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## EARTHQUAKE AND SAFETY STOP DESIGN REQUIREMENTS

			APPROVALS		
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## Scope

This document details the features and requirements for the earthquake and safety stops for a single pendulum suspension.

## **Applicable Documents**

### STOPS

D960499 LOS Chamfer Stop D970311 Safety Stop, Conductive D970312 Safety Stop, Conductive, Small D970313 Safety Stop, Conductive, Long D970562 Beamsplitter Chamfer Stop D970563 Beamsplitter Safety Stop D990690 LOS Safety Stop D990691 LOS3 Chamfer Stop D010213 40m TM Short Stop D010214 40m TM Long Stop D020527 40m TM Stop

### LOS ASSEMBLIES

D960132 Large Optic Suspension Assembly (LOS1a) for End Test Mass D970560 Large Optic Suspension Assembly (LOS1b) for Input Test Mass, 4k (ITM,4k) D970564 Large Optic Suspension Assembly (LOS1c) for Recycling Mirror,4k (RM,4k) D970572 Large Optic Suspension Assembly (LOS1d) for Input Test Mass,2k (ITM,2k) D970577 Large Optic Suspension Assembly (LOS1e) for Recycling Mirror,2k (RM,2k) D970561 Large Optic Suspension Assembly (LOS1a) for MMT3, 4k D970578 Large Optic Suspension Assembly (LOS1a) for MMT3, 2k D970505 Large Optic Suspension Assembly (LOS2a) for Beamsplitter, 4k (BS,4k) D970539 Large Optic Suspension Assembly (LOS2b) for Beamsplitter, 2k (BS,2k) D970507 Large Optic Suspension Assembly (LOS3) for Folding Mirror (FM)

### SOS ASSEMBLY

D960001 Small Optic Suspension Assembly





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#### **BAFFLES AND TARGETS**

D020421 MMT2 Beam Blocking Baffle D990490 Little Beam Dump Assemblies D980689 Mode Cleaner Baffle Assembly (MC1) D990002 Elliptical LOS Baffle Assembly (RM, ITM)

### Function

The earthquake and safety stops, called stops for brevity, serve three functions, in order:

- 1) The stops are used to **facilitate suspending and balancing** of a mass or optic. For example, Teflon-capped screws are positioned under the LOS optic prior to suspending to support the optic. The wire is strung around the optic and then the stops are slowly moved out from under the optic, so that the optic hangs by the wire. The Teflon caps rotate with little friction on the optic. Prior to final alignment in-situ, these caps are removed and replaced with vacuum compatible Viton tips. The stops are used during the balancing process to protect the optic from swinging too much.
- 2) The stops are used to **clamp** the optic(s) in place prior to transport. This type of stop must secure the optic, in its balanced position within the structure, to facilitate safe transport, either by cart or by crane. It must secure the optic throughout the structure positioning operations in the vacuum chamber.
- 3) After the suspension has been moved into position on the optical table, the stops then perform a new function. They are used to **protect** the mass or optic, and the objects around it, in the event of an earthquake or other sudden movement. Generally, the stops are placed within 0.5 to 1mm of the optic.

# Requirements

- 1) The stops must have sufficient mechanical compliance to keep impact stresses minimal on the mass or optic.
- The stops must have low runout error so that the contact point does not wander with axial adjustment. We require that the axial variation due to the runout be less than 1/10 of the intended gap or \_\_\_\_\_\_. This may be too small given a ¼-20 screw but possible for future designs
- 3) The stops must have contact geometry that is axisymmetric with respect to the axial adjustment axis.



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4) The stops must have very smooth, fine axial position adjustment. Adjustment resolution should be less than or equal to 1/10 of the gap between the end of the stop and the optic. This may be too small too....

5a) The stops must have sufficient conductivity of \_\_\_\_\_\_ to bleed off electrostatic charge, but be resistive enough not to cause eddy current damping. M. Barton

OR

5b) The stops must have a contacting tip that is the same material as the optic such that electrostatic charge will not be transferred. The tip must be backed by vacuum compatible compliant material so that damage to the optic is minimal. See technical discussion below.

- 6) The stops must be designed to allow for installation of baffles, targets and other components that are positioned near or on the suspension structure. See list under Applicable Documents.
- 7) The stops must be designed to damp the optic in 10 bounces or less, to the point where the stops are no longer contacting the optic. D. Coyne to rewrite
- 8) The stops must be designed to set a gap between the tip of the stop and the optic between .18mm and .53mm for the SOS and .01mm and .62mm for the LOS. See technical discussion on this issue below.
- 9) There must be provisions to measure each of the gaps with sufficient accuracy to meet the requirement in # 8. It is permissible to use the motion of the optic (for example using an optical lever, a theodolite or the sensor/actuator readouts) to determine the make/break contact event.
- 10) The stops shall have a non-rotating tip or shall have a maximum coefficient of friction between the stop tip and the optic of 1.

