wave number which defined in equation (17).

### Tilts

The information of tilts, tilt on the pitch direction  $\theta_X$  and tilt on the yaw direction  $\theta_Y$ , can be extracted from the  $TEM_{01}$  and  $TEM_{10}$  respectively. Say if the mirror tilts  $\theta_X$  on the pitch direction and  $\theta_Y$  on the yaw direction, when the incident beam impinges from the front surface of the mirror, the operational matrix for the reflected beam is [5]:

$$\mathbf{M}^{Op} = \begin{pmatrix} 1 & -2i\hat{\Theta}_Y & 2i\hat{\Theta}_X \\ -2i\hat{\Theta}_Y & 1 & 0 \\ 2i\hat{\Theta}_X & 0 & 1 \end{pmatrix}^5, \quad (27)$$

with

$$2\hat{\Theta}_{X(Y)} = 2\pi w \sin \theta_{X(Y)} / \lambda = wk \sin \theta_{X(Y)}, \quad (28)$$

where  $w$  is the spot size and  $k$  is the wave number. Apply matrix (27) to the incident  $TEM_{00}$  beam, one will obtain

$$\begin{pmatrix} 1 & -2i\hat{\Theta}_Y & 2i\hat{\Theta}_X \\ -2i\hat{\Theta}_Y & 1 & 0 \\ 2i\hat{\Theta}_X & 0 & 1 \end{pmatrix} \begin{pmatrix} 1 \\ 0 \\ 0 \end{pmatrix} = \begin{pmatrix} 1 \\ -2i\hat{\Theta}_Y \\ 2i\hat{\Theta}_X \end{pmatrix} \quad (29)$$

Clearly, the information of the tilts will be contained in the imaginary part of the coefficients of the reflected beam's  $TEM_{nm}$  components.

Thus,  $\theta_X$  and  $\theta_Y$ , can be extracte by the following fomulae:

$$c_{01} = \int Refl \cdot TEM_{01}^* dx dy = Re_{01} + iIm_{01} \quad (30)$$

$$\Rightarrow \theta_X = \frac{\sin^{-1} Im_{01}}{wk} \quad (31)$$

$$c_{10} = \int Refl \cdot TEM_{10}^* dx dy = Re_{10} + iIm_{10} \quad (32)$$

$$\Rightarrow \theta_Y = \frac{\sin^{-1} Im_{10}}{-wk} \quad (33)$$

where  $w$  is the spot size.

---

<sup>5</sup>In expression (27), all higher order terms of  $\hat{\Theta}_{X(Y)}$  are neglected.

<sup>6</sup>In the Tilt Removal program, the calculation of  $\theta_Y$  uses a slightly different formula,  $\theta_Y = \frac{\sin^{-1} b_{10}}{-wk \cos \theta_X}$ . That's because while untilting the mirror on the pitch direction, the amount of tilt on the yaw direction is changed, since the hight function  $z$  at  $(x, y)$  due to the tilt is  $z(x, y) = x \sin \theta_X + (y \cos \theta_X) \sin \theta_Y$  [1].

However, with  $\theta_X$  being so small, the extra  $\cos \theta_X$  hardly contribute any substantial difference from the results calculated by equation (33).