# **Physical Configuration**

This is a retrofit so all designs are constrained to fit into the LOS and SOS as currently designed. However, the stops need not be in the same locations nor, must their numbers be identical to the number currently used. One design need not serve all functions. Multiple designs may be utilized. For LOSs and SOSs the stops are all screws. Redesign of the stops requires consideration of the screw threads.

LOS



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The LIGO1 LOS suspensions use stops that range in screw size from <sup>1</sup>/<sub>4</sub>-20 to <sup>1</sup>/<sub>2</sub>-13. The stops currently touch the barrel and the chamfers of the optic. It is optimal to shift away from chamfer contact unless contact can be assured to be square and centered on the chamfer.

## SOS

The LIGO1 SOS suspensions use stops that range in screw size from 6-32 to <sup>1</sup>/<sub>4</sub>-20. The screws touch the barrel, and front and back faces of the optic.

### Material

All materials and processes used to fabricate the stops must comply with LIGO Vacuum Compatible Materials List, LIGO-E960050. If the stop is removed prior to installation in the vacuum chamber, other materials may be considered, as long as they do not contaminate the optic or other suspension components. Questions about materials should be addressed to the LIGO Vacuum Standards Board.

# Background

The PNI suspension at MIT utilized ¼-20 screws with counter bores in the tips for stops. Into the counter bore, a compression spring was pressed. This type of stop is still used on the small optic suspensions (SOS.) It is not used extensively as the spring can cause more bouncing of the optic in the even of a sudden movement rather than damping of the movement.

A number of stop designs have been prototyped. Teflon screws were tried but the material is so soft that it is not appropriate for screw material because, when paired with metal internal threads, it peels away easily (creating "spaghetti".) In close proximity to fused silica optical material, problems with electrostatic charging becomes worse.

Carbon-doped Teflon stop screws were prototyped. The conductive material removed the electrostatic problem but the screws themselves created particulate matter when screwed into metal threads.

Viton corks were fabricated and pushed into counter bores in metal screws. These corks are used on the LIGO1 LOSs. However, it is difficult to line up the centerline of the cork with the centerline of the screw. This problem may make the positioning of the cork 0.5mm away from the optic's chamfer difficult for an unseasoned installer. Also, there is quite a bit of friction between the optic and the Viton, so it requires finesse for suspending and balancing operations.

Rectangles made from Viton cable clamp liners are press fit into counter bores in screws. This stop design is used on the SOSs. Again, the centerline of the rectangle is often misaligned from the screw centerline, making positioning difficult. Again, the friction factor makes an alternate for suspending and balancing attractive.

# Technical



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### Conductivity

One limit on the conductivity of the stops is set by eddy current damping between the stops and the magnets on the optics. A discussion of the theory of eddy current damping with application to the suspensions can be found in LIGO-T000119-0, "Use of magnets in the suspension design". There, the following limits on the net force-per-velocity parameter b for the suspensions are derived:

 $b_{Max,LOS} = 3.6 \times 10^{-6} \text{ N/(m/s)}$ 

 $b_{max,SOS} = 3.1x10^{-6} \text{ N/(m/s)}$ 

The above limits should be trivial to achieve provided that a few basic facts about eddy current damping are kept in mind. The basic scaling factors are: linear in conductivity, quadratic in magnet strength, linear in the perimeter of the typical loop, linear in the cross-section over which there can be loops and inverse sixth power in the distance from the magnet. Geometrically the force is maximised when the tangent to the loop, the magnetic field and the relative velocity are mutually perpendicular. To give a wildly pessimistic scenario for reference, the above damping would be created by one 1/8" diameter aluminum screw along the axis of an optic magnet with a distance of approximately 5 mm from the center of the magnet to the end of the screw. However using stainless steel instead of aluminum would give a factor of 30 improvements, and every factor of 2 in separation would give a factor of 64.

Other limits on the conductivity derive from the need to reduce electrostatic forces between the optic and the stops, but the limits will depend on the strategy employed. Some general observations:

To have electrostatic force requires charge on both the optic and on the stop. Depending on the scenario, the charge can be either permanent static charge, or induced charge that has been pulled into the area by charges on the other side.

Charge separation will tend to occur by mechanical contact between objects of dissimilar materials (especially glass and Viton) contacting over large areas. Unless some of the resulting charge dissipates, this will cause a strong attractive force between the two objects.

The resistivity of clean fused silica is so high that charge that accumulates on the optic will probably remain there indefinitely. Therefore it's important to minimize the amount that does so. The fate of charge on the stop side is more controllable.