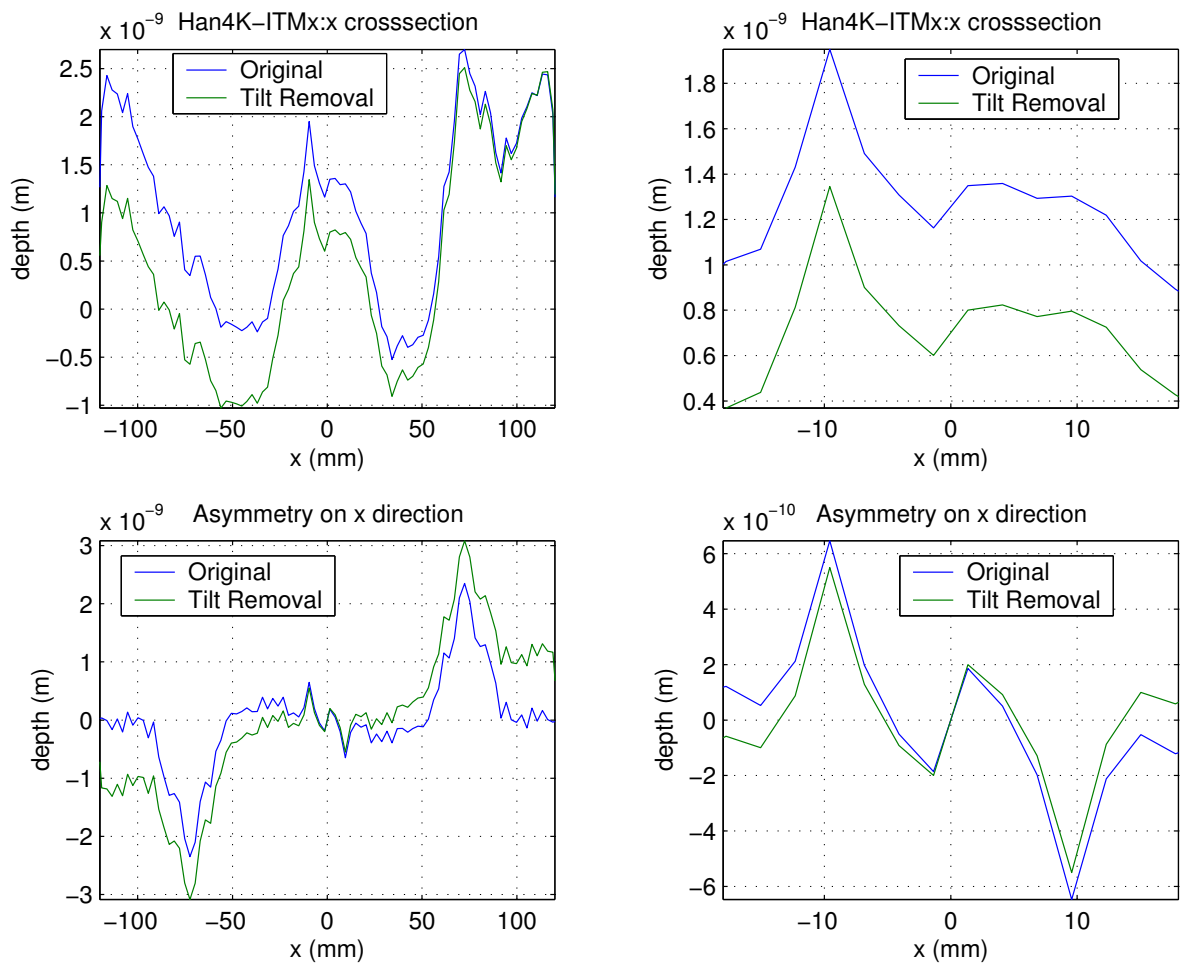


Figure 11: Example of the effect of tilt removal

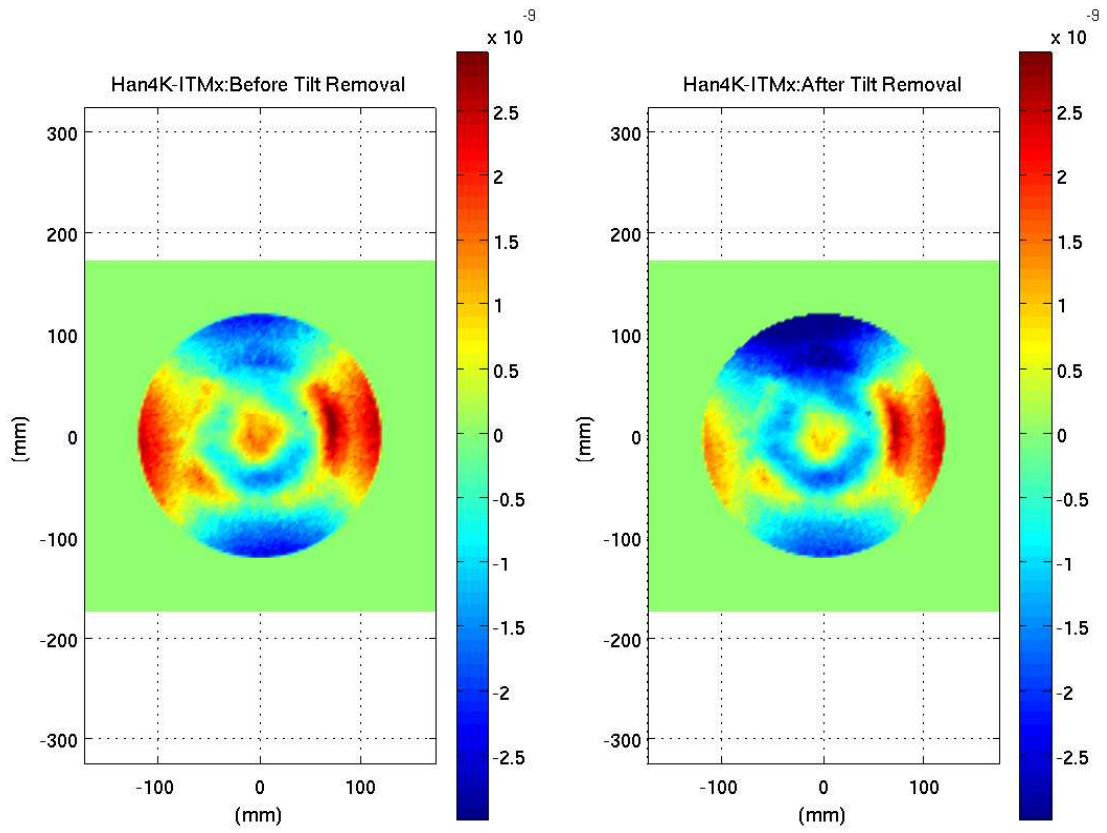


Figure 12: Example of final input phasemap to FFT program

After obtaining the piston and tilts, one can adjust the phasemap accordingly. The effect of the Tilt Removal process is indicated in Fig 11. The plots in the first row are the comparison of the phasemaps' x crosssection, which indicate that the piston is indeed somewhat removed. The plots in the second row are the comparison of the x crosssection of the odd components of the phasemaps. The plot in the first column seems to indicated that after tilt removal, the entire phasemap seems to tilt even more than the original; however, as the plot in the second column indicated, the surface within the range characterized by spot size has less tilt, or its odd components are minimized.

After this last step (Fig 12), the phasemaps are finally ready to be input to the FFT Program.

## Reference

1. B. Bochner, *Modelling the Performance of Interferometric Gravitational-Wave Detectors with Realistically Imperfect Optics*, LIGO-P980004-00-R
2. G. Billingsley,  
  
[www.ligo.caltech.edu/~gari/COCAAsBuilt.htm](http://www.ligo.caltech.edu/~gari/COCAAsBuilt.htm)
3. R. Dodda, *Impact of Imperfect Optics on the Performance of Laser Interferometer Gravitational Wave Observatory (LIGO)*, 2003
4. M. Born and E. Wolf, *Principles of Optics*, 1959
5. B. Bhawal, M. Evans, M. Rakhmanov and H. Yamamoto, *Time Domain Modal Model in End-to-End simulation package*, LIGO-T009981-03-E, 2004
6. M. Eichenfield, *Modelling and Commissioning the Wavefront Sensing Auto-Alignment System of a Triangular Mode Cleaner Cavity*, LIGO-T030234-00-D
7. B. Bhawal, X. Xu, and H. Yamamoto, *Effects of As-Built Mirror Aberrations (I)*, LIGO-T040173-04-E, 2004