To the extent that charge separation occurs between the optic and the stop, it will generally be better for the stop to be conducting and grounded, so that most of the charge on the stop side drains away (leaving only a small residual component maintained by induction). Since the capacitance of the tip is only a few picofarads, the resistance of the stop could be as much as a giga-ohm and still dissipate charge in an acceptable time. Any metal part will meet this with orders of magnitude to spare.



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However if the charge separation is within the tip of the stop (e.g. and i.e., because some sandwich contruction with a glass tip and a Viton elastic element has been used), then it is probably better for the shaft of the stop to be insulating (or contain an insulating section as near as practical to the tip). This is because if the positive and negative components of the charge remain close together they will have almost cancelling effects at the optic.

### **Quantitative Limits on Motion**

Limits on motion are required to protect the magnets in the event of seismic or other unforeseen activity. These limits define the limits of the rigid body motion of the body.

The limits are dependent on the tolerance buildup of the sensor/actuator head itself, the bias range of the electronics, the allowable magnet de-centering due to the magnet gluing fixture and the geometry of the magnets with respect to the optic.

For this analysis, motion of the optic is defined by 6 degrees of freedom; three translation and three rotation. These are lateral (x), vertical (y), position (z), pitch ( $\alpha$ ), yaw ( $\beta$ ) and roll ( $\theta$ ). Note, these axes are different from the global axes definition for LIGO.

For both the SOS and the LOS, the tolerance buildup in the sensor/actuator assembly provides a minimum gap between the edge of the magnet and the photodiode filter of .049"[1.25mm] in the y axis and .176" [4.46mm] in the x and z axes.

The bias range for the SOS is 28mrad p-p in pitch and yaw. The bias range for the LOS is 0.5mrad p-p in pitch and yaw. Data is from the Large and Small Optics Suspension Electronics Final Design, LIGO-T980043. Bias range influences the allowable motion budget of the translations.

The allowable magnet de-centering relative to the sensor/actuator central axis (as measured by a telescope mounted in the sensor/actuator bracket) is 0.5mm for the SOS and LOS.

### <u>SOS</u>

The SOS optic is 75mm diameter and 25mm thick, see figure below. The distance between the center of the optic and the center of the magnet, in x and y, is .972"[24.7mm=.0247m.] The distance between the center of the optic and the end of the back magnets, in z, is .709"[18mm=.018m.]

The distance between the center of the optic and the end of the side magnet is 1.59" [40.4mm=.0404m.] The minimum length of the side dumbbell standoff is .080"[2.03mm=.002m.] The z distance from the centerline of the magnet and the center of the magnet is .0625"[1.6mm=.0016m.]



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Below are the calculations for the movement of the optic, and therefore the magnets, due to bias and the minimum allowable movement of the magnets in translation.

#### **Bias Movement**

The optic moves under the influence of the actuator. The amount of movement is dependent on the maximum bias voltage. Below are calculations of the maximum translation/rotation for the maximum bias voltage. This is defined below as maximum desirable range of motion.

#### <u>a Pitch</u>

Bias of back magnets translated into z motion = 14mrad(.0247m) = .346mm



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# EARTHQUAKE AND SAFETY STOP DESIGN REQUIREMENTS

Bias of back magnets translated into y motion = 14mrad(.018m) = .252mm

<u>β Yaw</u>

Bias of back magnets translated to z motion = 14mrad(.0247m) = .346mm

Bias of back magnets translated into x motion = 14mrad (.018m) = .252mm

Bias of side magnet translated to z motion = 14mrad(.0404m) = .566mm

Bias of side magnets translated into x motion = 14mrad(.0016m) = .022mm

<u>θ Roll</u>

There is no induced roll bias.

#### **Movement Boundaries**

Earthquake stops are needed to allow for the bias movement of the optic but to stop/damp the movement of the optic before it hits something and damages the optic or the magnet/dumbbell assemblies. Generally, the maximum translation/rotation is defined by the optic banging into the sensor/actuator or the magnet in the sensor/actuator banging into one of the components. The calculations below define the maximum allowable movement of the optic, in a worst-case tolerance build-up situation.

X translation

Optic contacting the face of side sensor/actuator = 2.03mm

Back magnet contacting the sensor/actuator inside wall = 4.46mm - .5magnet de-centering = 3.96mm

Y translation

Side magnet contacting the photodiode filter = 1.25mm

Back magnets contacting the photodiode filter = 1.25mm - .5mm de-centering = <u>.75mm</u>

Z translation

Side magnet contacting the sensor/actuator inside wall = 4.46mm

Optic contacting the face of sensor/actuator = 2.03mm



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# EARTHQUAKE AND SAFETY STOP DESIGN REQUIREMENTS

The tables below compare for each axis the desirable range due to bias and the minimum optic motion before fouling.

Back sensor/actuators:

Axis	max. bias, mm	min. gap,mm	
X	.252	3.96	
Y	.252	.75	
Z	.346	2.03	
Side s	ensor/actuators:		
X	.022	2.03	
Y		1.25	
Ζ	.566	4.46	

The smallest gap is .75mm vertically. It's corresponding max. bias is 14mrad in pitch. Because pitch motion causes the optic to hit something in the shortest distance, it will be used to define the minimum and maximum positions for the earthquake stops. The earthquake stops that are closest to the center of the optic (i.e. that have the shortest lever arm) will define the min. and max positioning of the earthquake stop tips. The front earthquake stops are positioned below the optic centerline at the position of the magnets in back, or r=.018m, along the z axis. Then .75mm/.018m = 42 mrad is the max pitch rotation, because that is the point where, in a worst case tolerance build-up, a magnet will hit a photodiode filter in a sensor/actuator head.

For the SOS, the earthquake stops should be positioned at a distance from the optic of 14mrad(.0125m) = .175mm min. and 42mrad(.0125m) = .525mm max.

### LOS

The LOS optics are 250mm diameter and 100mm thick. The distance between the center of the optic and the center of the magnet, in x and y, is 3.182"[80.82mm=.0808m] for most LOSs. However, for the BS, x=2.382"[60.5mm=.0605m] and y=4.125"[104.78mm=.1048m]. The distance between the center of the optic and the end of the side magnet is 5.047"[128.19mm=.128m]. The minimum length of the side dumbbell standoff is .120"[3.05mm=.0304m]



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# EARTHQUAKE AND SAFETY STOP DESIGN REQUIREMENTS



Below are the calculations for the movement of the optic, and therefore the magnets, due to bias and the minimum allowable movement of the magnets in translation.

### **Bias Movement**

<u>α Pitch</u>

Bias of back magnets translated into z motion = .25mrad(.0808m)=.0202mm

Bias of back magnets translated into y motion = .25mrad(.056m) = .014m

Bias of back magnets translated into z motion for BS = .25mrad(.1048m)=.026mm



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# EARTHQUAKE AND SAFETY STOP DESIGN REQUIREMENTS

Bias of back magnets translated into y motion for BS = .25mrad(.0288m) = .007mm

<u>β Yaw</u>

Bias of back magnets translated to z motion = .25mrad(.0808m)=.0202mm

Bias of back magnets translated into x motion = .25mrad(.0288m)=.007mm

Bias of side magnet translated to z motion = .25mrad(.128m)=.032mm

Bias of side magnet translated into x motion = .25mrad(.0016m) = .0004mm

<u>θ Roll</u>

There is no induced roll bias.

#### **Movement Boundaries**

Earthquake stops are needed to allow for the bias movement of the optic but to stop/damp the movement of the optic before it hits something and damages the optic or the magnet/dumbbell assemblies. Generally, the maximum translation/rotation is defined by the optic banging into the sensor/actuator or the magnet in the sensor/actuator banging into one of the components. The calculations below define the maximum allowable movement of the optic, in a worst-case tolerance build-up situation.

X translation

Optic contacting the face of side sensor/actuator = 3.05mm

Back magnet contacting the sensor/actuator inside wall = 4.46mm - 0.5magnet de-centering = 3.96

Y translation

Side magnet contacting the photodiode filter = 1.25mm

Back magnets contacting the photodiode filter = 1.25mm - 0.5mm de-centering = .75mm

Z translation

Side magnet contacting the sensor/actuator inside wall = 4.46mm

Optic contacting the face of sensor/actuator = 2.03mm



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The tables below compare for each axis the desirable range due to bias and the minimum optic motion before fouling.

Back sensor/actuators:

Axis	max. bias, mm	min. gap,mm
X	.007	3.96
Y	.014	.75
Ζ	.026	2.03
Side s	ensor/actuators:	
X	.0004	3.05
Y		1.25
Ζ	.032	4.46

The smallest gap is .75mm vertical, like the SOS. It's corresponding max. bias is .25mrad in pitch. Because pitch motion causes the optic to hit something in the shortest distance, it will be used to define the minimum and maximum positions for the earthquake stops. So, .75mm/.0288m = 26mrad is the max pitch rotation, because that is the point where, in a worst case tolerance build-up, a magnet will hit a photodiode filter in a sensor/actuator head. The earthquake stops that are closest to the center of the optic (i.e. that have thee shortest lever arm) will define the min. and max. positioning of the earthquake stop tips. All stops are at the OD of optic, with the BS stops having the shortest lever arm along the z axis of .024m.

For the LOS, the earthquake stops should be positioned at a distance from the optic of .25mrad(.024m) = .006mm min. and 26mrad(.024m) = .624mm max.